

# LYNX HMA Mk 8 SRU

# AIRCREW MANUAL

# BOOK 1 - FLIGHT SYSTEMS, HANDLING & EMERGENCIES

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Prepared by Handling Squadron



### **NOTES TO USERS**

- 1. This Manual is complementary to the Lynx HMA Mk 8 SRU Release to Service (RTS) document and the associated Flight Reference Cards (FRC) (AP 101C-1308-14A & B). Performance data for the Lynx HMA Mk 8 SRU is contained in the Operating Data Manual (ODM) (AP 101C-1308-16). It should be read in conjunction with AP 101C-1308-15D, Book 2, CENTRAL TACTICAL SYSTEM (CTS).
- 2. The manual is divided by marker cards as follows:

Preliminary Matter

- Part 1. Description and Management of Systems
- Part 2. Limitations (Refer to RTS)
- Part 3. Handling
- Part 4. Emergencies and Malfunctions
- Part 5. Illustrations
- 3. Each Part is divided into Chapters as listed on its marker card. Each sheet is identified by a Part, Chapter and Page reference at the foot of the page. Thus, a page bearing the reference: 1 4 Page 5 is page 5, of Part 1 Chapter 4.
- 4. The limitations are removed from Book 1, Part 2 at AL 2. The Lynx HMA Mk 8 RTS should be consulted to ascertain the latest release information. The contents of other parts of the Manual are mainly advisory but instructions containing the words 'is to' and 'are to' are also mandatory.
- 5. The Manual and its associated FRCs aim to provide the best operating instructions and advice currently available for normal and abnormal procedures. Nothing in these publications removes the requirement to comply with MAA regulatory requirements. The application of sound airmanship and judgement applies at all times and is paramount. Any deviation from the prescribed procedures and drills will need to be fully justifiable and users are strongly advised to record this justification to aid any subsequent inquiry or investigation.
- 6. Amendment Lists will be issued as necessary and each amendment list instruction sheet will state the main purpose of the amendment and include a list of modifications covered by the amendment. New or amended matter of importance will be indicated by change bars, positioned in the outside margin alongside the amended text. The number of the amendment list by which a sheet was issued appears at the bottom of the right-hand page and any amendment marks on either side of the sheet refer to that amendment. However, when a new Chapter is issued with an amendment list, or an existing Chapter is completely revised, this fact is noted within the heading of the Chapter and the symbols do not appear.
- 7. The following conventions are observed throughout this Manual:
  - a. The actual markings on controls or indicators are shown in the text by CAPITAL letters.
  - b. Unless otherwise stated, all airspeed, heights and temperatures are indicated values.
  - c. Information is published using the following:

WARNINGS Imply the possibility of death or injury.

CAUTIONS Imply the possibility of damage to the aircraft or its equipment.

NOTES are inserted to clarify the reason for a procedure or to give information which, while not essential to the understanding of the subject, is useful to the reader.

8. Modification numbers are only referred to in the Manual when it is necessary to differentiate between pre- and post-mod states. For ease of reference, a list of the modifications mentioned in the text is included in the preliminary pages of the book, with a cross reference to the location in the text of the modification details.

### IMPORTANT

Comments and suggestions regarding this Manual should be forwarded to the User Authenticator, using MoD Form 765X for onward transmission to the Lynx PT and Officer Commanding, Handling Squadron, Boscombe Down, Salisbury, Wilts, SP4 0JE

The User Authenticator for this manual is:

NFSF (RW) RNAS Yeovilton Somerset

**BA22 8HT** 

The Officer Commanding Handling Squadron Boscombe Down Salisbury Wilts SP4 0JE

The Publications Authority for this manual is:

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OFFICIAL SENSITIVE

MOD Form 765X (Revised Apr 14)

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### Instructions for Use

- 1. MOD Form 765X has been introduced to maintain a full audit trail of changes to aircrew publications and documents to ensure that both the User Authenticator and Handling Sqn are involved at the earliest opportunity after the form has been raised.
- MOD Form 765X is to be raised by the Individual who observed a deficiency, omission or inaccuracy in the Aircrew Manual, Flight Reference Cards, Operating Data Manual, Mission Operating Procedure cards, Flight Test Schedule or Aircrew Landaway Flight Servicing Schedule. Apart from typographic errors, a separate MOD F765X is to be raised for each deficiency, omission or inaccuracy being reported.
- 3. When an individual raises a MOD Form 765X (by completing the header detail and Part 1) he is to send the form to the User Authenticator, (RAF: STANEVAL; Army: A Avri Stas, HQAAC; RN: Naval Flying Standards Flight, RNAS Culdrose or RNAS Yeovilton as appropriate)
- 4. On receipt the User Authenticator is to complete Part 2, enter a serial number consisting of a 3-letter MOB designator, a 3-digit number (starting with 001 from1Jan each year) and 2 digits for the year (eg BZN/016/05), comment as appropriate and pass the form to the Project Team (PT), with a copy to Handling Sqn and a copy to the appropriate Release to Service Authority (RTSA). An ejectronic version of the Form is available on the Defence Intranet and the Form can be submitted electronically by the UA in the first instance but must be followed up by a signed hard copy.
- 5. The User Authenticator is to keep a register of all MOD Form 765X arisings.
- 6. The PT is to complete Part 3 of the MOD Form 765X and forward it to Handling Sqn or the Publication Organisation (copy to Handling Sqn), as appropriate, for action.
- 7. When the change proposed in the F765X is deemed by the UA or PT to be of an urgent flight safety or operational nature, the PT can authorize HS by e-mail to proceed with the appropriate amendment action in advance of the completion and signature of Part 3 of the F765X. When issue of an ANA closes a F765X, the Publication Organisation is to raise a Tech Pubs task to ensure that the change is incorporated at the next routine amendment.

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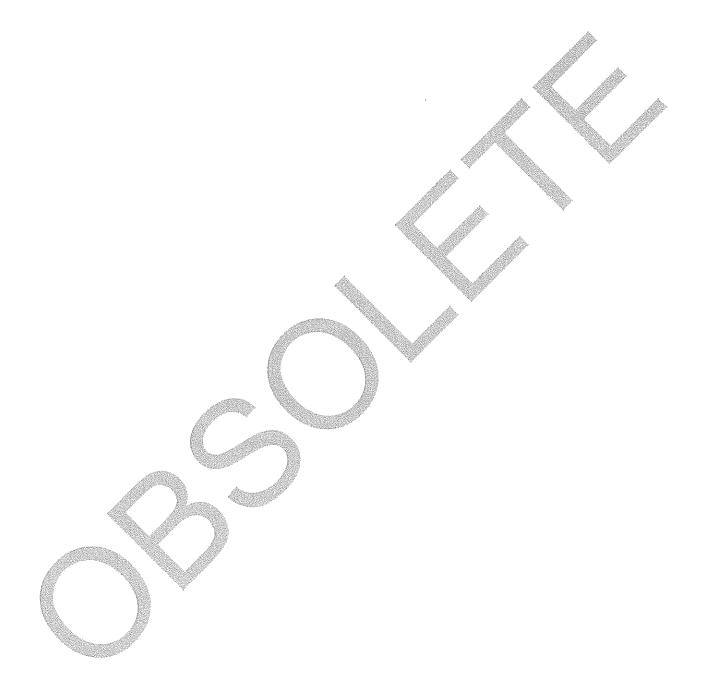
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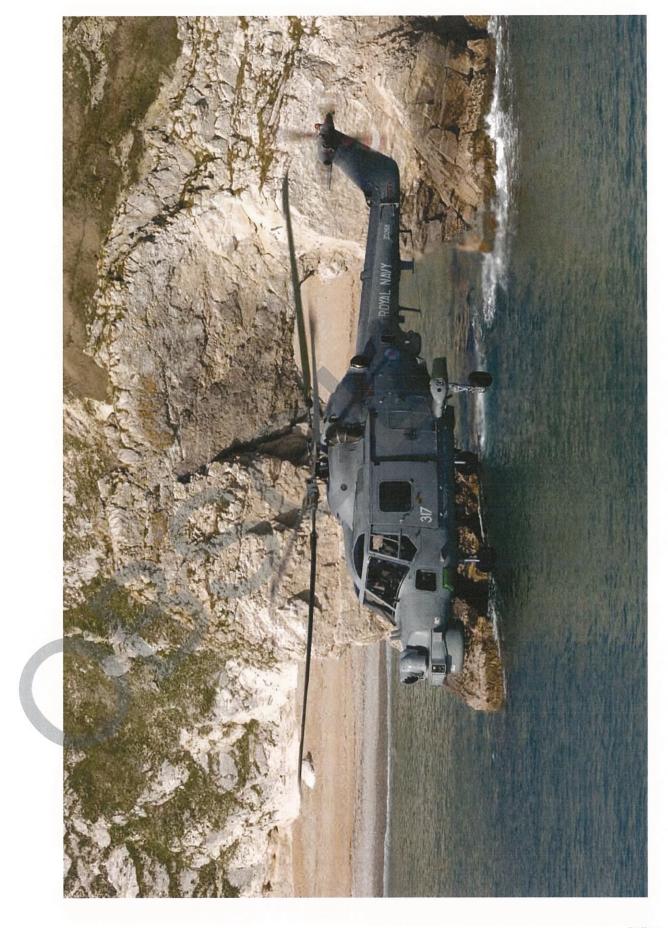
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Flight Reference Cards ....... AP 101C-1308-14A & 14B

Operating Data Manual ... ... ... ... ... AP 101C-1308-16



# LYNX HMA MK 8

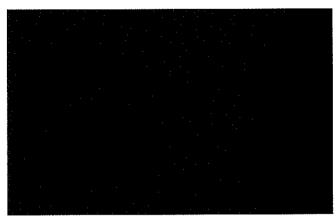


### INTRODUCTION

### General

1. The Lynx HMA Mk 8 is an anti-surface and anti-submarine search and strike helicopter. Secondary roles include search and rescue, vertical replenishment and troop carrying. The aircraft is powered by two Rolls Royce Gem Mk 204/205 series turbo-shaft engines driving a 4-bladed main rotor and a 4-bladed tail rotor. Fig 1, Fig 2 and Fig 3 show the general arrangement of the aircraft, Fig 4 and Fig 5 show the principal dimensions, and Fig 6 shows heavy and light stores.

### Fuel System



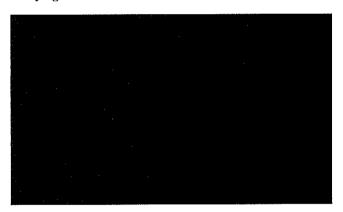
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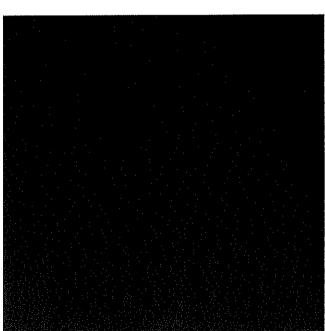
Landing Gear



Flying Controls



Engines



**Rotor Systems** 



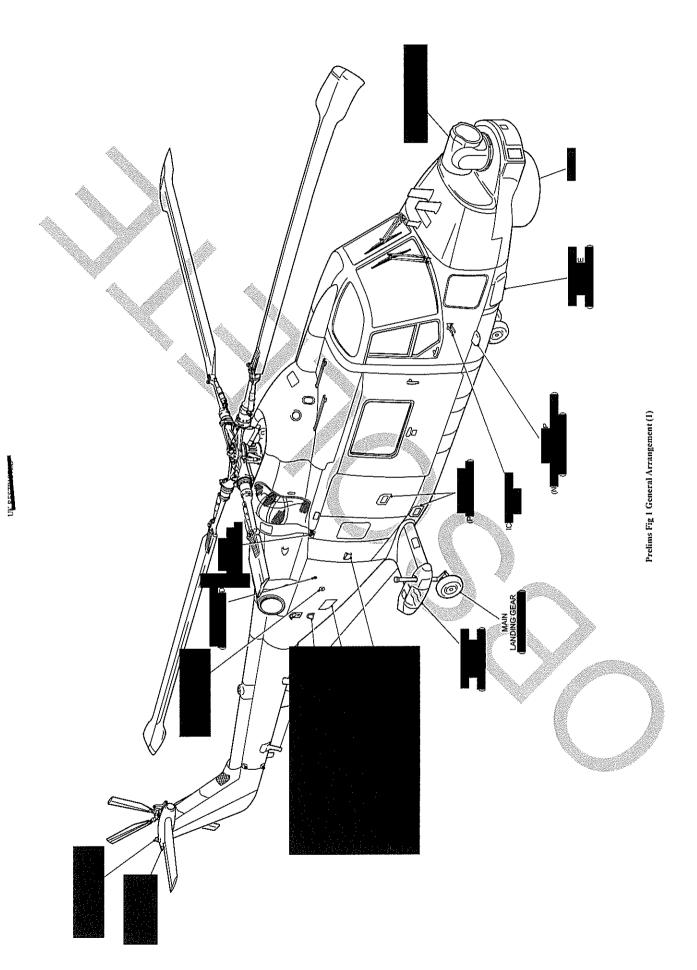
Flight Control System



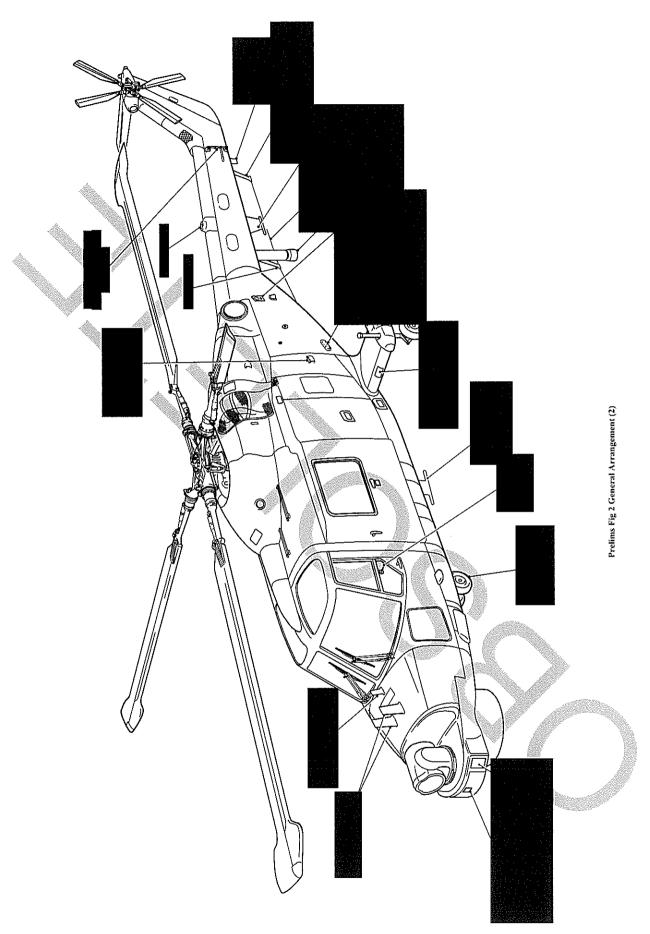
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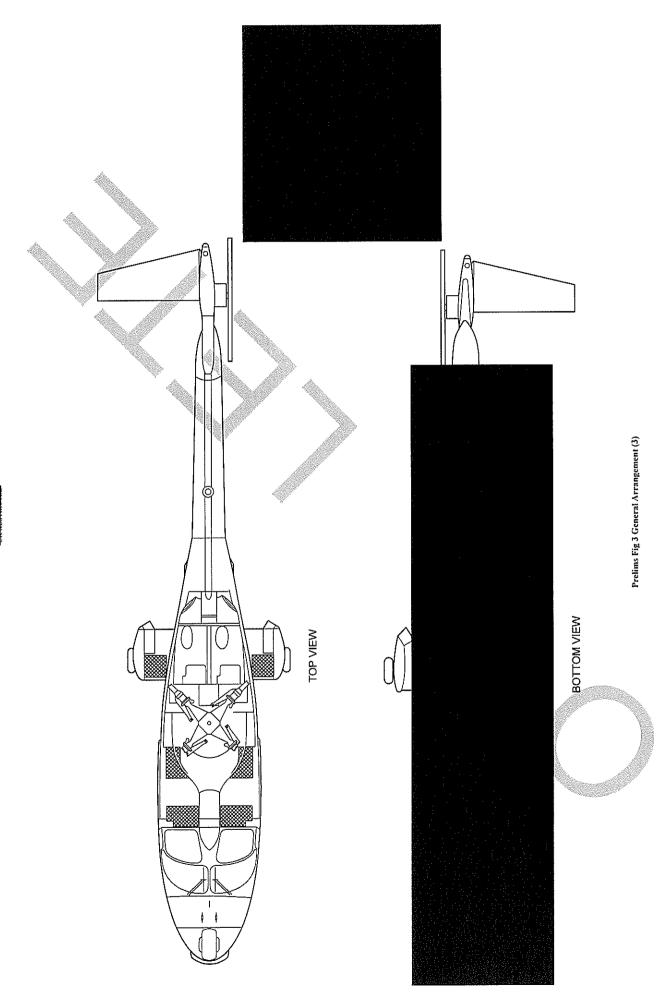
Prelims

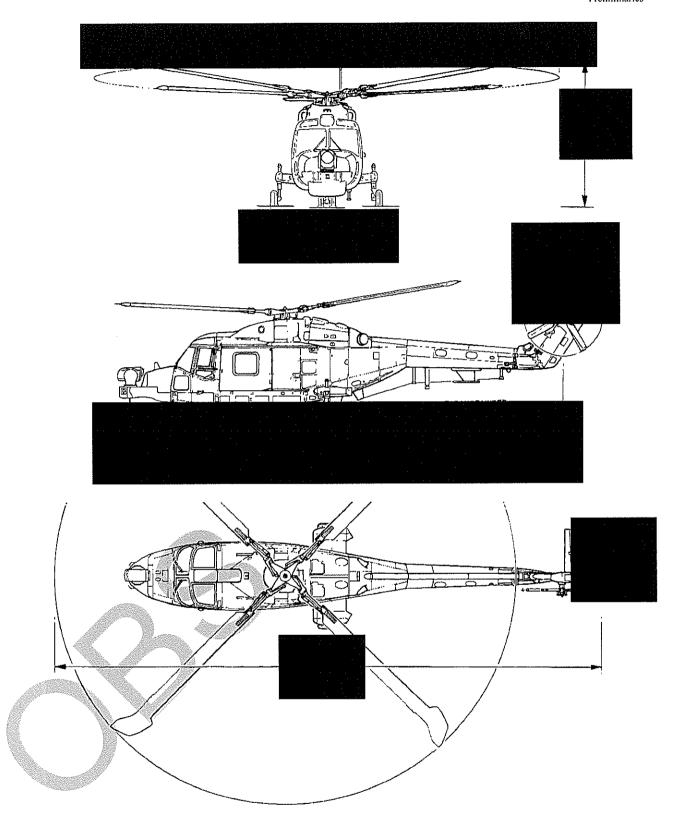


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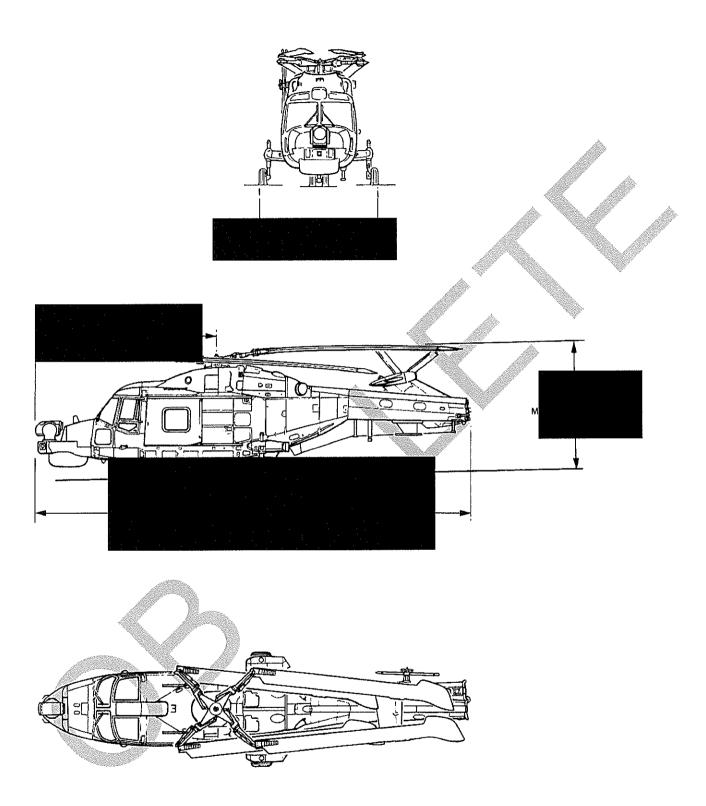


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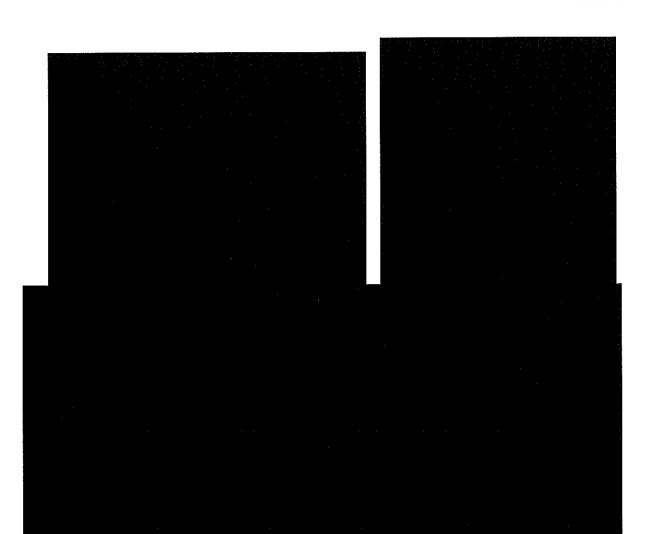


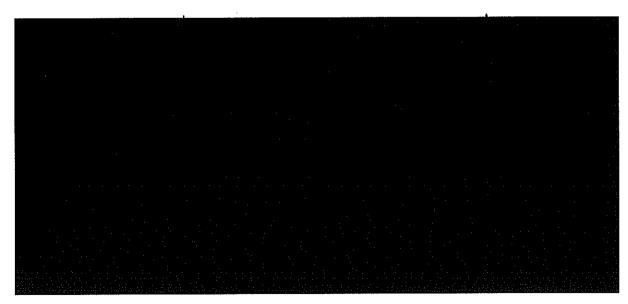


Prelims Fig 4 Principal Dimensions (1)



Prelims Fig 5 Principal Dimensions (2)





Prelims Fig 6 Heavy and Light Stores

### **LEADING PARTICULARS**

### LEADING PARTICULARS

NAME LYNX HMA Mk 8

TYPE Two-engined helicopter with main and tail rotors

CREW Pilot and Observer

DUTIES Primary Role: Anti-Surface Vessel

Anti-Submarine Search and Strike

Secondary Search and Rescue

Roles: Vertical Replenishment

Casualty Evacuation

### MAIN DIMENSIONS

(Refer to Principal Dimension diagrams, Fig 4 and 5)

### **POWER UNIT**

Rolls-Royce Gem Engines (x 2)

Type

**Engine Oil System** 

Oil type

Oil tank capacity

Usable oil quantity

### **FUEL SYSTEM**

Approved fuels

Alternative fuels

### Fuel Tank Capacities

The fuel tank capacities are as follows:



Tank .	Litres	Kilograms	Gallons
Normal System			
No 1 collector			
No 2 collector			
No 1 main			
No 2 main			
Forward			
Total, gravity refuelled			
Total, pressure refuelled			
Overload System			
Overload			
Total, all tanks gravity refuelled			
Note 1: Fuel Specific Gravity (SG) is 0.8			
Note 2: Approximately 10 litres of fuel i		ngauge d in th	e aircraft basic mass

### **ROTOR TRANSMISSION**

### Gearbox Oil Type

Main rotor damper fluid Sleeve spindle fluid

### **Gearbox Oil Capacities**

Main

Intermediate

Tail

### LANDING GEAR

Type

### Tyre pressures

Shock absorber capacities (three)

Shock absorber fluid

Wheel locks and nosewheel castoring

### HYDRAULIC SYSTEM

Fluid reservoir (two) capacity

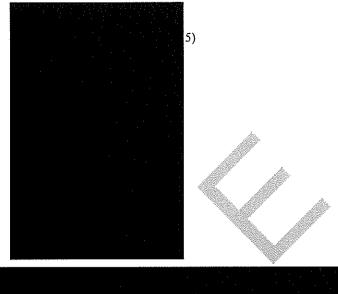
Approved fluid

### **ELECTRICAL INSTALLATION**

DC starter/generators (two)

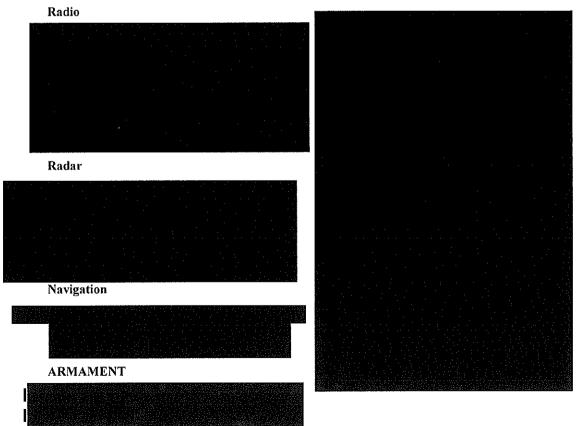
Alternators (two)

Battery





### RADIO, RADAR AND NAVIGATION FACILITIES



### LIL' DEFFERENCE DE

### ABBREVIATIONS USED IN TEXT

AC	Aircraft	EVT	Exercise Variant Torpedo
ACP	Auxiliary Control Panel		
ACU	Acceleration Control Unit	FCS	Flight Control System
ADS	Air Data System	FCU	Fuel flow Control Unit
ADU	Air Data Unit	FIAM	Flight In-Air Material
AFCS	Automatic Flight Control System	FRC	Flight Reference Card
AGL	Above Ground Level		
ΑI	Attitude Indicator	HEU	Headset Electronics Unit
ALU	Antenna Logic Unit	HF	High Frequency
AMSL	Above Mean Sea Level	HIRTA	High Intensity Radio Transmitting Area
ARI	Airborne Radio Installation	HP	High Pressure
ASE	Automatic Stabilisation Equipment	HTA	Helicopter Type Allowance
ASI	Airspeed Indicator	Hz	Hertz
ASL	Above Sea Level		
AUM	All-Up Mass	KVA	Kilo-Volt Amps
		Kw	Kilowatts
BIT	Built-In Test		
BOS	Buoyant Orange Smoke	IAS	Indicated Airspeed
		ISA	International Standard Atmosphere
CAC	Computer Acceleration Control		
CBLS	Carrier Bomb Light Stores	LP	Low Pressure
CCP	Communication Control Panel	LRU	Line Replacement Unit
CCS	Communications Control System		
CDNU	Control, Display and Navigation Unit	MASS	Master Armament Safety Switch
CG	Centre of Gravity	MGB	Main Rotor Gearbox
CPU	Control and Protection Unit	MI	Magnetic Indicator
CSCIU	Clear Signal Conversion Interface Unit	MoD	Ministry of Defence
CTS	Central Tactical System	MPOG	Minimum Pitch on Ground
CWP	Central Warning Panel	MSS	Marine Sound Signal
DCU	Display Control Unit	NATO	North Atlantic Treaty Organisation
DTD	Data Transfer Device	Nf	Free Power Turbine Speed
		NH	High Pressure Compressor Shaft Speed
ECL	Engine Condition Lever	NL	Low Pressure Compressor Shaft Speed
EED	Electro-Explosive Device	Nr	Main Rotor Speed
EIRA	Emergency Intercom and Radio Amplifier	NRV	Non Return Valve
EMRU	Electromagnetic Release Unit		
ESM	Electronic Support Measures	OAT	Outside Air Temperature
ESSB	Emergency Services Safety Break	ODM	Operating Data Manual

P2	Air Pressure, Low Pressure Compressor (4th Stage)
Р3	Air Pressure, High Pressure Compressor (4th Stage)
PC	Control Air Pressure (reduced P3 pressure)
PCU	Protective Control Unit
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PSI Pounds per Square Inch

QRF	Quick Release Fitting
QRM	Quick Release Mechanism

RAF	Royal Air Force
RCU	Remote Control Unit
RF	Radio Frequency
ROAW	

ROAW	<u> </u>
S S	Rotor Overspeed Audio Warning System
ROU	Rotor Overspeed Unit
SACRU	Semi-Automatic Cargo Release Unit
SCCS	Secure communications Control System
SCMU	Signal Conversion and Matrix Unit
SG	Specific Gravity
SGMS	Simulator Guided Missile System

SMS	Stores Management System
SMU	Stores Management Unit
SSCIU	Secure Signal Conversion and Interface Unit
SSL	Speed Select Lever
SUS	Signals Underwater Sound
T6	Power Turbine Inlet Temperature
TAS	True AirSpeed
TCRIT	Critical Time
Tq	Torque
TR	Tail Rotor
TRCF	Tail Rotor Control Failure
TRDF	Tail Rotor Drive Failure
TRT	Tail Rotor Thrust
TSD-G	Tactical Situation Display Graphic
TVT	Training Variant Torpedo
TVTH	Training Variant Torpedo Heavyweight
TVTL	Training Variant Torpedo Lightweight
TWRS	Torpedo Weapons Response Simulator

VDC	Volts Direct Current
VHF	Very High Frequency
$\boldsymbol{V}_{\text{max}}$	Maximum Permitted Speed

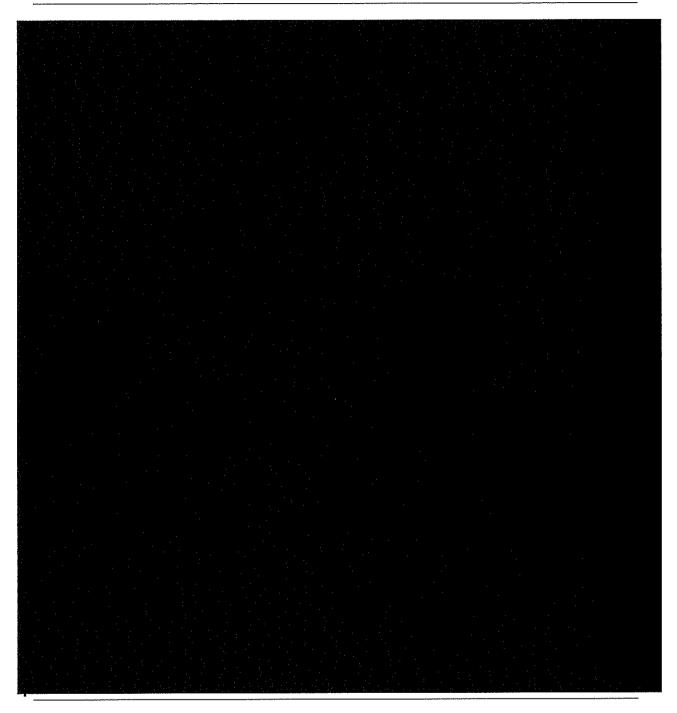
Ultra High Frequency

UHF

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# PART 1 DESCRIPTION AND MANAGEMENT OF SYSTEMS

# **List of Chapters**

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HYDRAULIC SYS	темs						•••	•••	
FLIGHT INSTRUM	MENTS		A.				•••	•••	
MISCELLANEOUS	S ROLE EQUI	PMENT						•••	,
GENERAL AND E	MERGENCY	EQUIP	MENT.						1
DEFENSIVE AIDS	EQUIPMENT								1

### PART 1

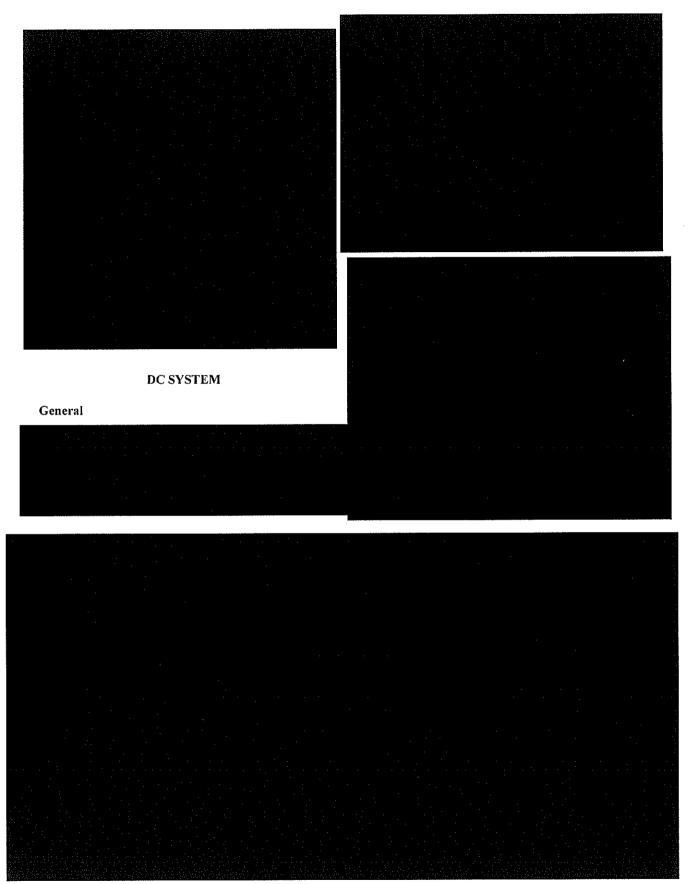
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AC Distribution								7
AC Ground Supplies								8
AC DC Supplies and D	istributi	on Pane	els Loca	ition				9
 7								

### **GENERAL**

Introduction



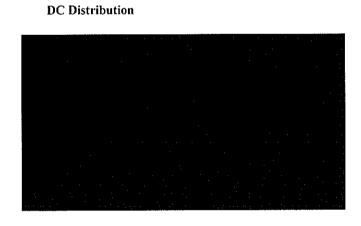
1 - 1 Fig 1 Electrical Systems Control Panel

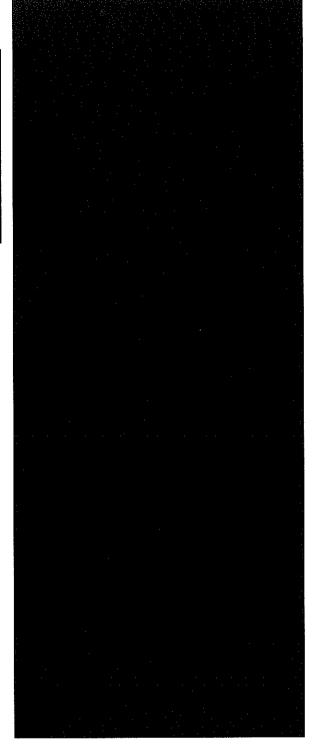
Table 1 - DC System Controls and Indicators

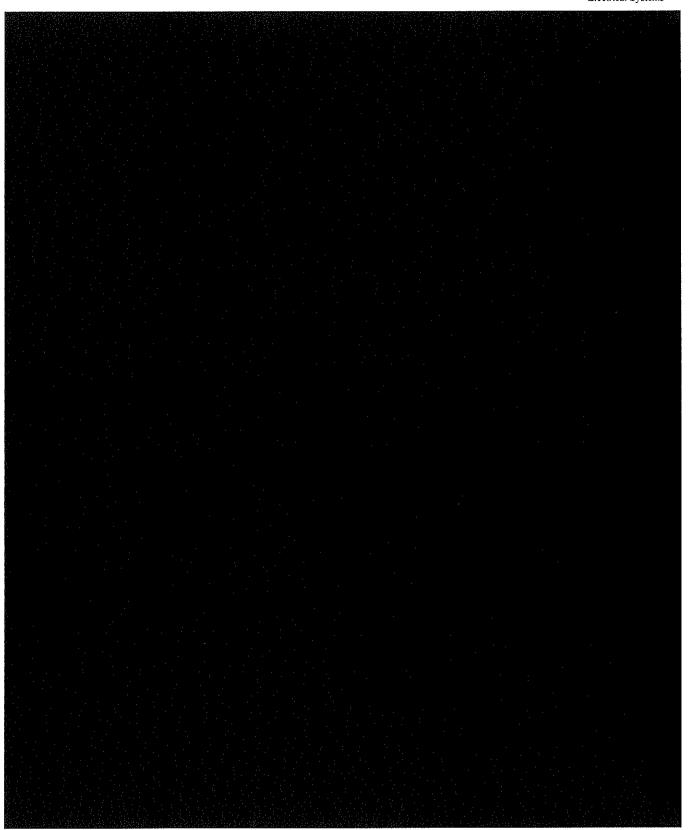
Table 1 - DC System Controls and Indicators
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봤을다. 마른 열면 그 중앙일이 되지만 하고 하는 사람이 보는 사람들이 가는 사람들이 일찍하는 수 있다.

# Battery Control Ground DC Supply

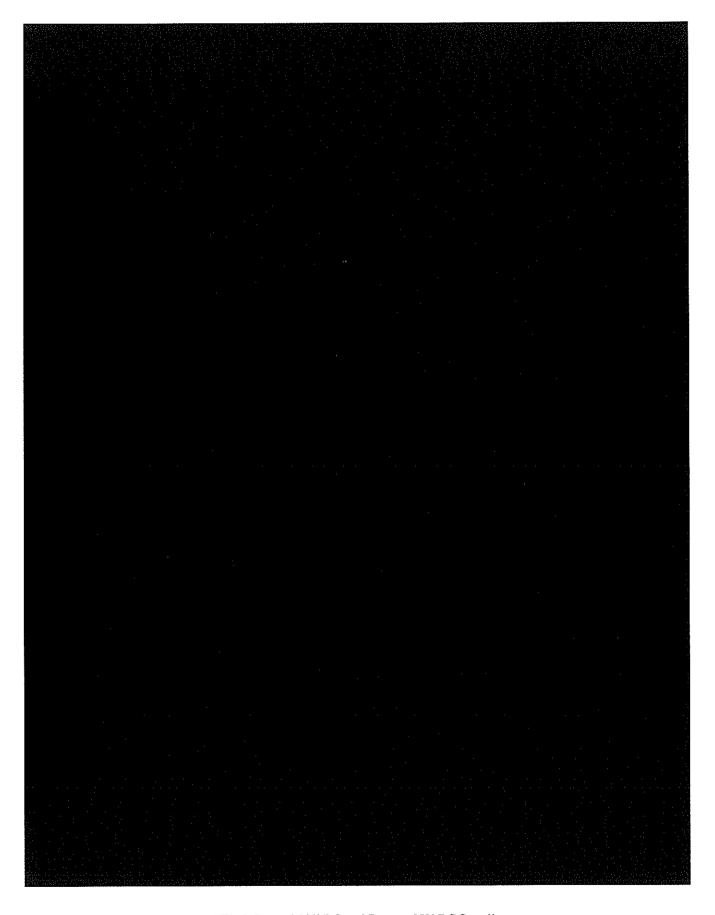




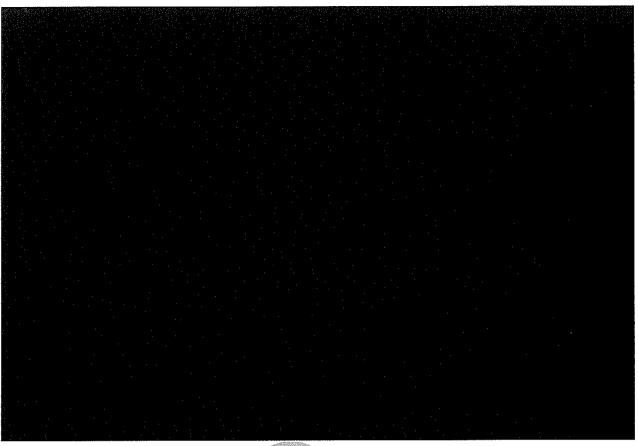




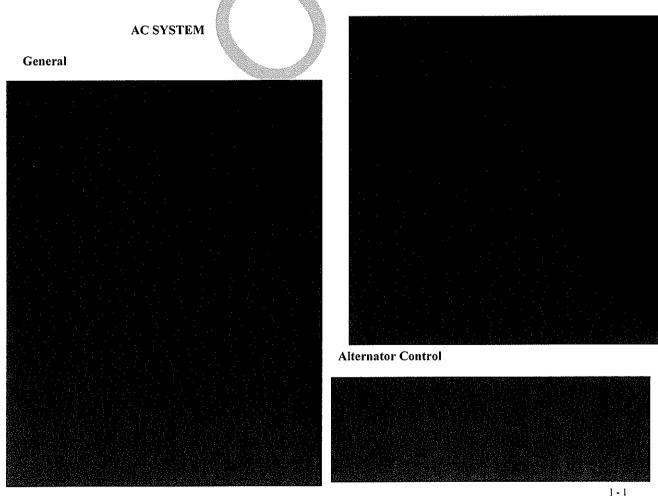
1 - 1 Fig 2 DC Generator Supplies and Indicators



1 - 1 Fig 3 Ground 28V DC and Battery 28V DC Supplies



1 - 1 Fig 4 DC Distribution Simplified



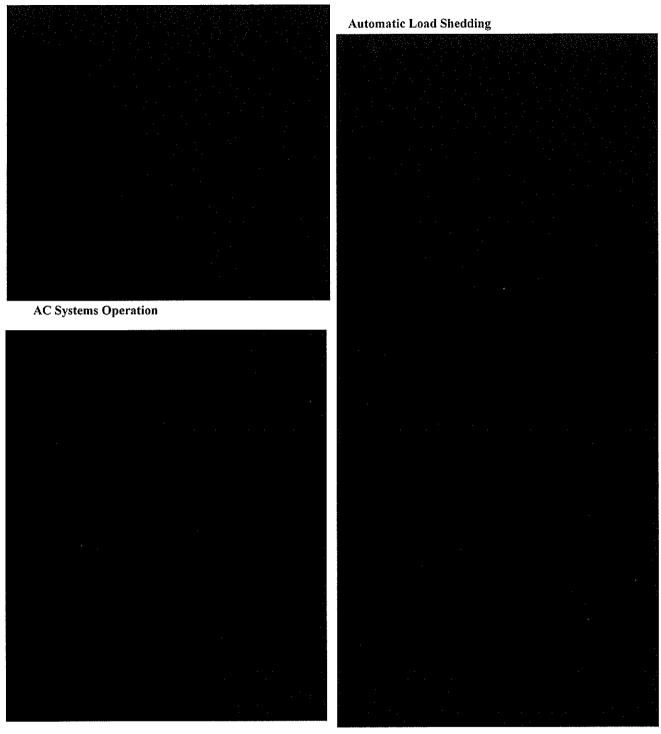
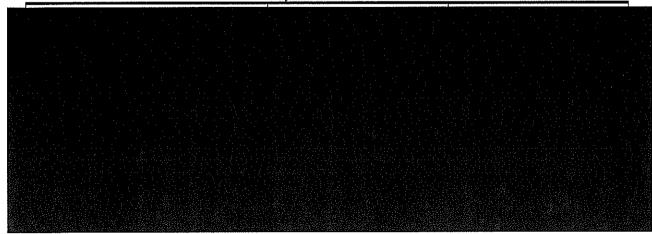
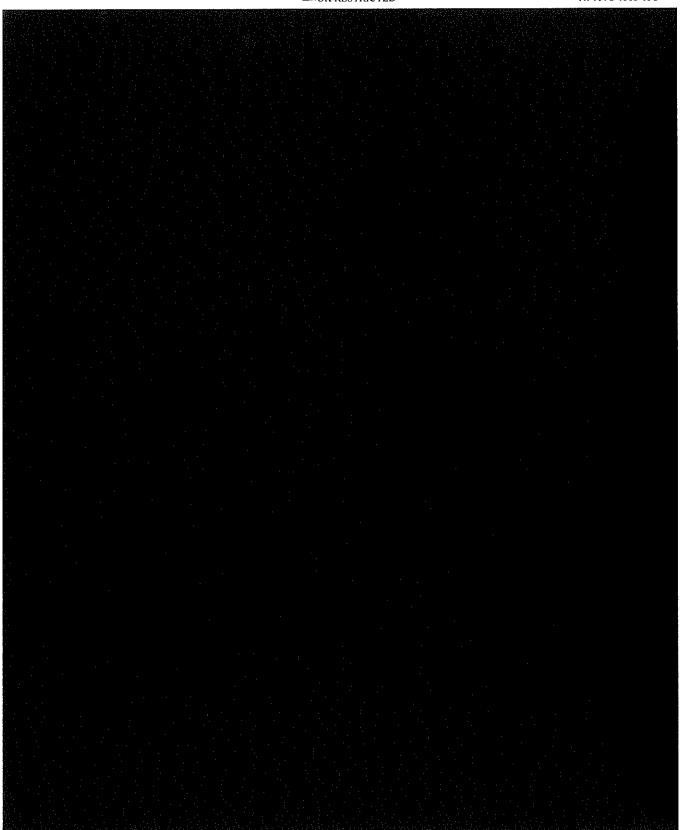


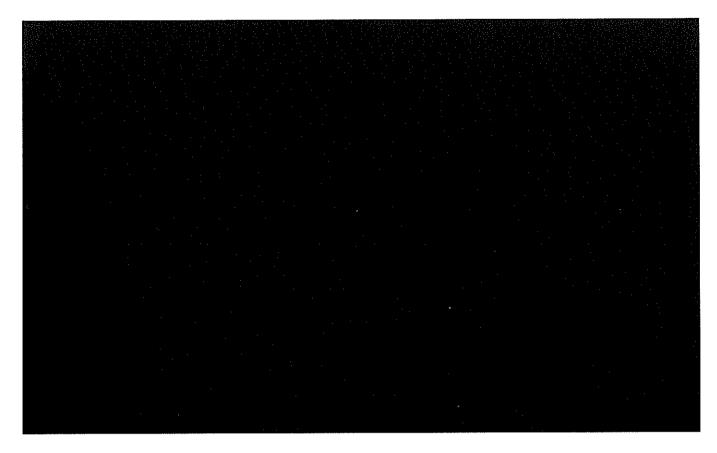
Table 2 - AC System Controls and Indicators



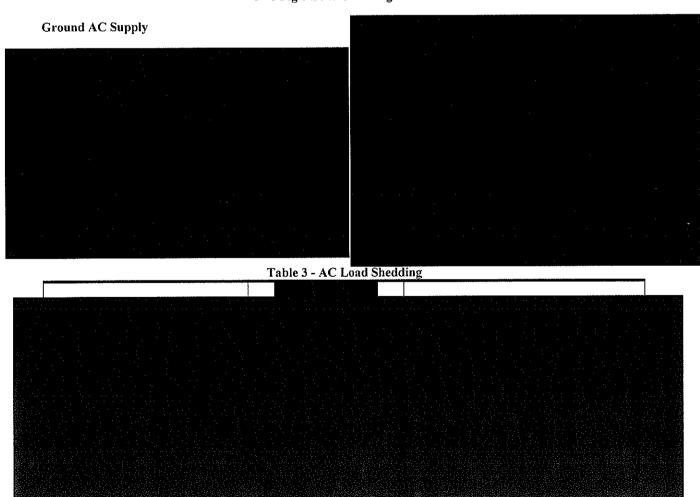
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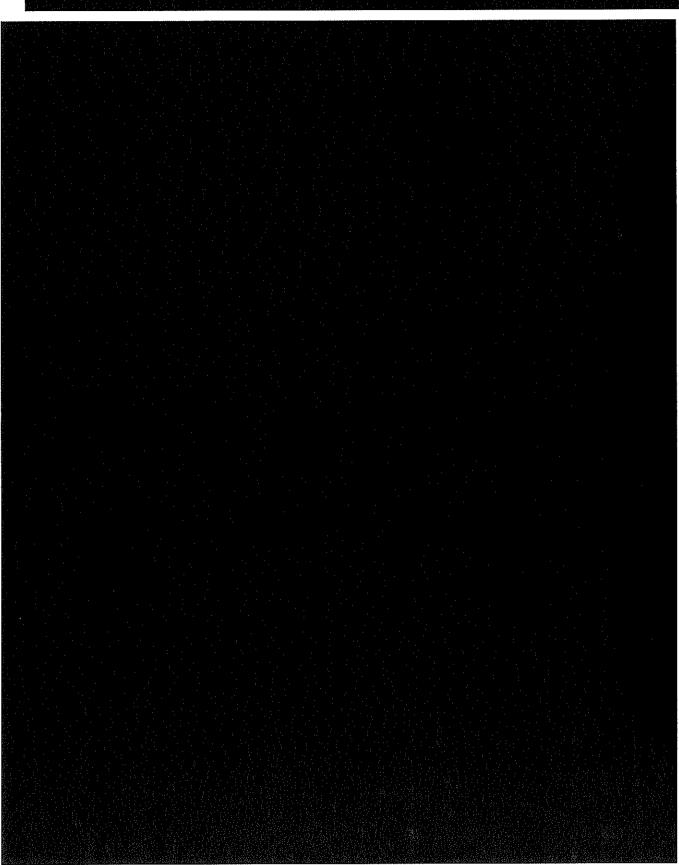
1 - 1 Fig 5 Alternator Supplies and Indicators



1 - 1 Fig 6 Load Shedding Circuit



AC Distribution

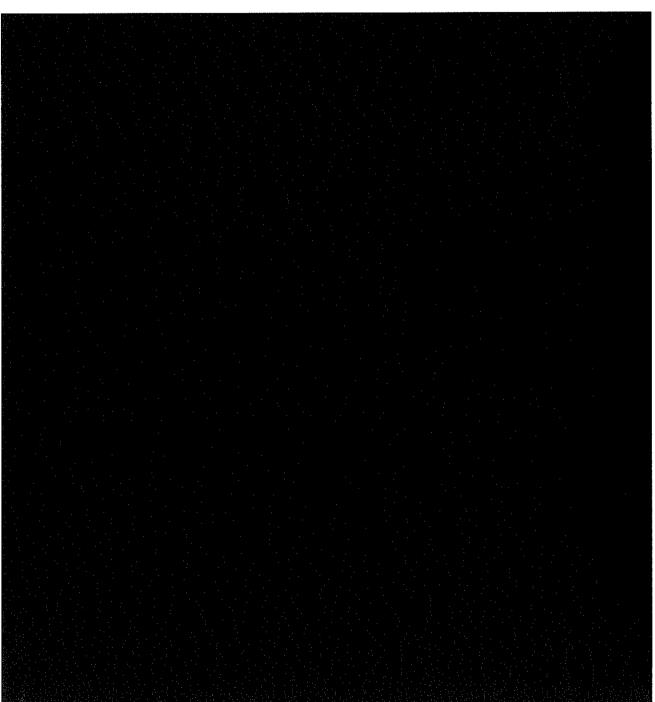


1 - 1 Fig 8 AC Ground Supplies

# Single Generator Failure NORMAL USE DC System **Double Generator Failure** wnite. Single Alternator Failure AC System **Double Alternator Failure**

1 - 1 Page 12





1 - 1 Fig 9 AC DC Supplies and Distribution Panels Location

Table 4 - Circuit Breakers Supplied from DC/AC Busbars

Circuit Breaker Panel/Number Service AC Supply, Control and Indications



Table 4 - continued

	WARNING SYSTEMS - CHAPTER 2
	FUEL SYSTEMS - CHAPTER 3
	Left Fuel System
, s	
	Right Fuel System
	ENGINE SYSTEMS - CHAPTER 4
	(2) 전환 경험 보고 있는데 보고 있다는 그 사람들은 아이들을 하는데 보고 있다. (2) 전환 전환 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1
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Table 4 - continued

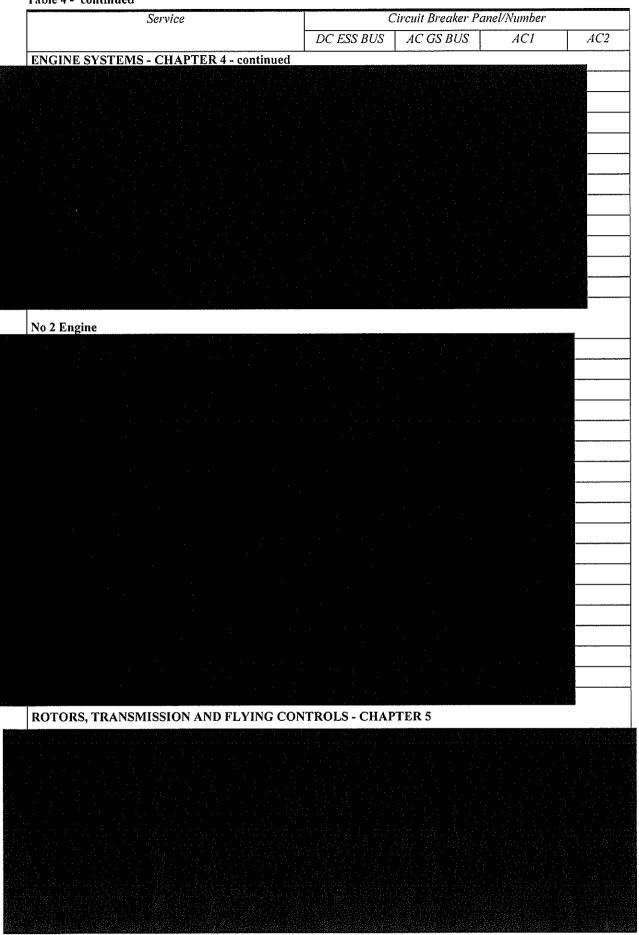


Table 4 - continued

Table 4 - Continueu				
Service		Circuit Breaker P		
	DC ESS BUS	AC GS BUS	ACI	AC2
ROTORS TRANSMISSION AND FLYING CO	NTROLS - CHAP	ΓER 5 - continu	ed	
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		·····	10 h	***************************************
Cyclic Stick Trim Control				
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			· **	
		<u>,                                    </u>		
FLIGHT CONTROL SYSTEM - CHAPTER 6				
HYDRAULIC SYSTEMS - CHAPTER 7				
				<u> </u>

Table 4 - continued

	Service	(	anel/Number				
		DC ESS BUS	AC GS BUS	AC1	AC2		
	HYDRAULIC SYSTEMS - CHAPTER 7 - continu	ied	aan ee waa ee madaa ee Sa				
	FLIGHT INSTRUMENTS - CHAPTER 8						
	MISCELLANEOUS ROLE EQUIPMENT - CHA	PTER 9					
	GENERAL AND EMERGENCY EQUIPMENT -	CHAPTER 10					
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				·			
					<u> </u>		
			_				
					1		
				u de la companya de l			
1.44							
	Landing Gear						
	Summit our						
					(Сопппиеа)		

ĺ	Service	(	Circuit Breaker Pa	inel/Number	
		DC ESS BUS	AC GS BUS	ACI	AC2
Ì	GENERAL AND EMERGENCY EQUIPMENT -	CHAPTER 10 -	continued	•••	
					<u> </u>
	Windscreen				V 100
	Cockpit Heating				
	Cocapa Henning				
	Cabin Heating				
٠.					
	Flotation Gear				
	Aft Equipment Bay				
	Art Equipment Day				
			· .		
ggar.					
Ša.					

Table 4 - continued Circuit Breaker Panel/Number Service DC ESS BUS AC GS BUS ACI AC2 The following circuit breakers refer to chapters in book 2 (Topic 15 D) but are detailed here to give an overall view of the electrical distribution system BOOK 2 (TOPIC 15D) **CENTRAL TACTICAL SYSTEM - CHAPTER 1 COMMUNICATIONS SUB-SYSTEM - CHAPTER 5** 

(Continued)

Table 4 - continued

	Service Service	Circuit Breaker Panel/Number						ervice Circuit Breaker Panel/Number		
		DC ESS BUS	AC GS BUS	ACI	AC2					
	COMMUNICATIONS SUB-SYSTEM - CHAPTE	R.5. continued								
	NAVIGATION SUB-SYSTEM - CHAPTER 8									
					<u>A</u>					
	WEAPONS SUB-SYSTEM - CHAPTER 9				Ì					
-				<u> </u>	<u> </u>					
	Armament Busbar 'A' Feeder Circuits									
	Armament Busbar 'B' Feeder Circuits				_					
	PID									

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#### PART 1

# **CHAPTER 2 - CENTRAL WARNING SYSTEM**

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1	Illustrations Central Warning System (Post Mod 5576)			•••		Fig 1
•						

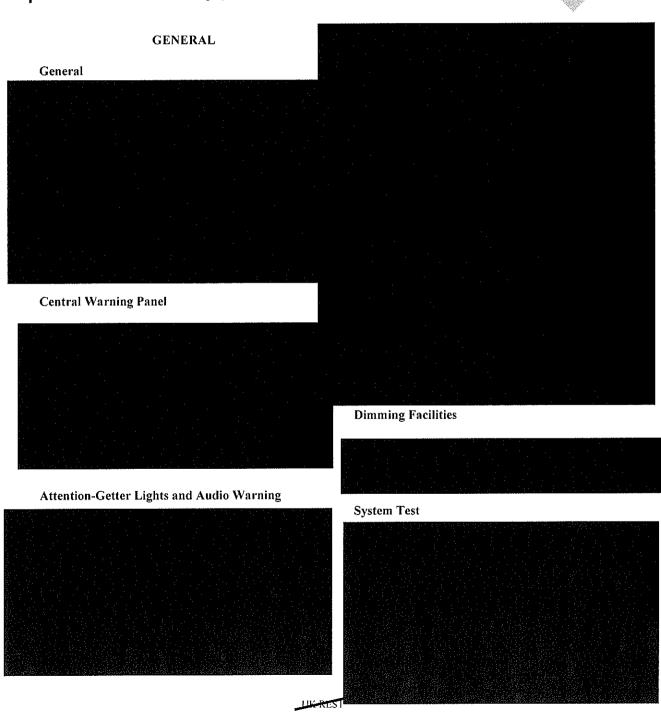
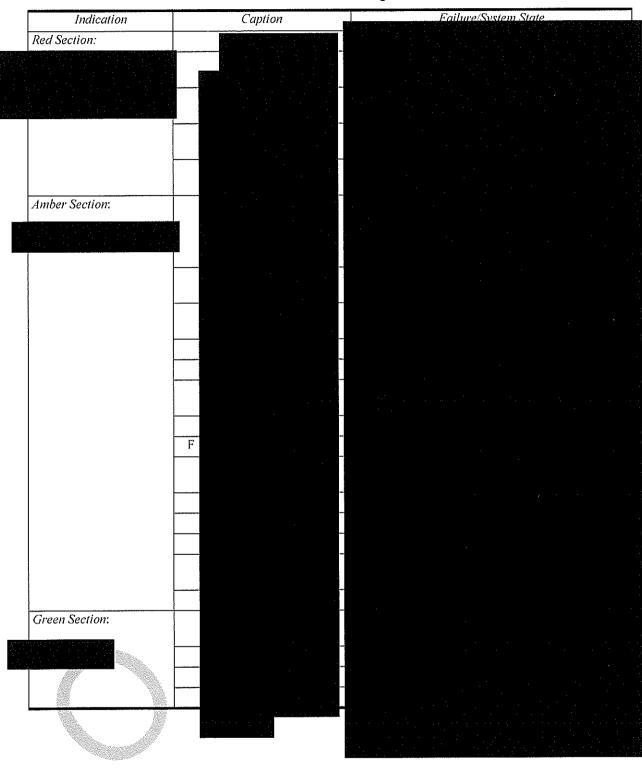
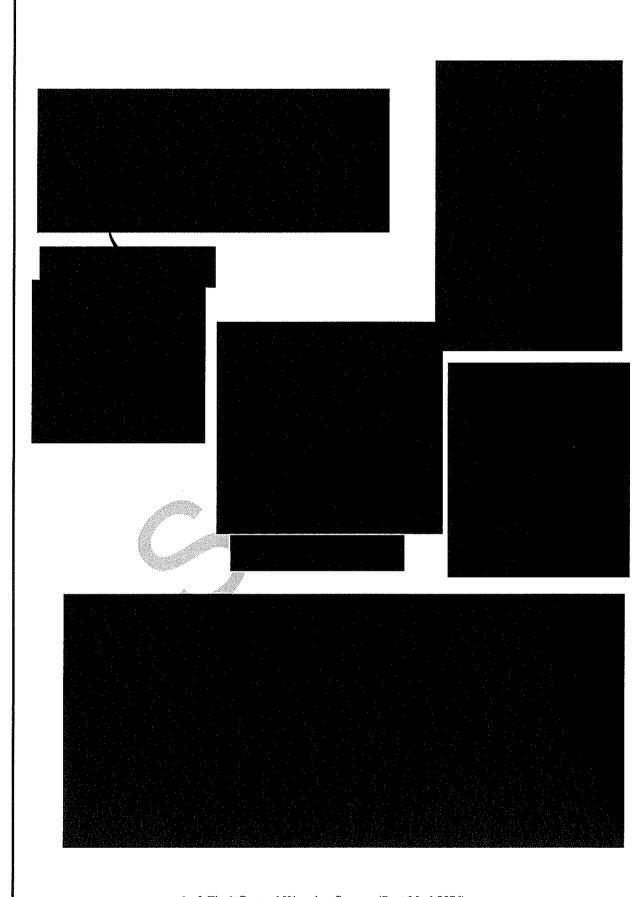
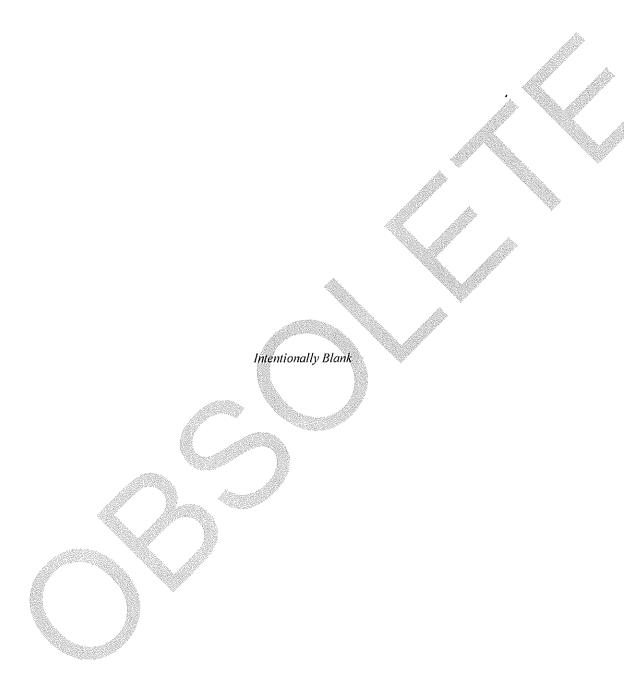


Table 1 - Central Warning Panel





1 - 2 Fig 1 Central Warning System (Post Mod 5576)



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# PART 1

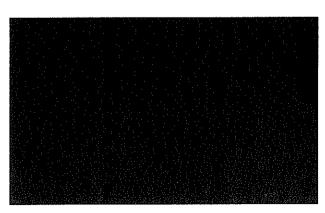
#### **CHAPTER 3 - FUEL SYSTEM**

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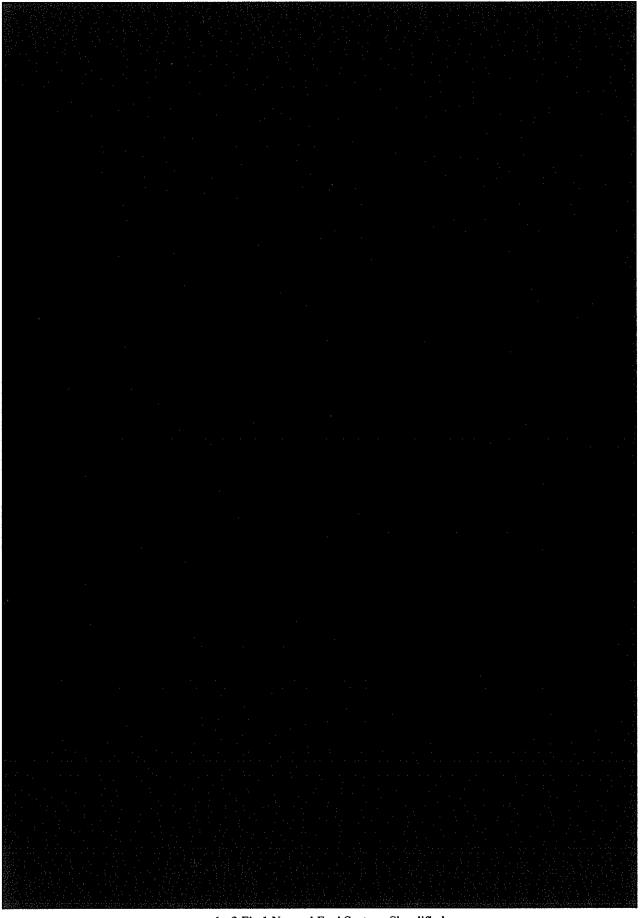
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Normal Fuel Syste		T	 atau aad	Land	 1 W.		•••	3
Forward Tank Fue			ator and	LOW-L	evel wa	uning	•••	4
1650,000 Junio 2010,000,000	009000	•••	•••		•••	• • • •	• • •	5
Overload Fuel Tra	£20000	• • • •		• • •	•••	• • •	• • •	6
Fuel Dumping Sy		• • •	•••	•••	• • •	• • •	* * *	=
Refuel/Defuel/Co	965966967	• • • •	•••	• • •	•••	• • •	• • •	7
Pressure Refuellin	-	• • • •	•••	• • •	• • •	• • • •	•••	8
Suction Defuelling	-	• • •	***	• • • •	***	• • •	• • •	9
Tanks Venting Sy	stem	•••	***	• • • •	• • • •	•••	• • •	10

#### GENERAL

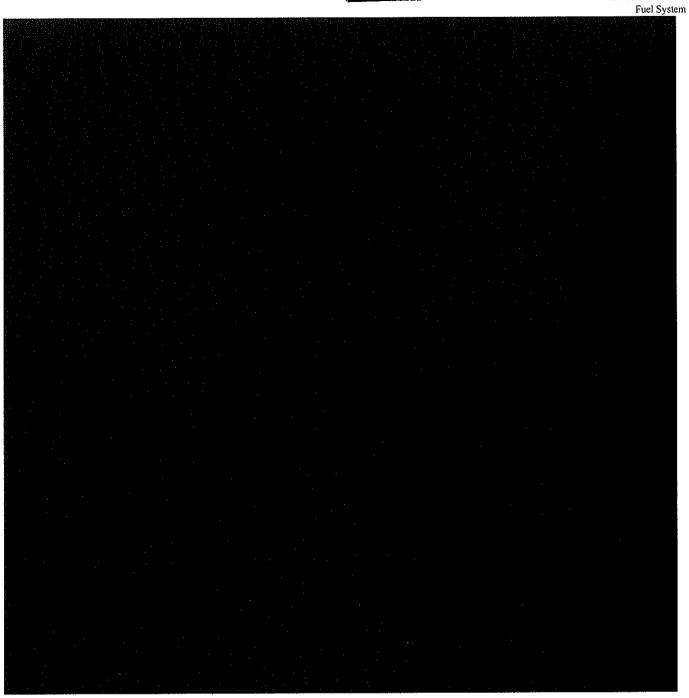
#### Introduction





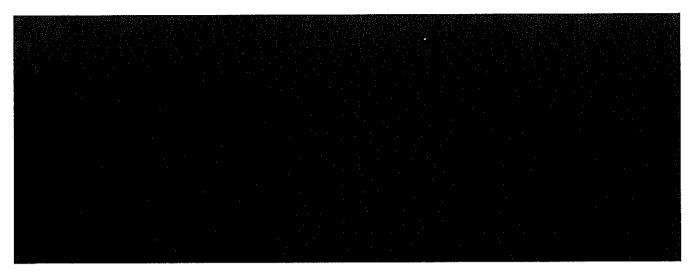


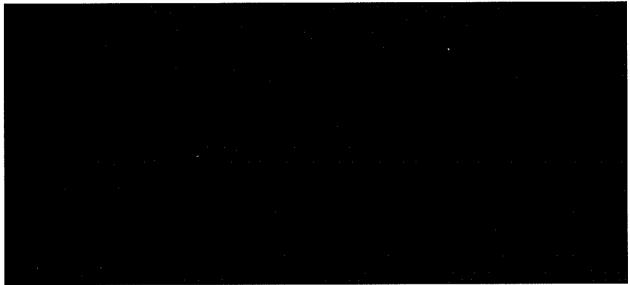
1 - 3 Fig 1 Normal Fuel System, Simplified



1 - 3 Fig 2 Fuel Panel







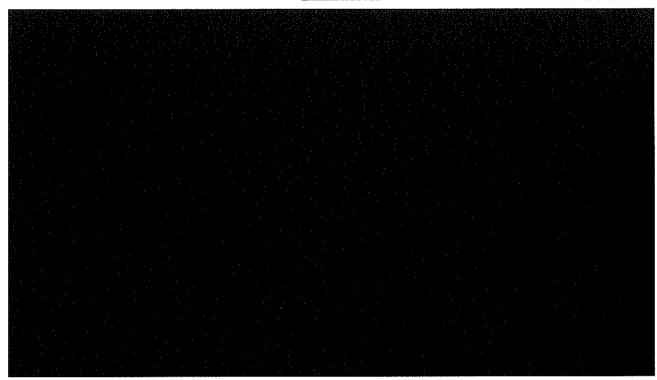
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# 1 - 3 Fig 3 Normal Fuel System Contents Indicator and Low-Level Warning

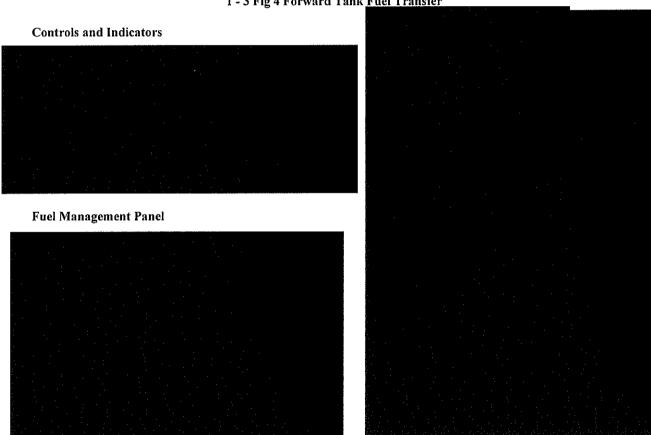
Table 1 - Fuel Tank Capacities

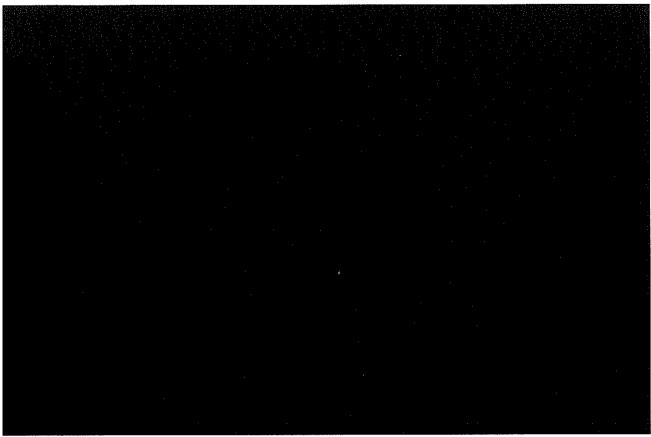
	Tank	Litres	kg (0.815 SG)	Gallons
Normal System				
	No 1 collector			
	No 2 collector	*		
	No 1 main			
	No 2 main			
"Virginia de Galeria de Carta	Forward			
	Total (gravity)			
	Total (pressure)			
Overload System		<u> </u>		
AND THE PARTY OF T	Overload			
	Total (all tanks gravity refuelled)			

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1 - 3 Fig 4 Forward Tank Fuel Transfer

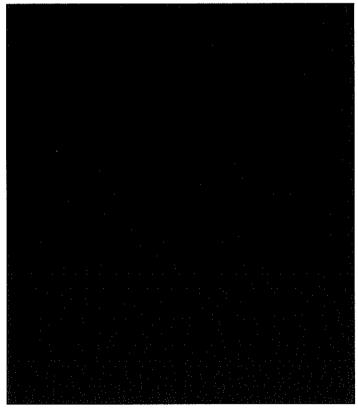


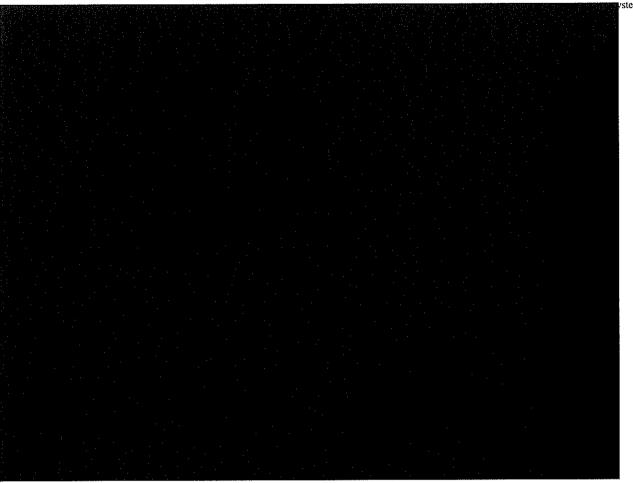


1 - 3 Fig 5 Overload Fuel Transfer

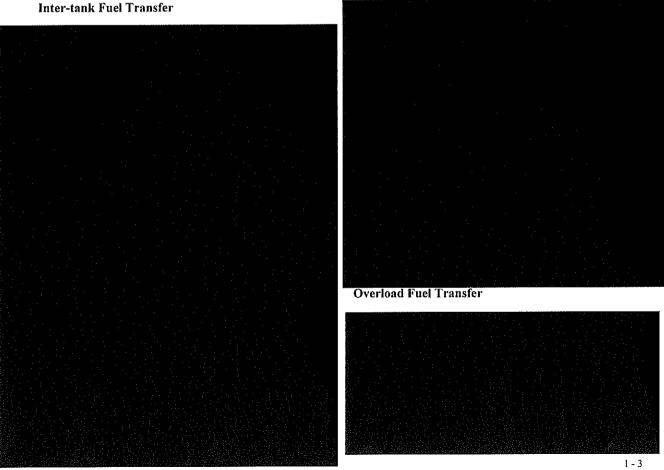
Normal System Contents and Low-level Warning

**Engine Fuel Supply** 

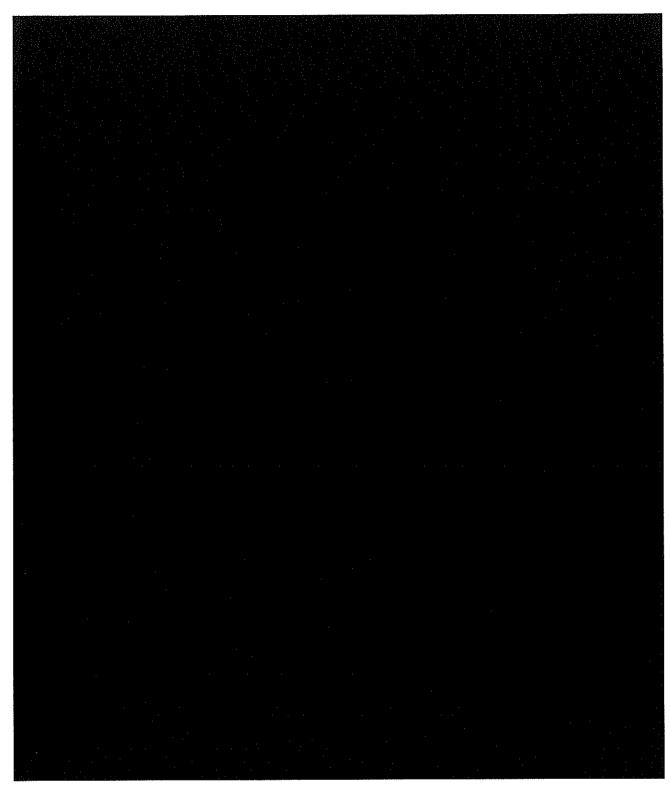




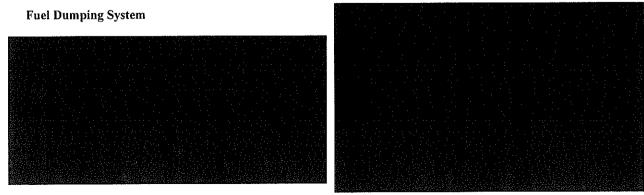
1 - 3 Fig 6 Fuel Dumping System



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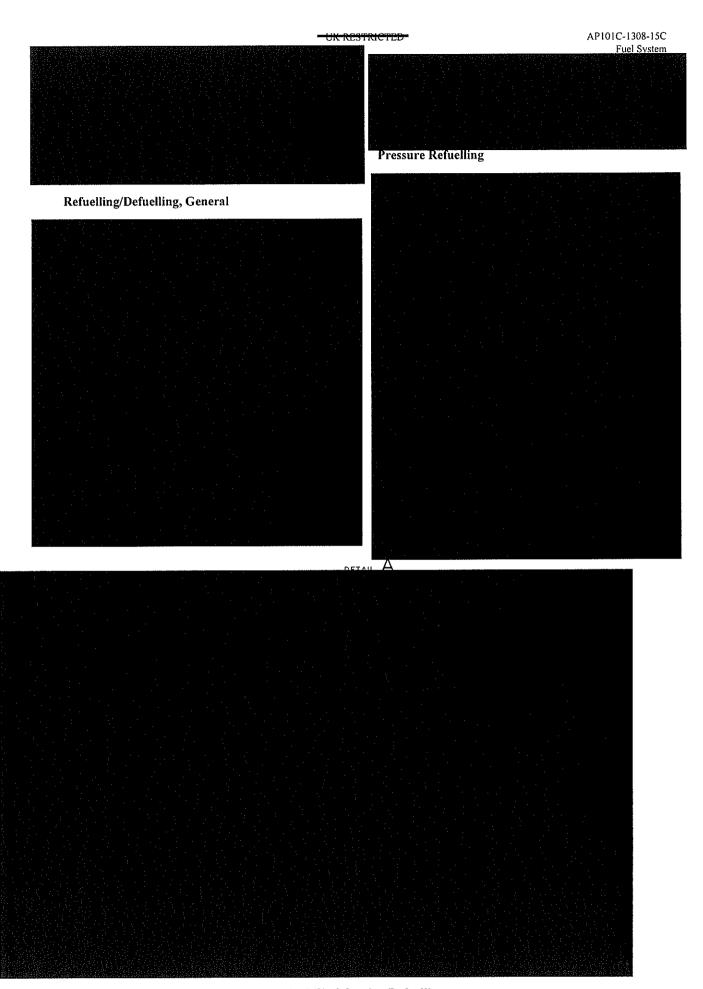


1 - 3 Fig 8 Pressure Refuelling

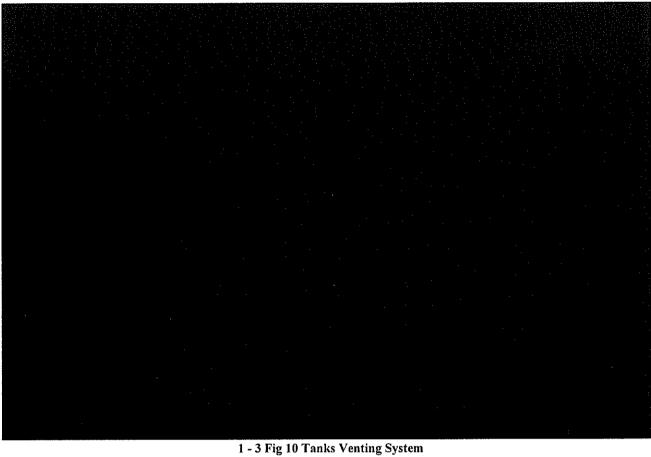


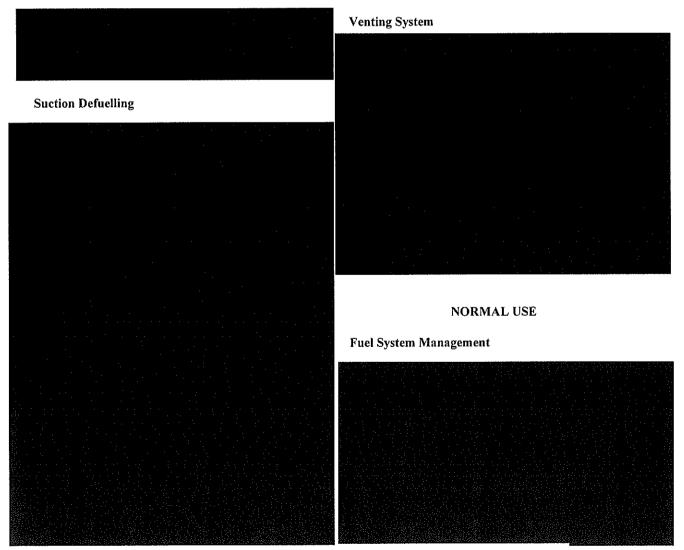
1 - 3 Page 8

OFFICIAL SENSITIVE



1 - 3 Fig 9 Suction Defuelling

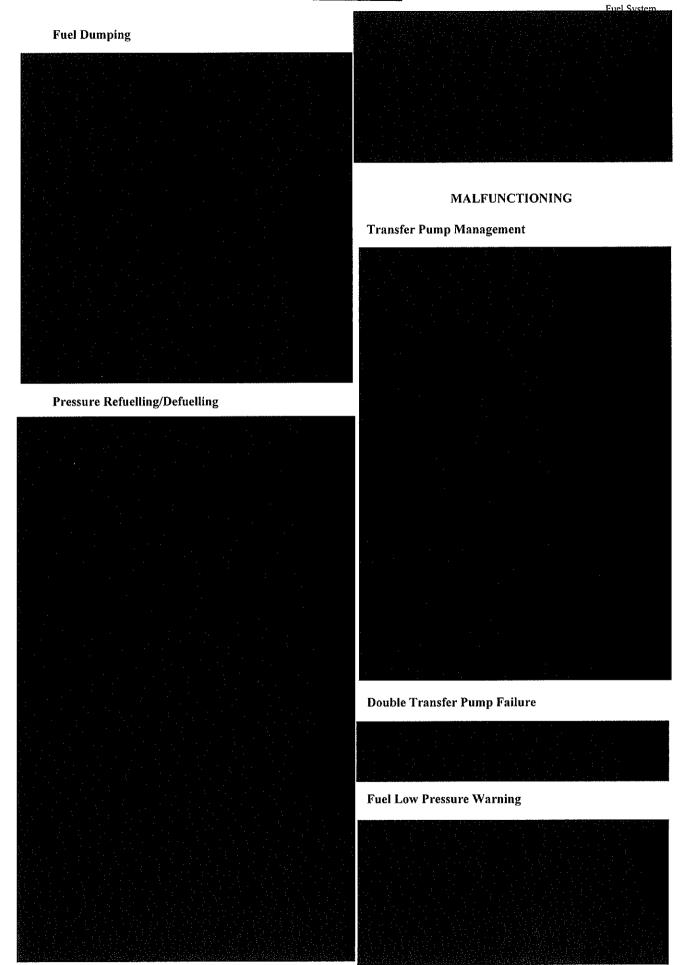




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# PART 1

# **CHAPTER 4 - ENGINE SYSTEMS**

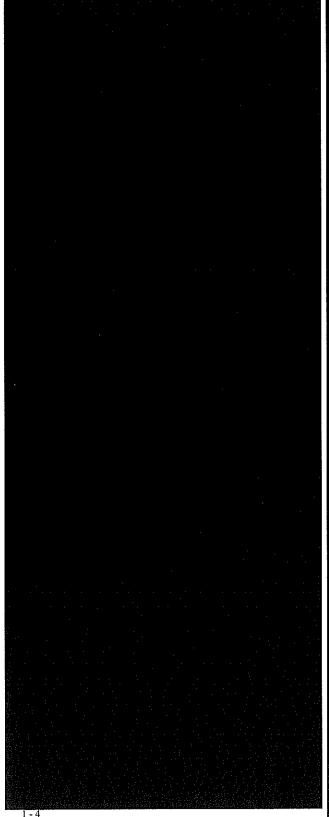
#### Contents

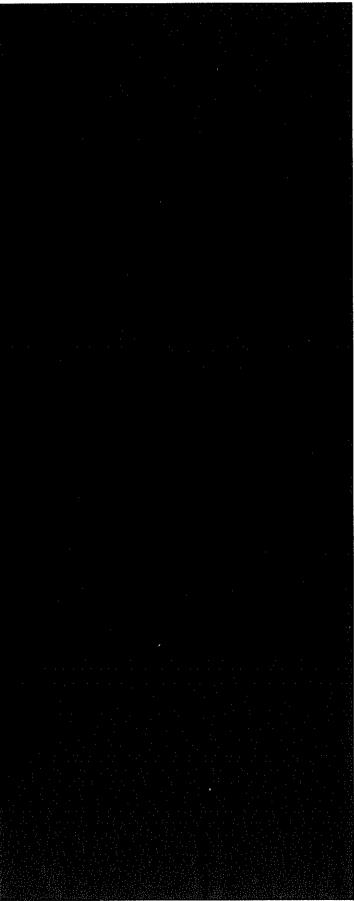
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GENERAL									
General		•••			•••	•••	•••	•••	1
FUEL CONTROL	SYSTE	EM							
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					9.5				
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Fuel Flow Char				•••	•••	•••	***		5
Throttle Contro			 atic	•••	•••	***	•••	•••	6
Gas Generator (					•••	•••	***		7
Power Turbine					***			•••	8
Torque Measuri			Jonetha		• • •				9
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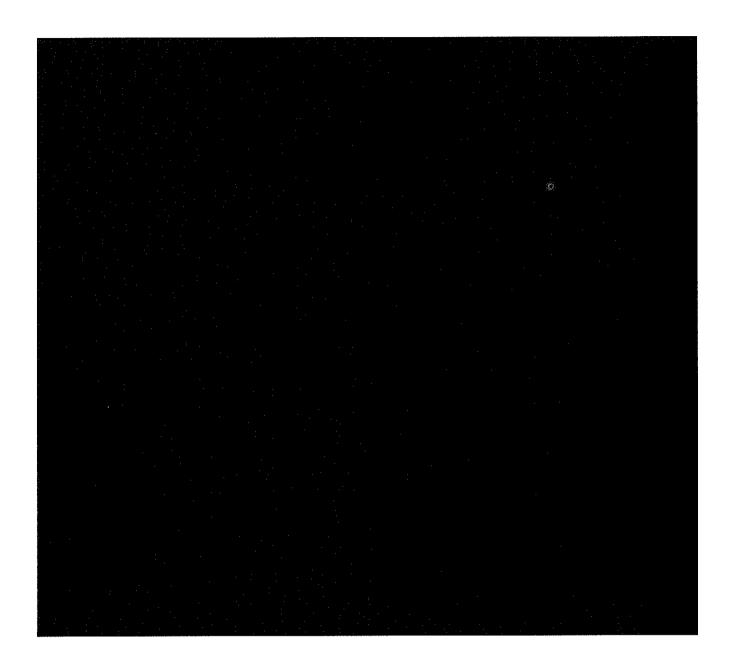
# **GENERAL**

#### General





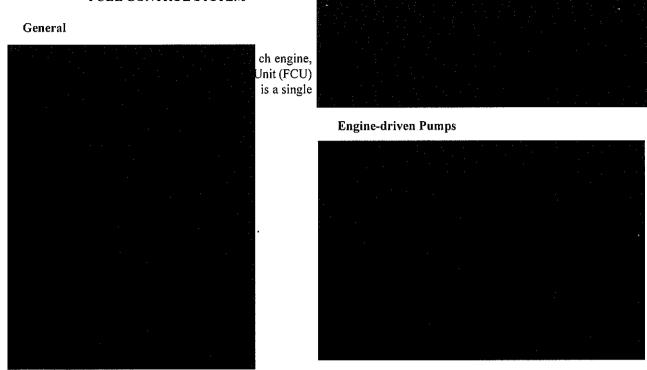
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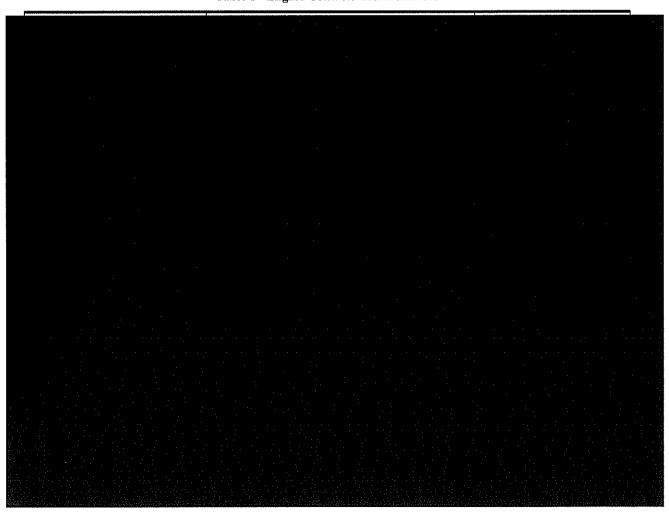
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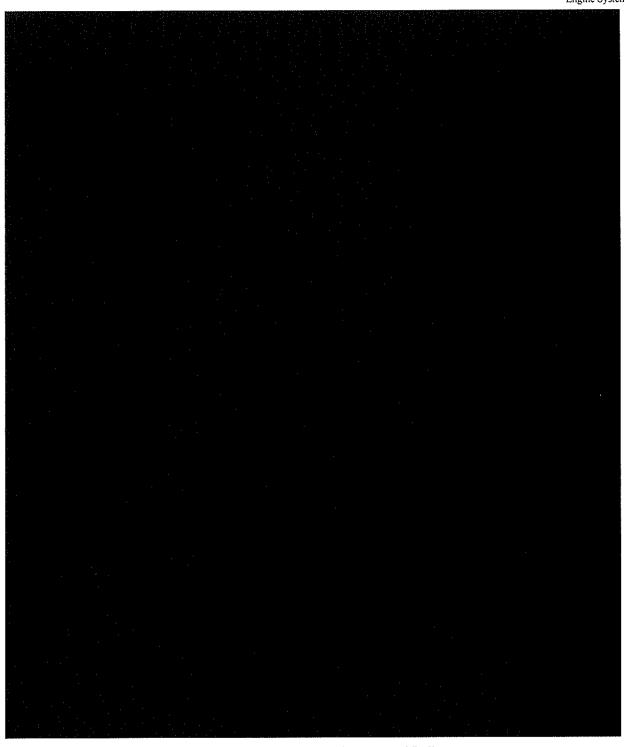
1 - 4 Fig 1 Gem Mk 204/205 Engine

# FUEL CONTROL SYSTEM

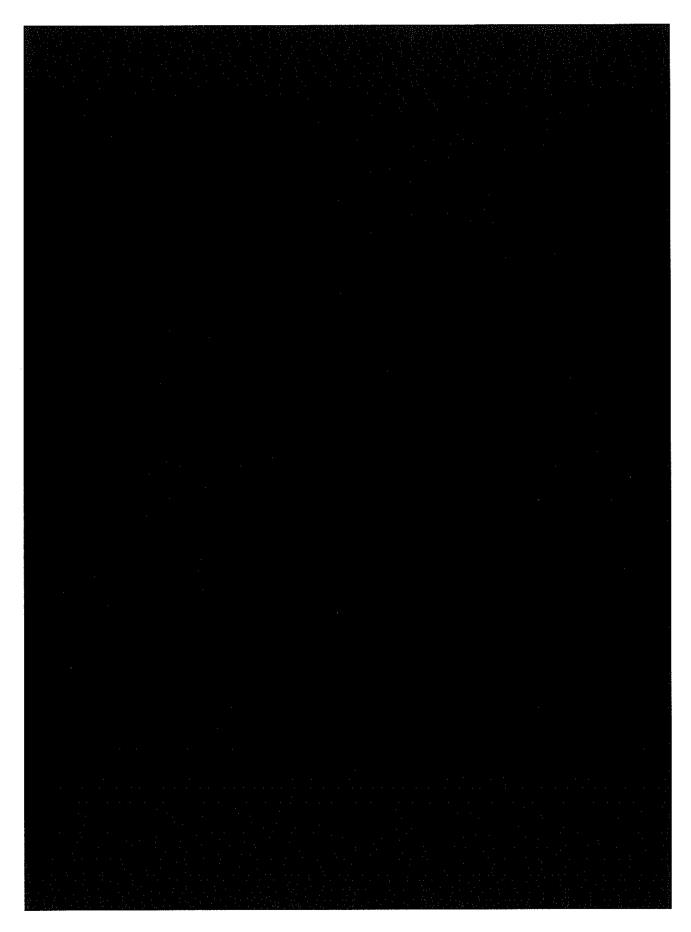


**Table 1 - Engine Controls and Indicators** 

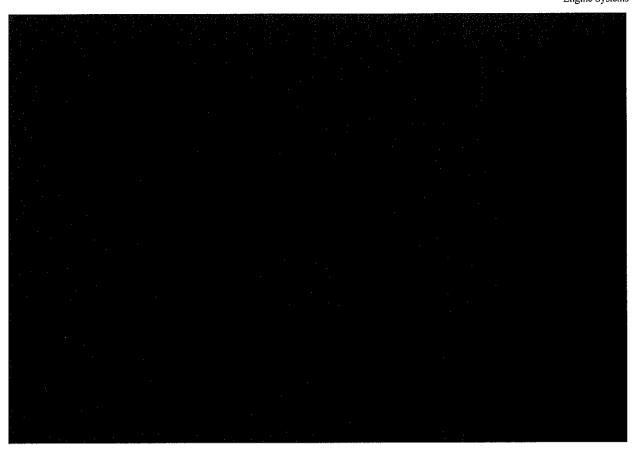


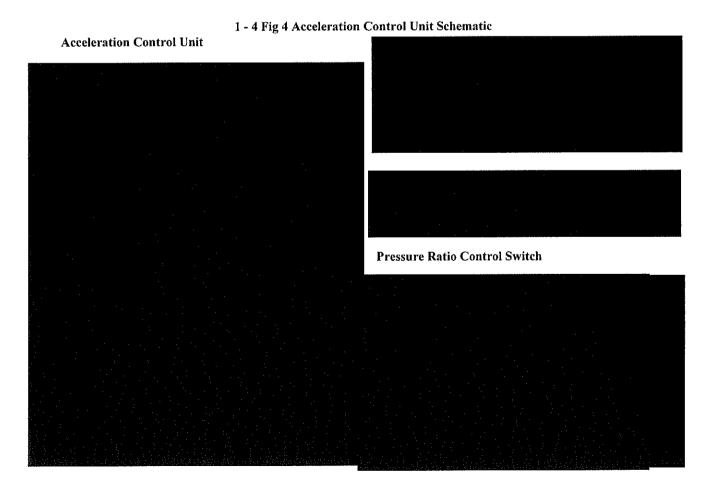


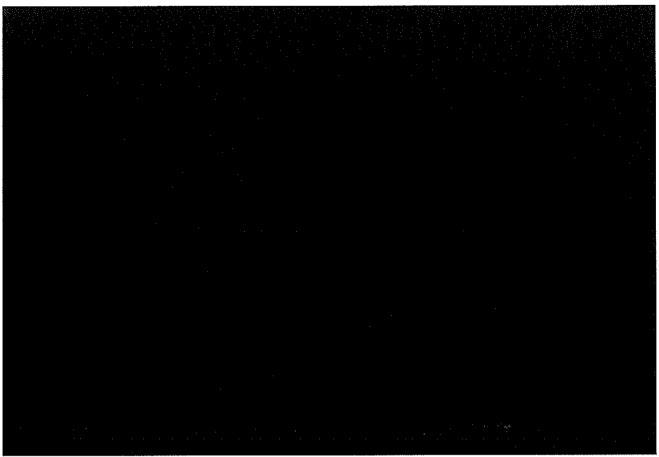
1 - 4 Fig 2 Engine Systems Controls and Indicators



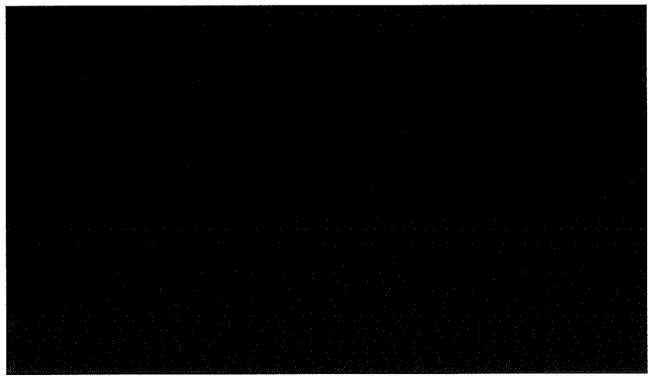
1 - 4 Fig 3 Fuel Control System Schematic



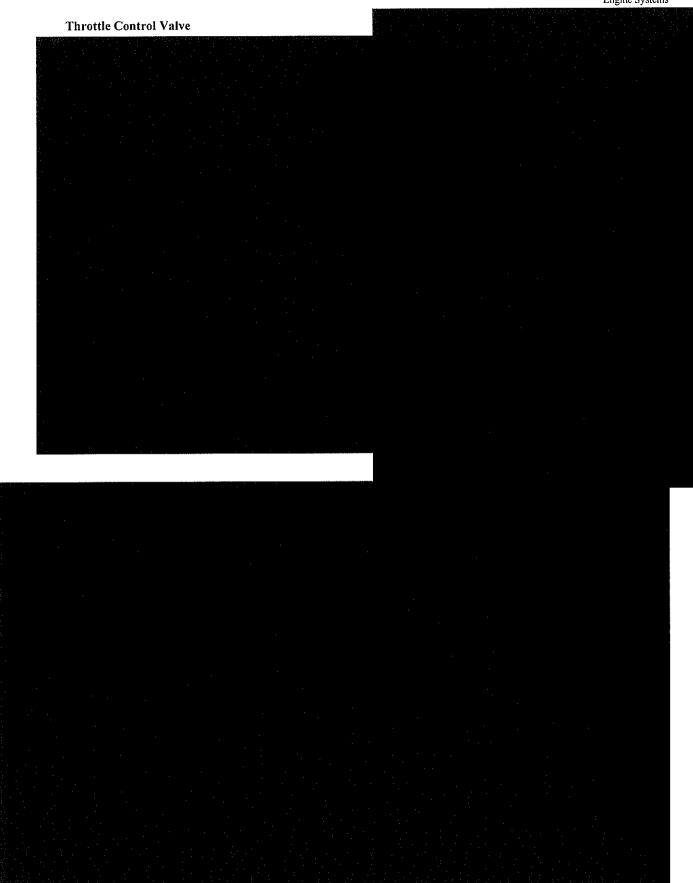




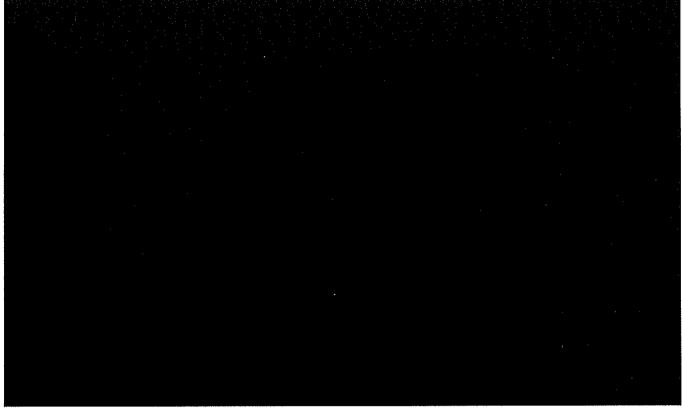
1 - 4 Fig 5 Fuel Flow Characteristics



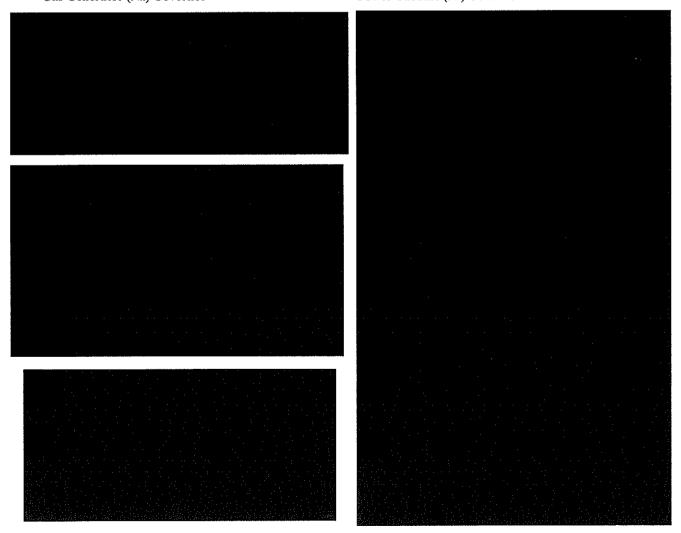
1 - 4 Fig 6 Throttle Control Valve Schematic



1 - 4 Fig 7 Gas Generator (NH) Governor Schematic

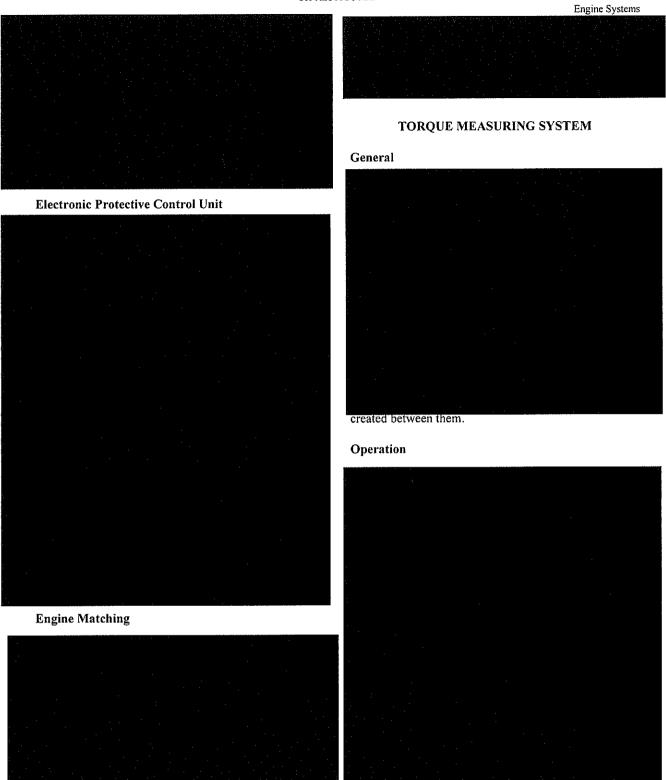


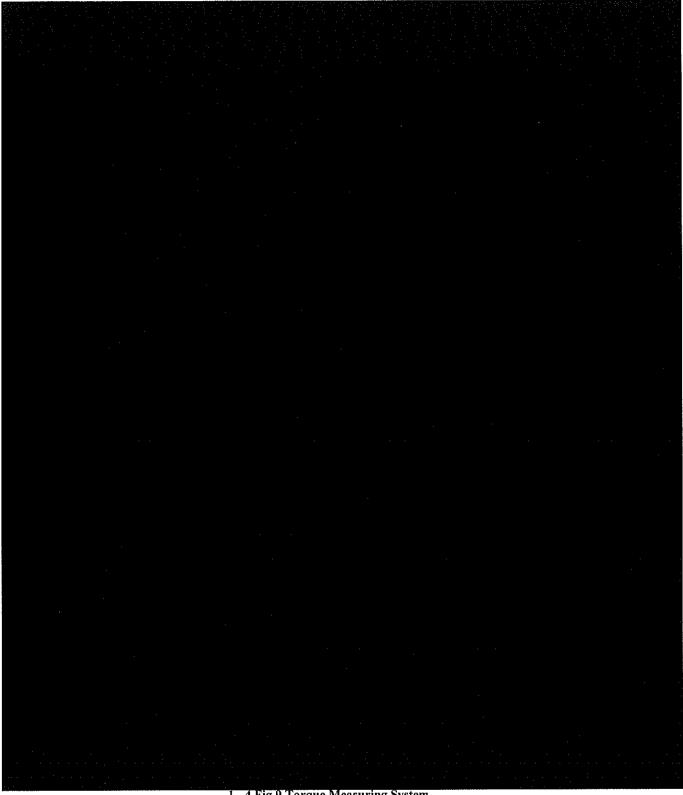
1 - 4 Fig 8 Power Turbine (Nf) Governor Schematic
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1 - 4 Fig 9 Torque Measuring System

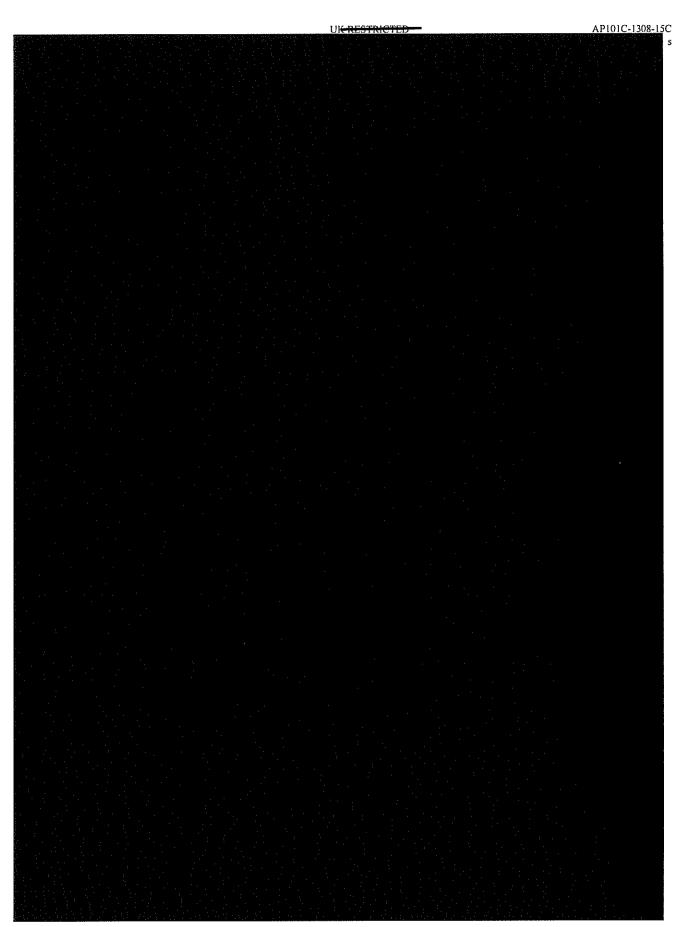
# **ENGINE LUBRICATION SYSTEM**

General



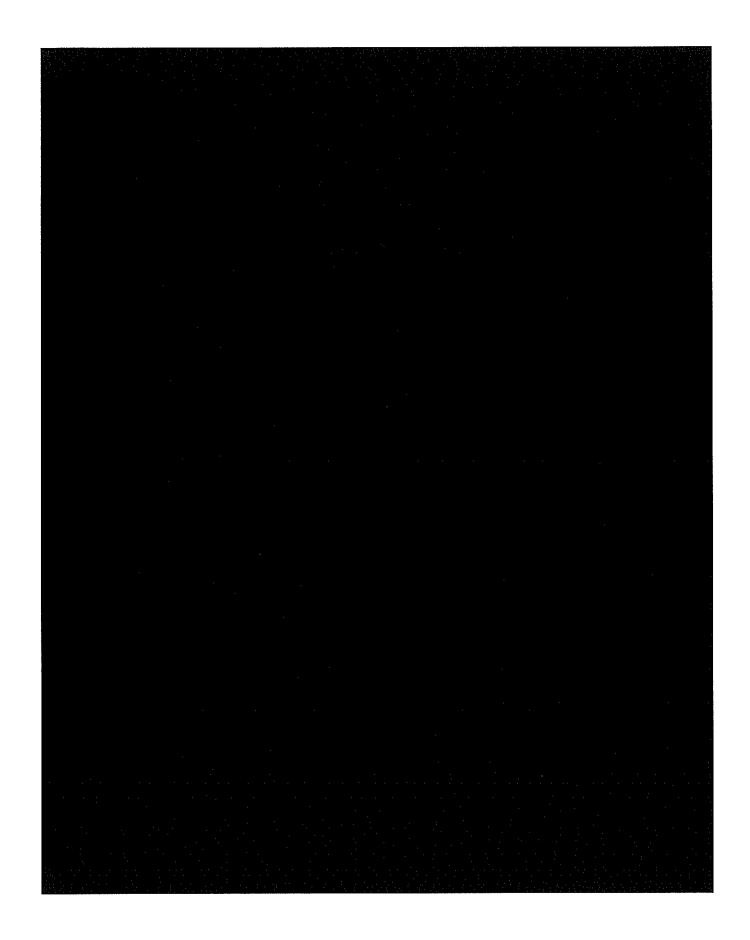


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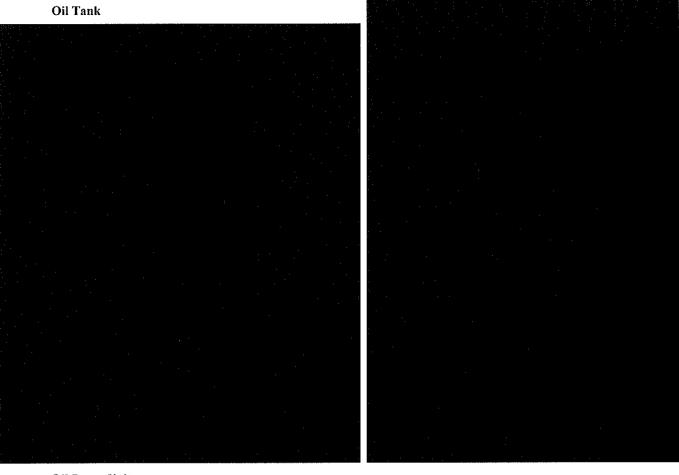


1 - 4 Fig 10 Oil Flow Schematic (Gem Mk 204)

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1 - 4 Fig 11 Oil Flow Schematic (Gem Mk 205)



Oil Pump Unit

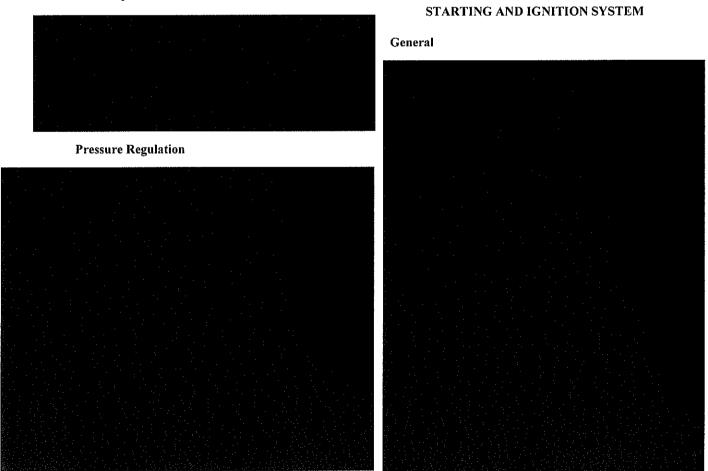
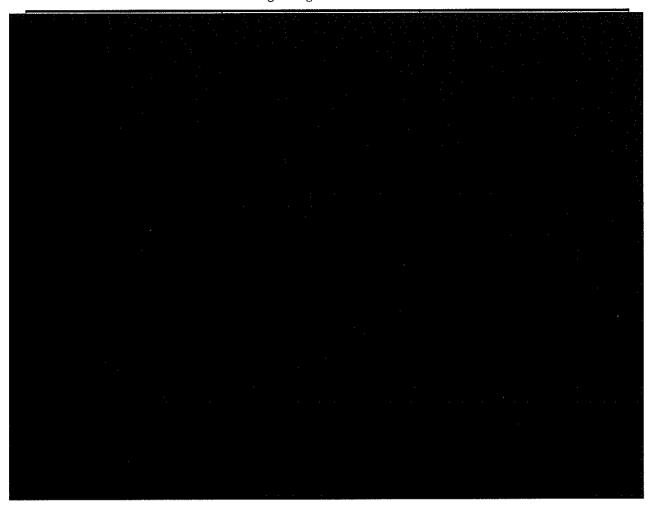
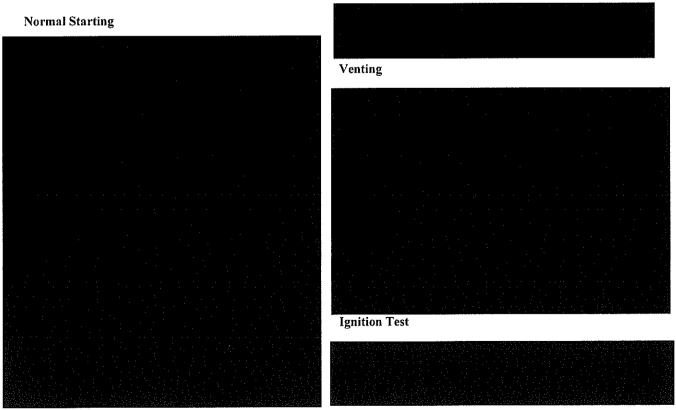


Table 2 - Starting and Ignition Controls and Indicators

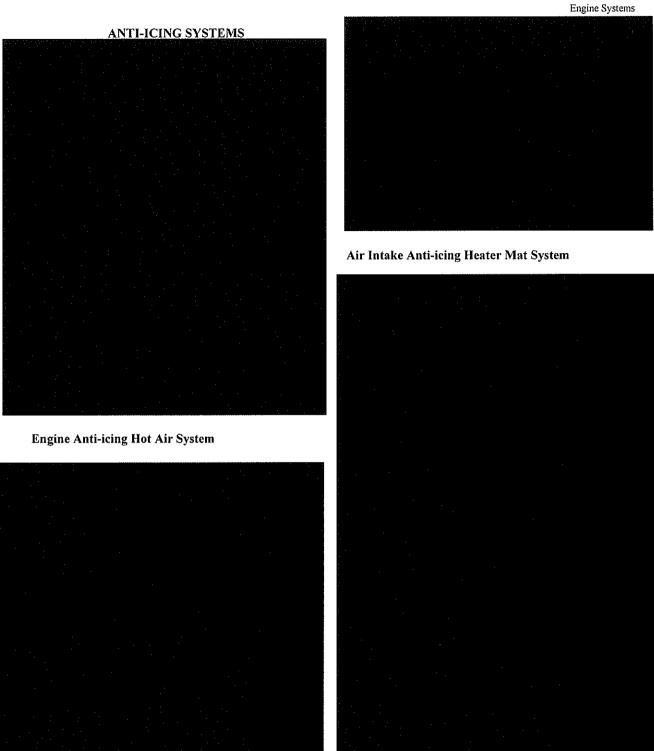


1 - 4 Fig 12 Starting and Ignition Electrical Circuits



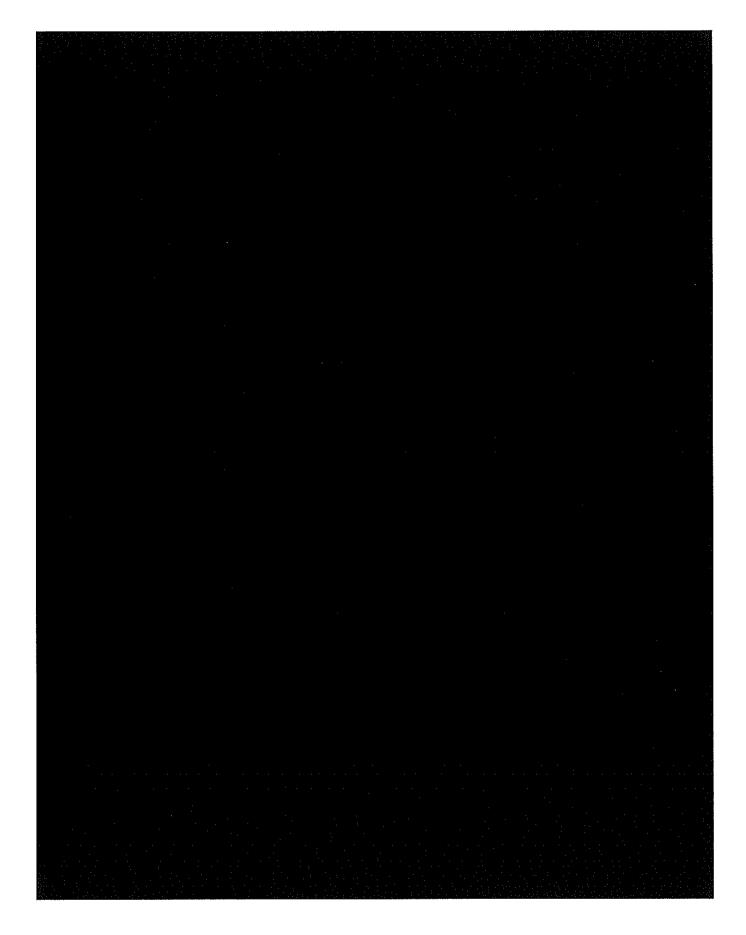
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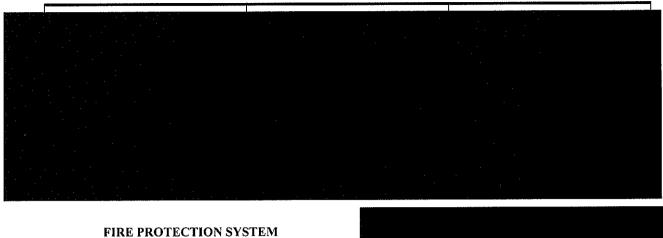
**Table 3 - Anti-icing Systems Controls and Indicators** 

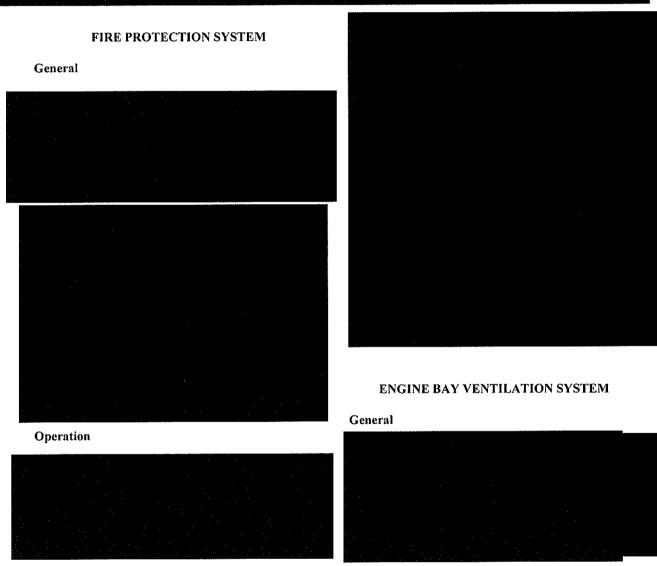


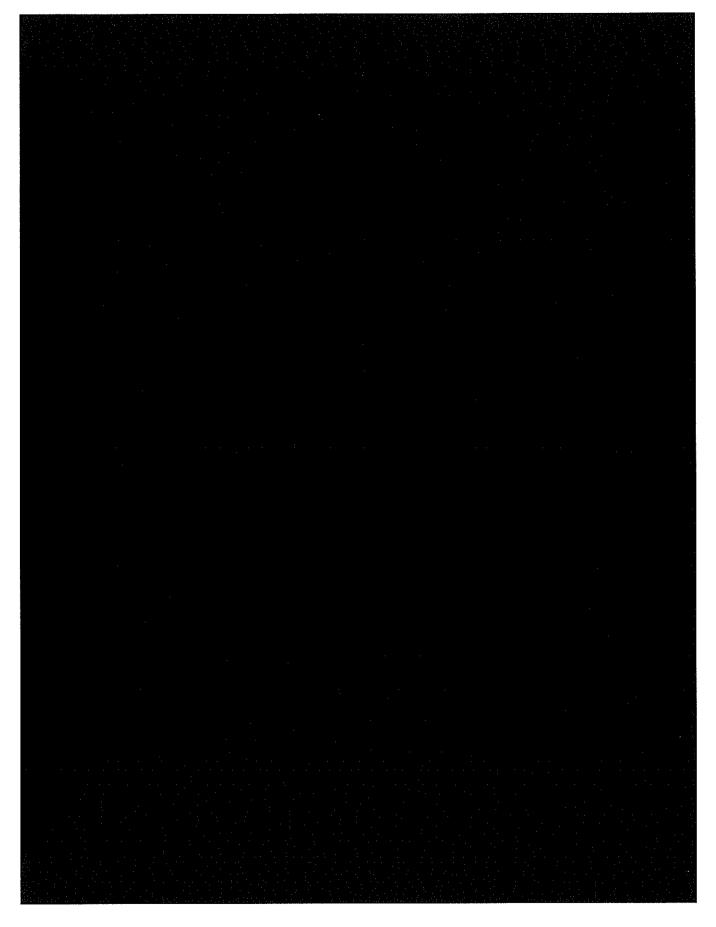


1 - 4 Fig 13 Engine Anti-icing Systems

**Table 4 - Engine Fire Protection System Controls and Indicators** 

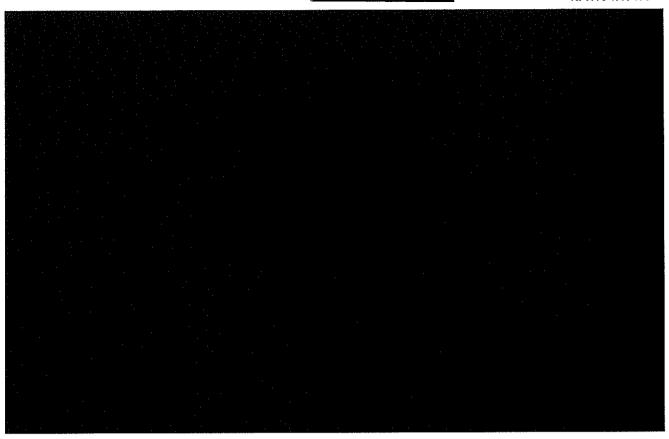


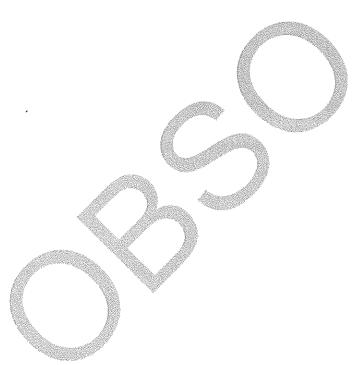




1 - 4 Fig 14 Fire Protection Electrical Circuits

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### UK RESTRICTED

# PART 1

# CHAPTER 5 - ROTORS, TRANSMISSION AND FLYING CONTROLS

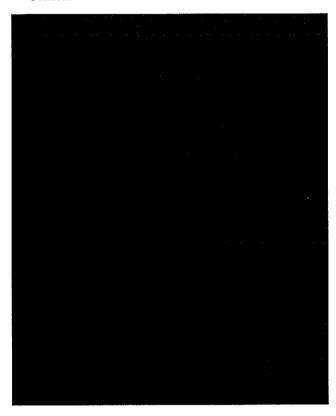
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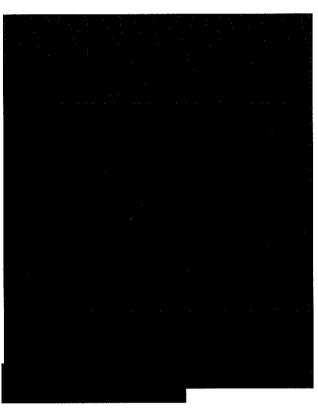
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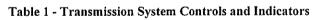
## **GENERAL**

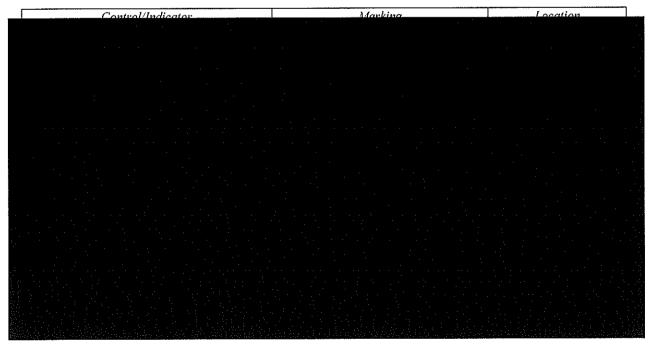
## General





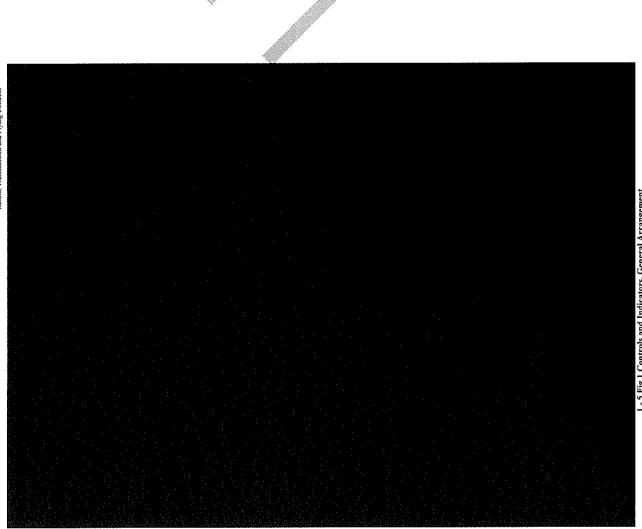
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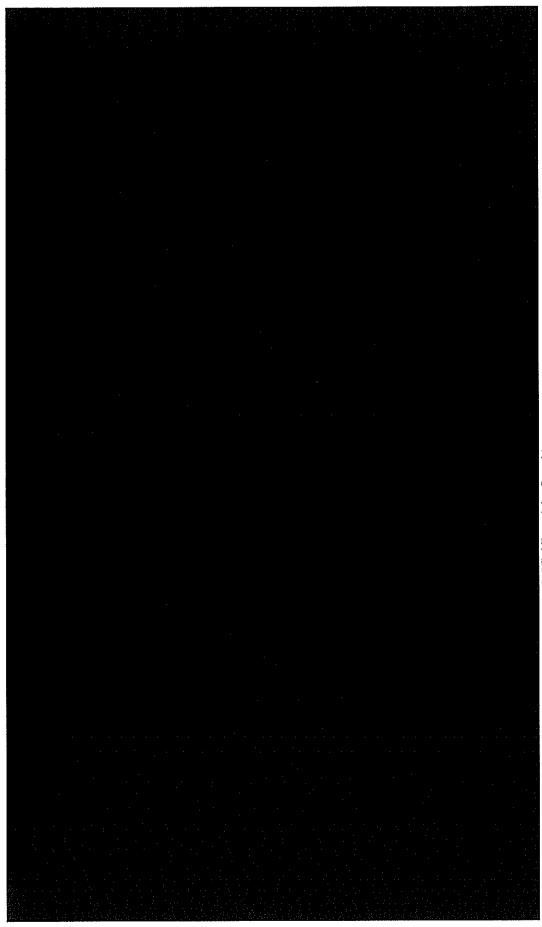


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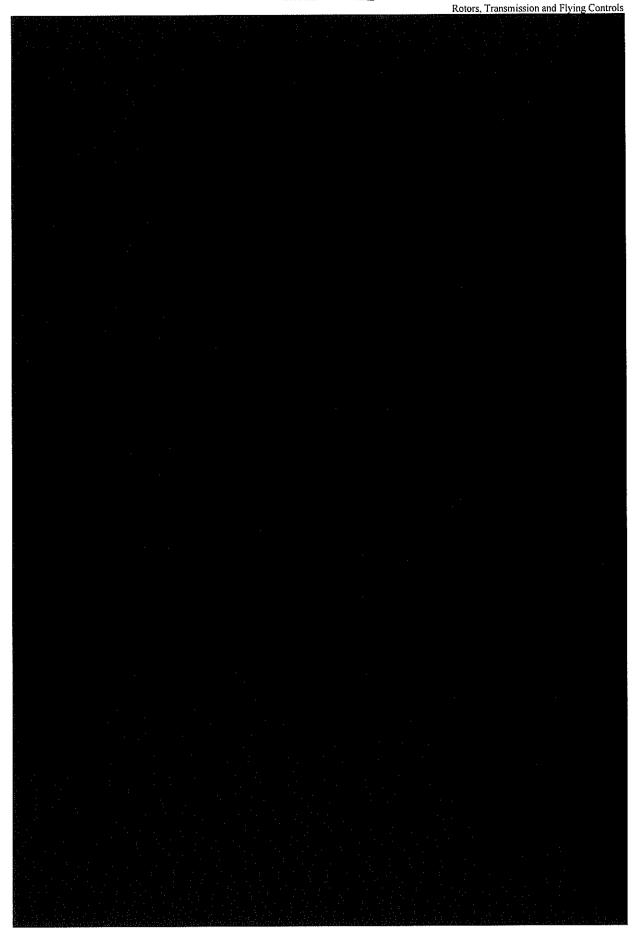
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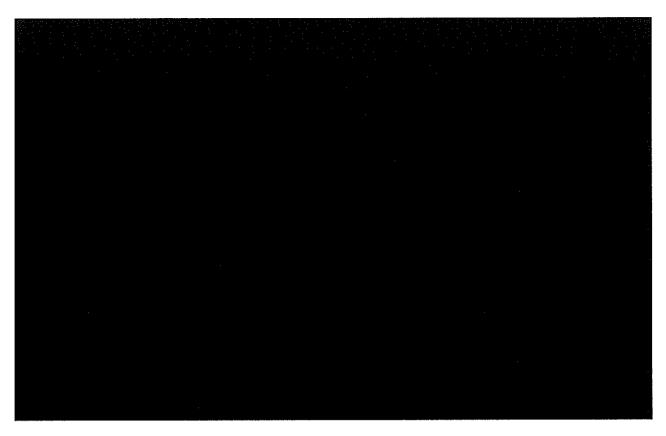
1-5 Fig 1 Controls and Indicators, General Arrangement



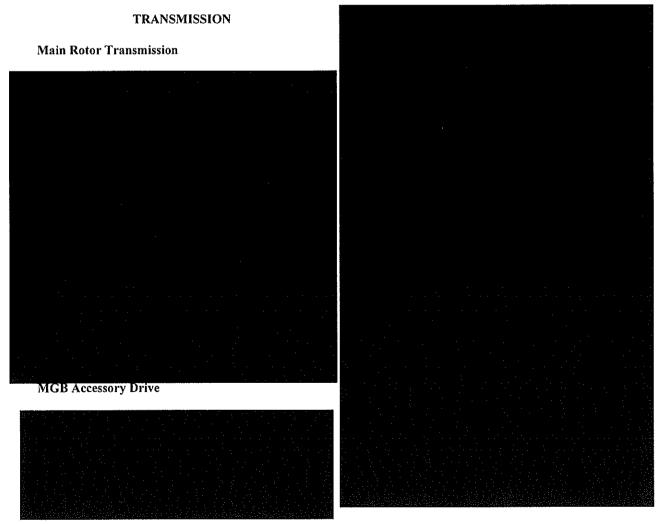
1 - 5 Fig 2 Transmission, General Arrangement



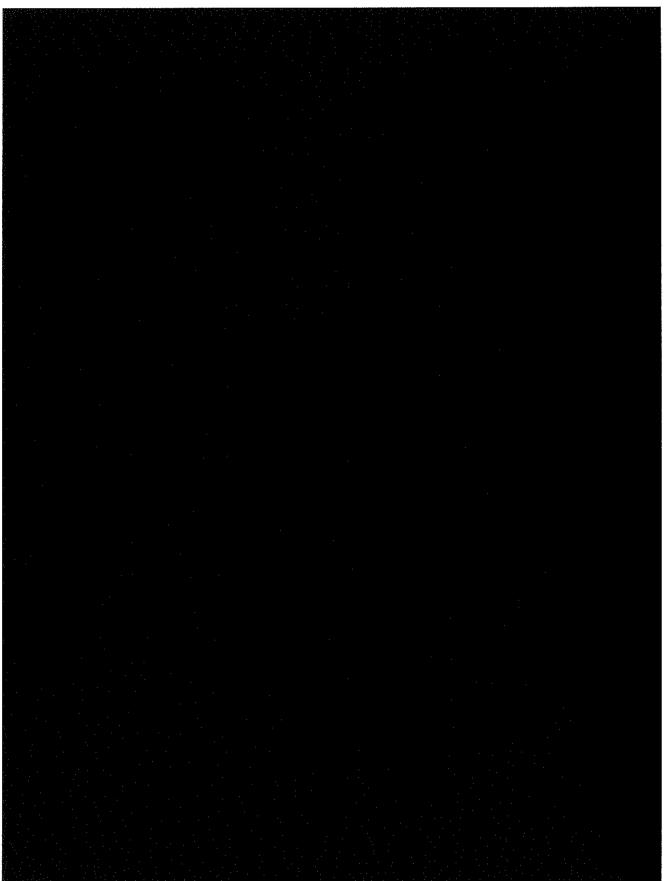
1 - 5 Fig 3 Transmission Schematic



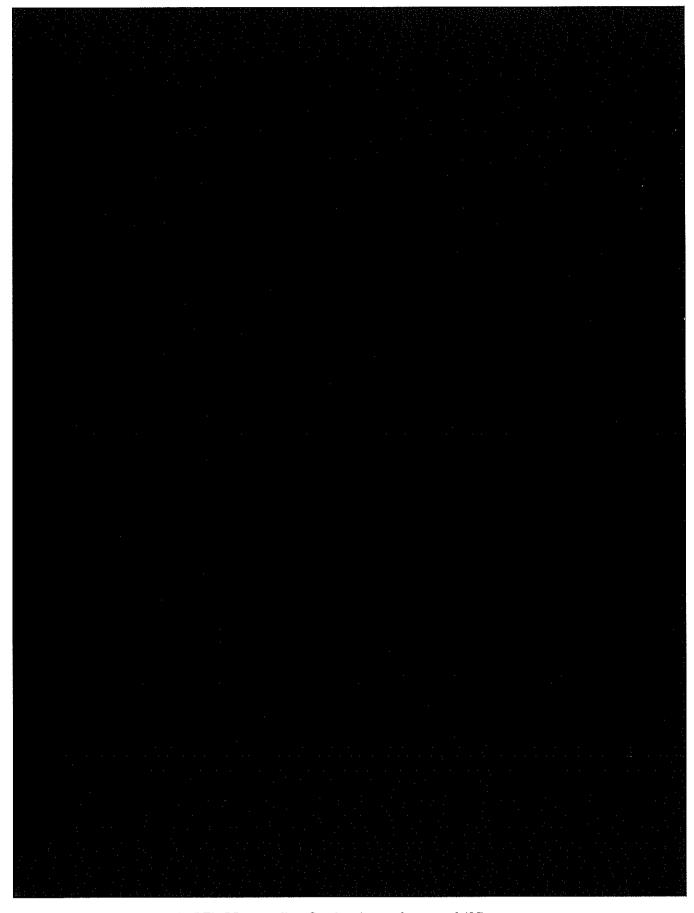
1 - 5 Fig 4 Accessory Drive Safety Interlocks



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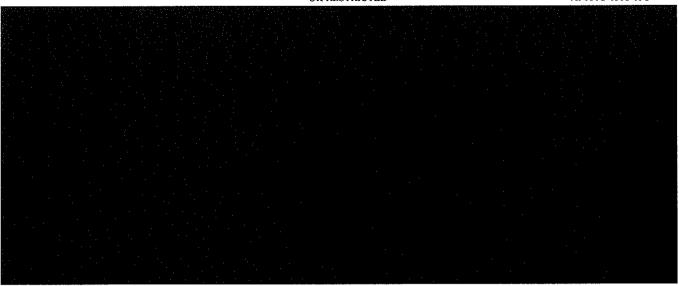


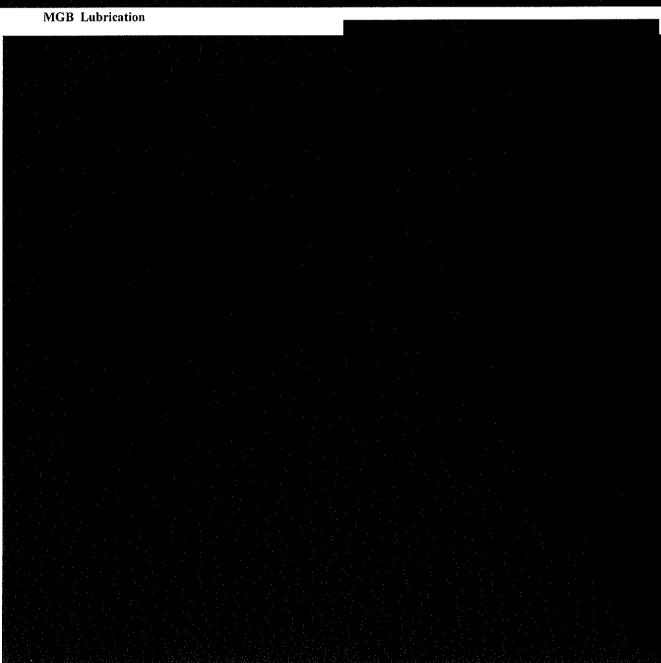
1 - 5 Fig 5 MGB Lubrication



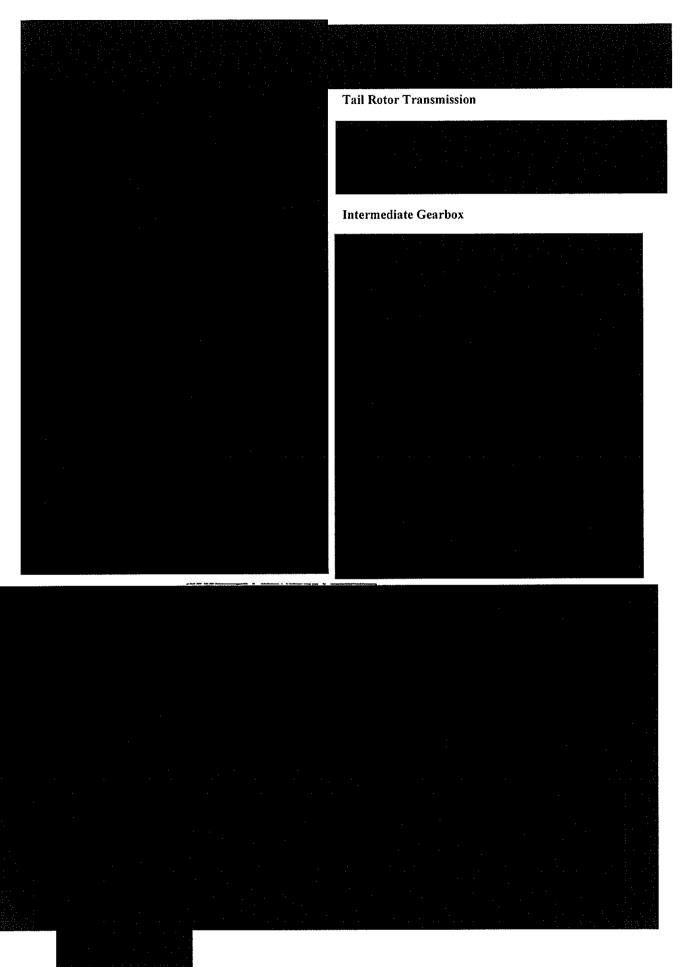
1 - 5 Fig 7 Intermediate Gearbox (pre and post-mod 495)

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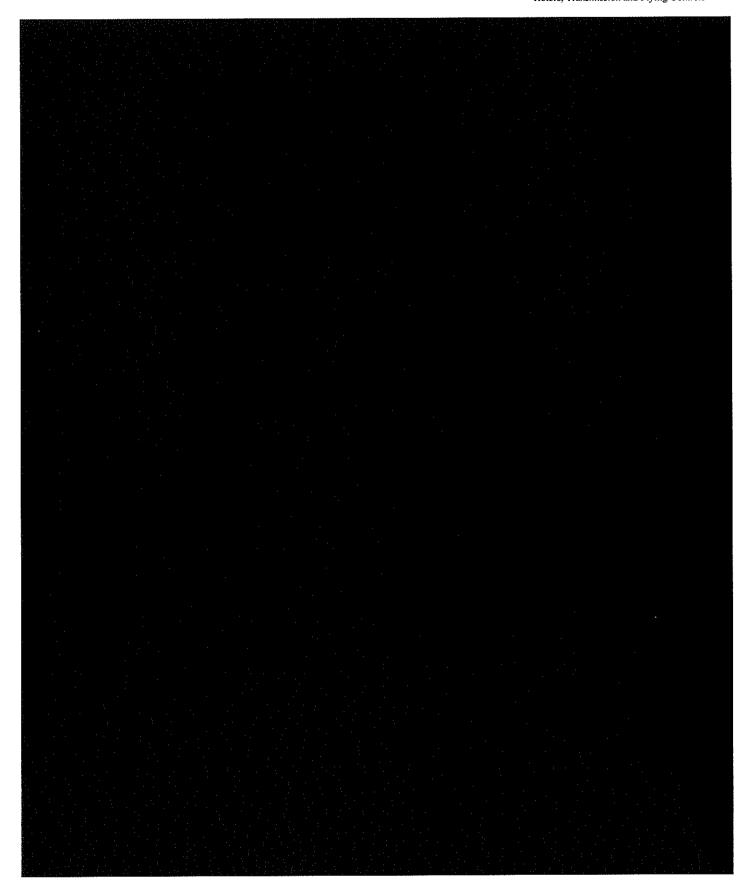




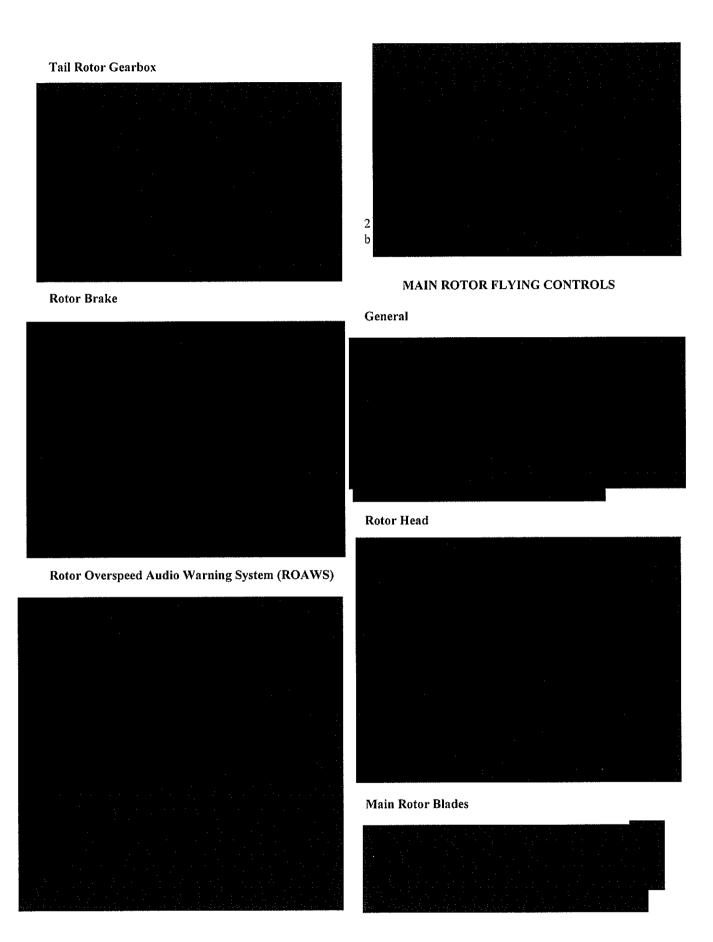
1 - 5 Fig 8 Tail Rotor Gearbox

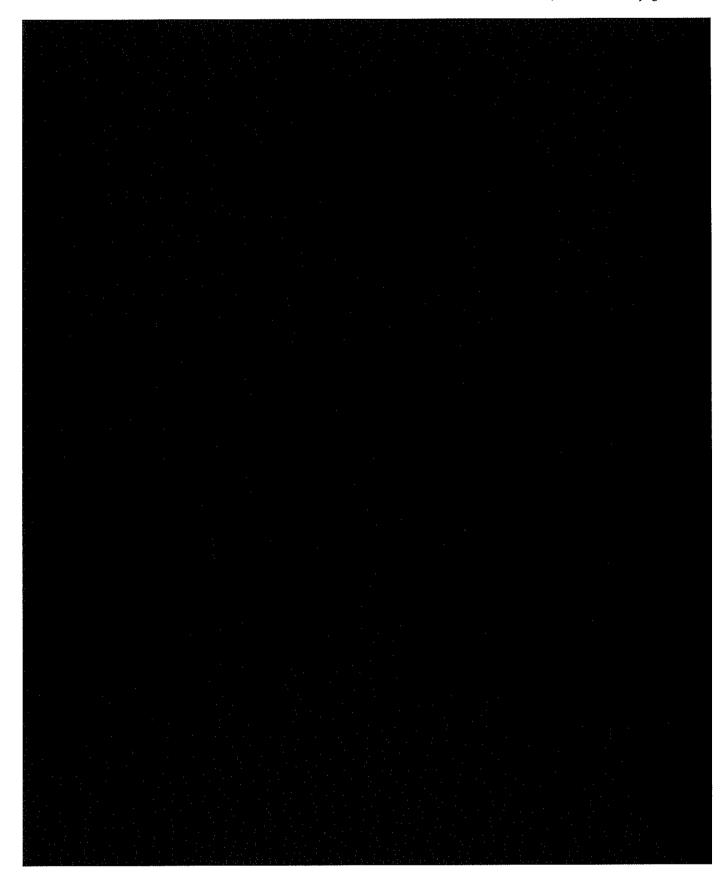


1 - 5 Fig 9 Rotor Brake

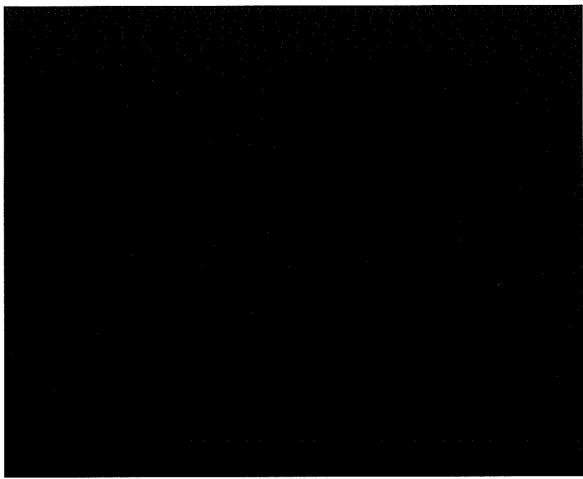


1 - 5 Fig 10 Main Rotor Assembly (Pre-mod 747A)

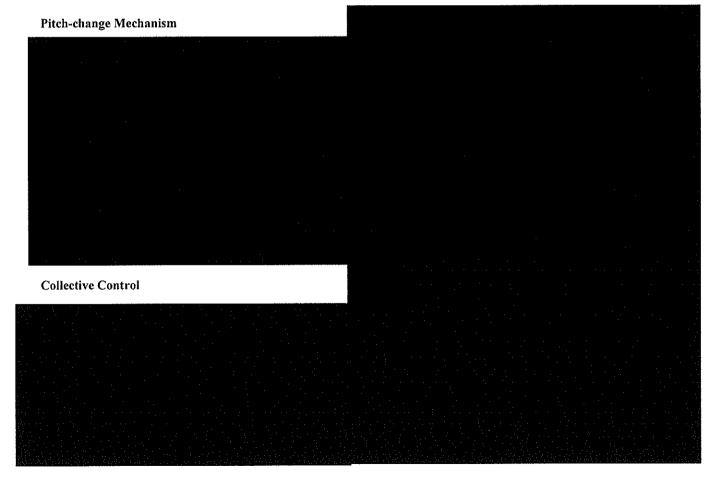


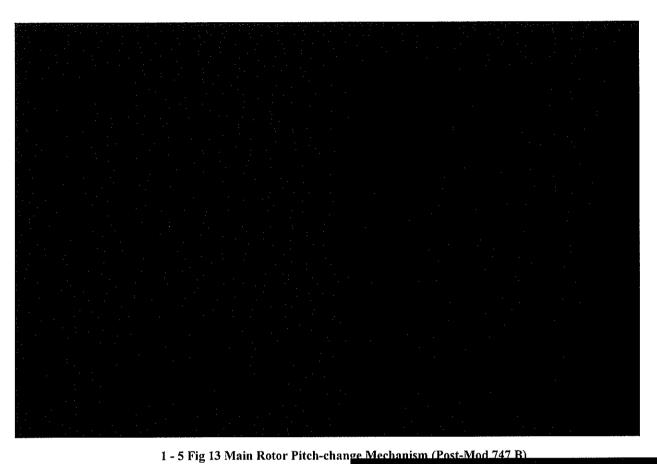


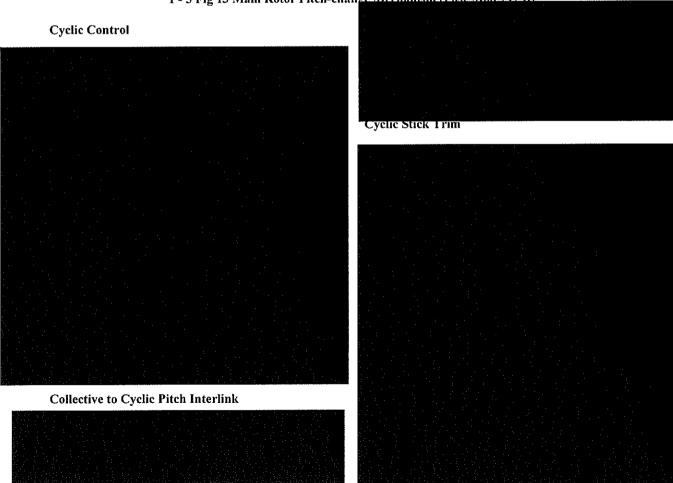
1 - 5 Fig 11 Main Rotor Assembly (Post-mod 747A)

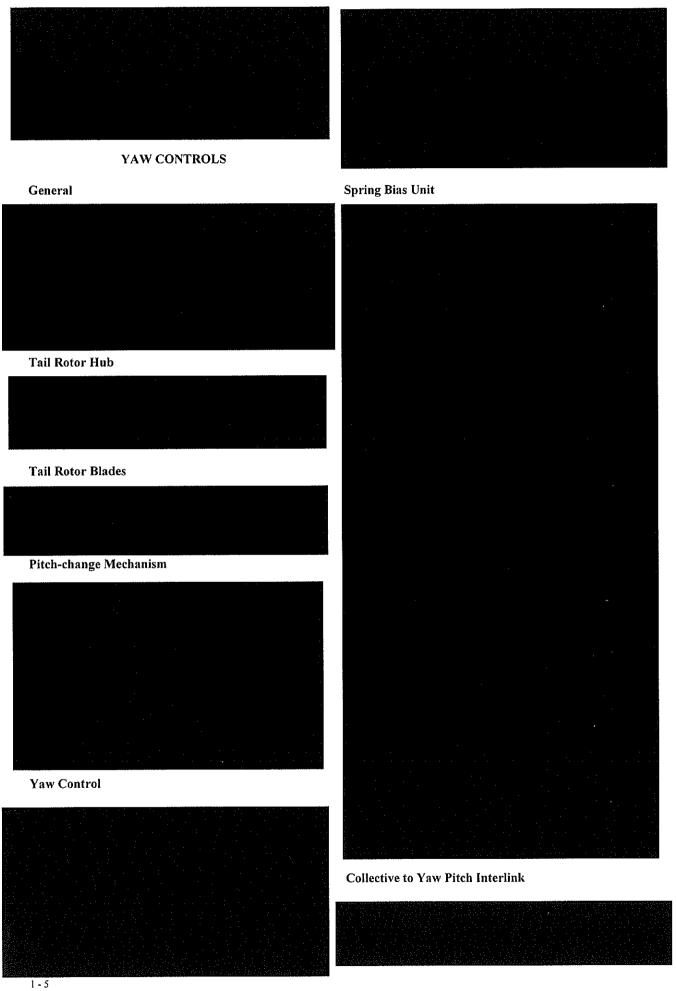


1 - 5 Fig 12 Main Rotor Pitch-change Mechanism (Pre-Mod 747 B)

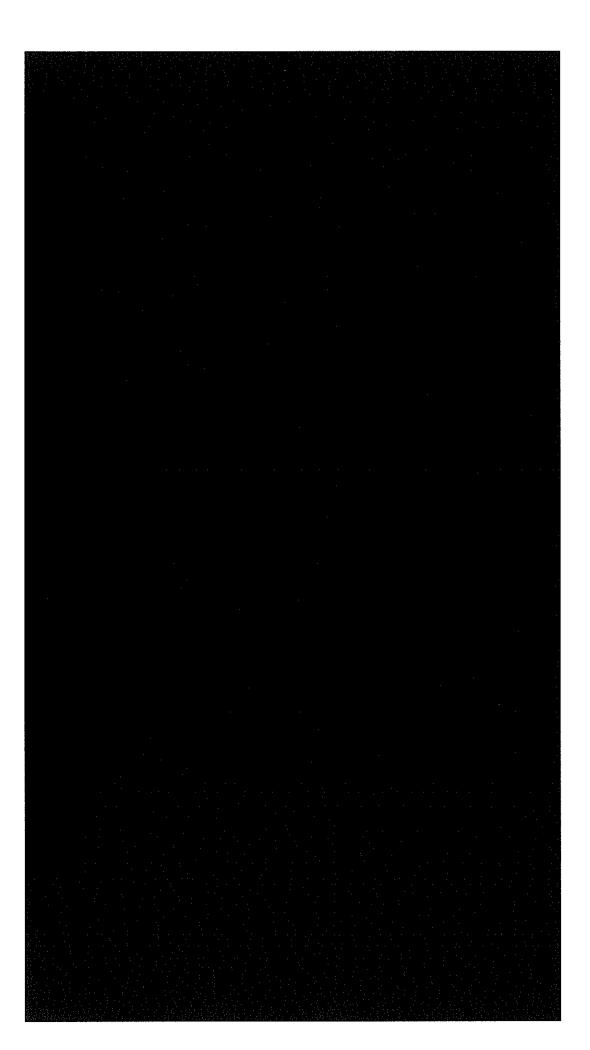




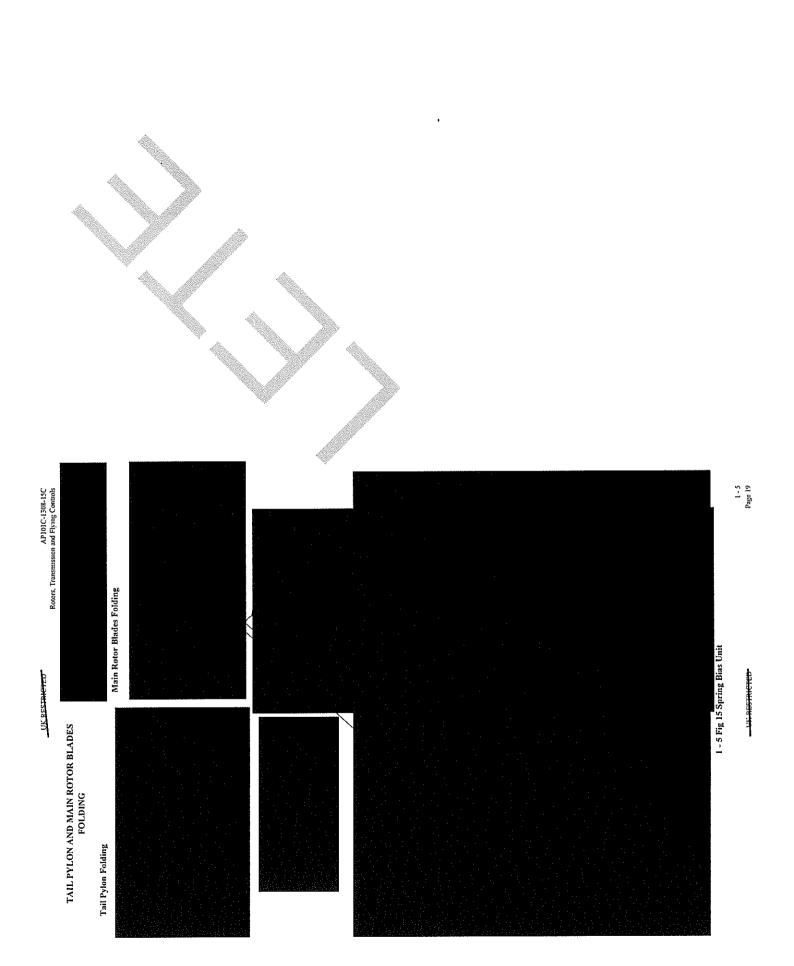




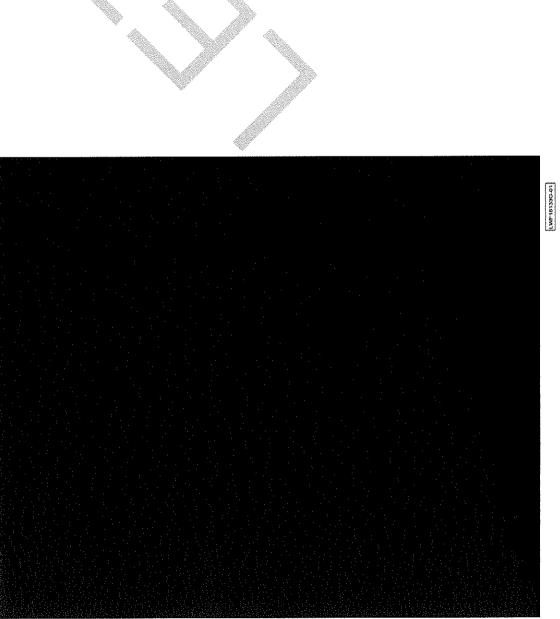
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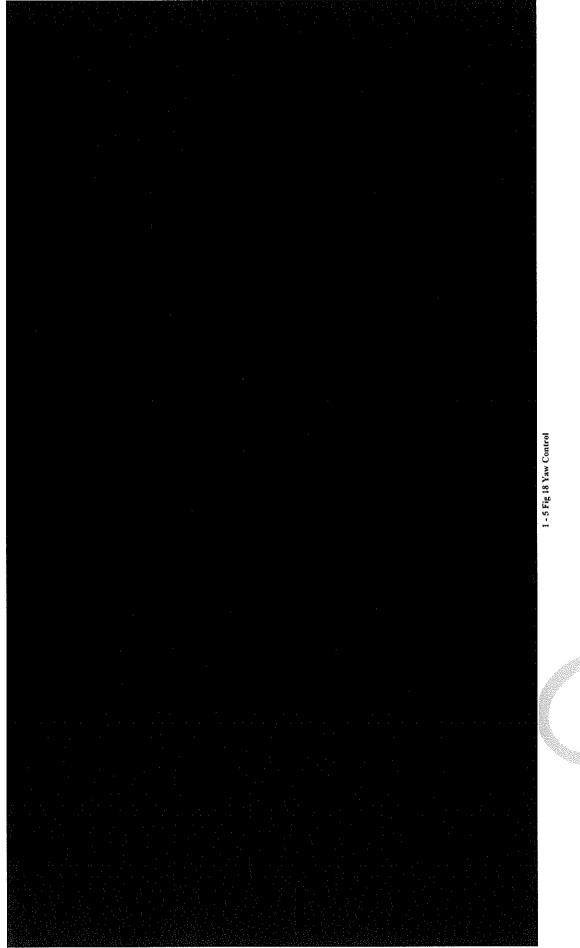


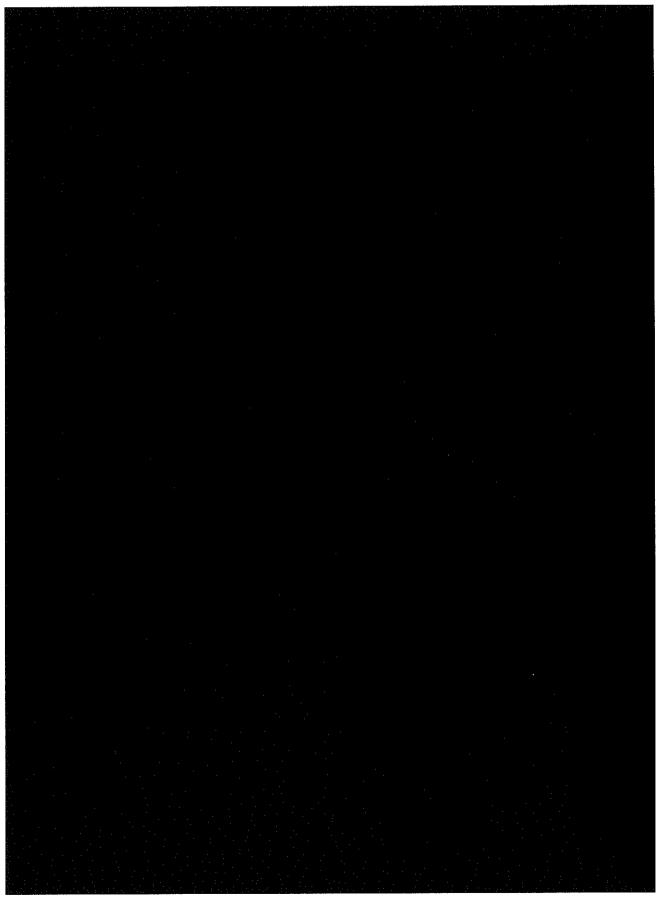
1 - 5 Fig 14 Collective Control



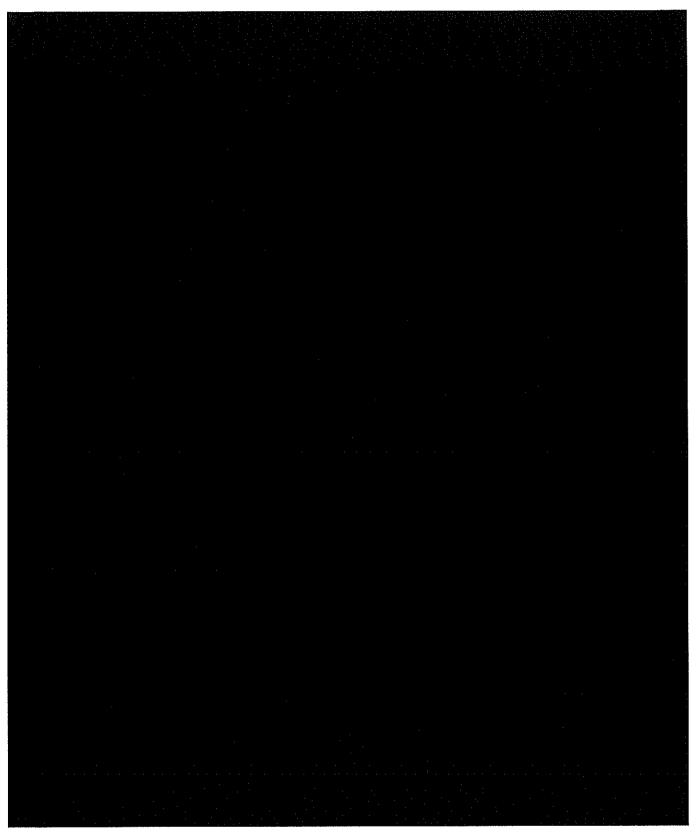
1 - 5 Fig 17 Tail Rotor Assembly







1 - 5 Fig 19 Collective to Yaw Pitch Interlink



1 - 5 Fig 20 Tail Pylon Folding

#### UV RESTRICTED

### PART 1

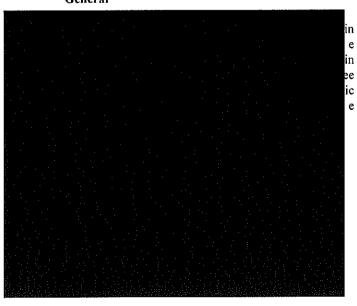
# **CHAPTER 6 - FLIGHT CONTROL SYSTEM**

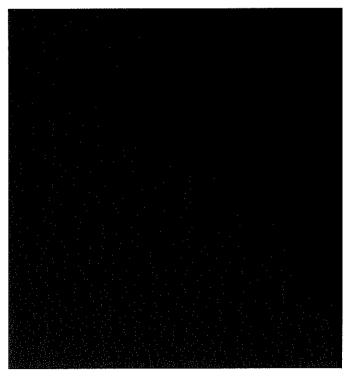
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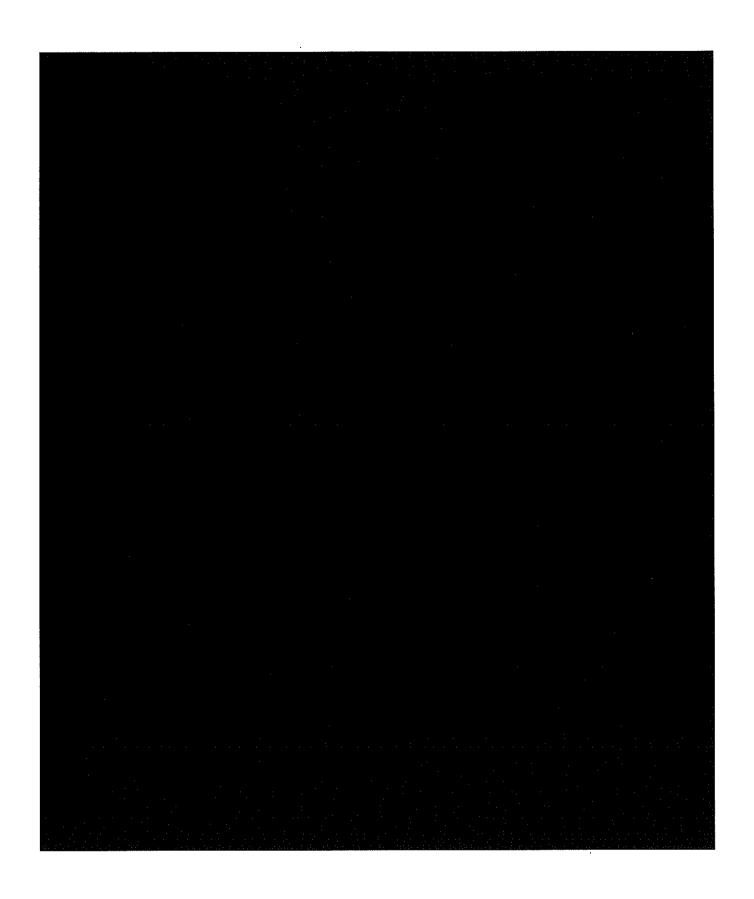
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# GENERAL

### General

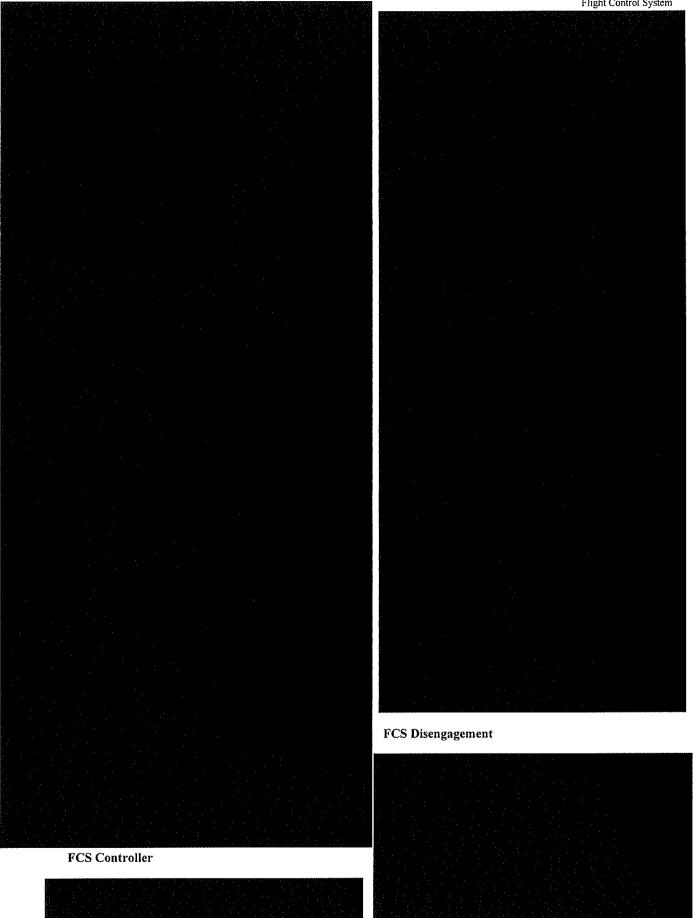


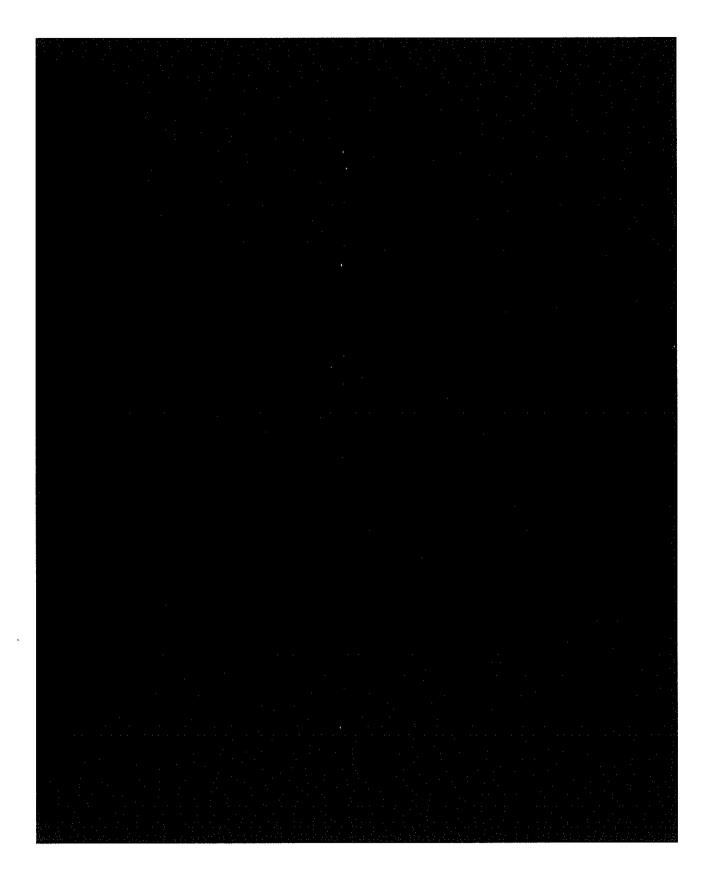




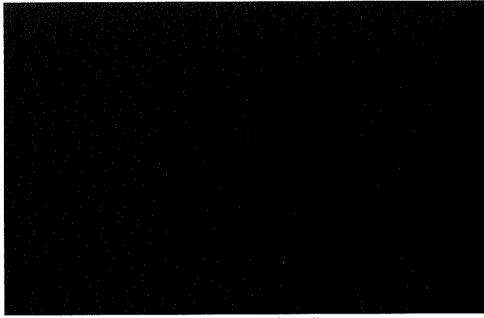
1 - 6 Fig 1 FCS General Arrangement

UK RESTRICTED AP101C-1308-15C Flight Control System

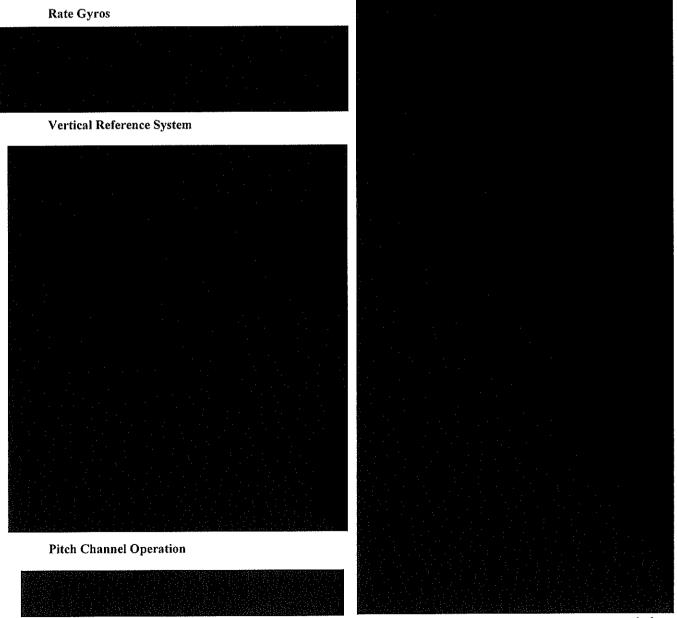


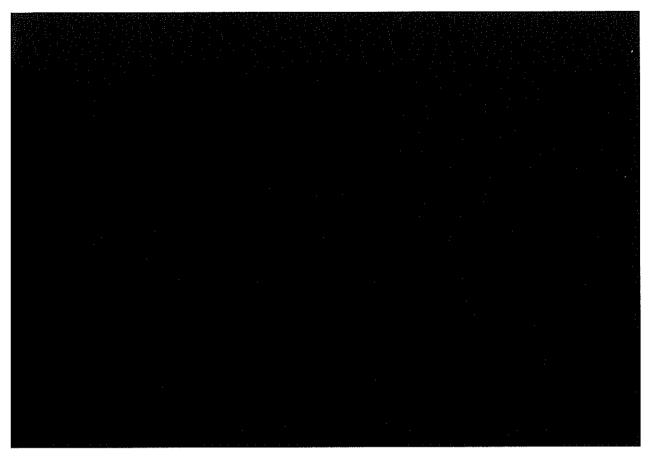


1 - 6 Fig 2 Controls and Indicators, General Arrangement

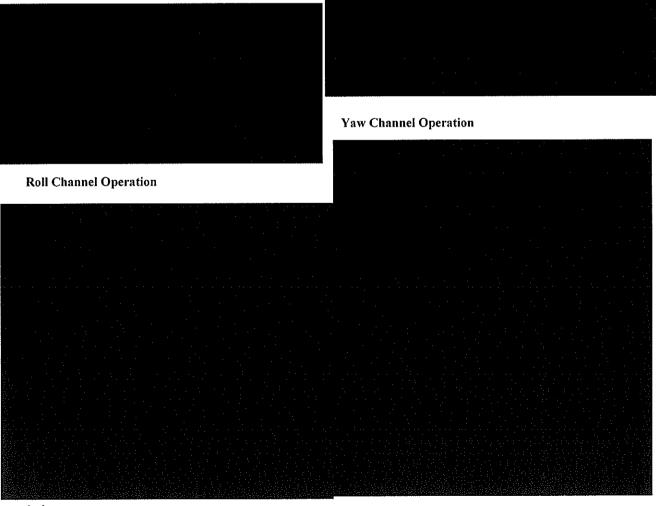


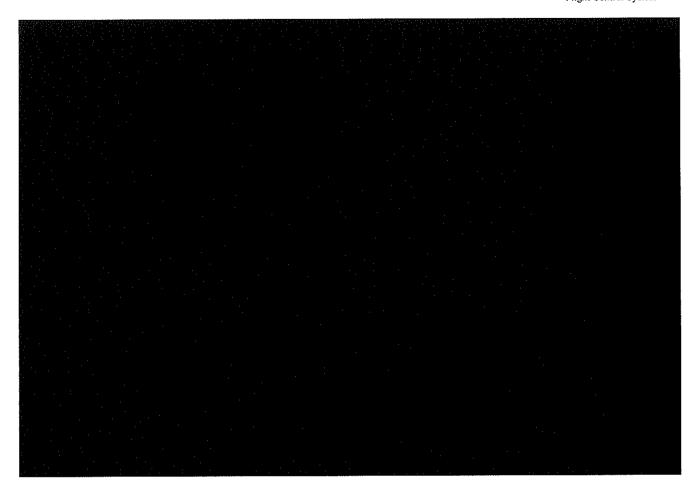
1 - 6 Fig 3 FCS Controller



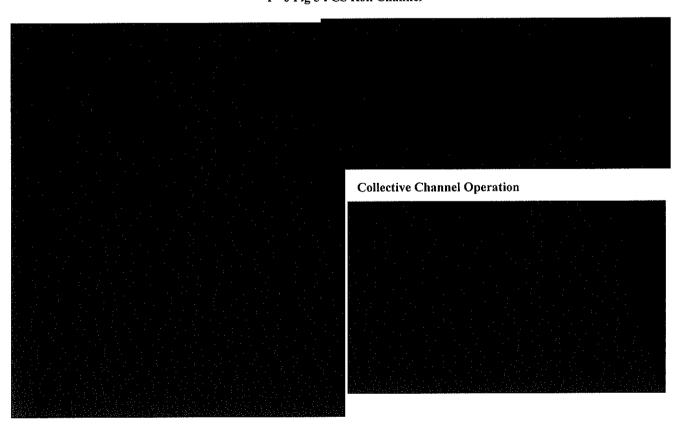


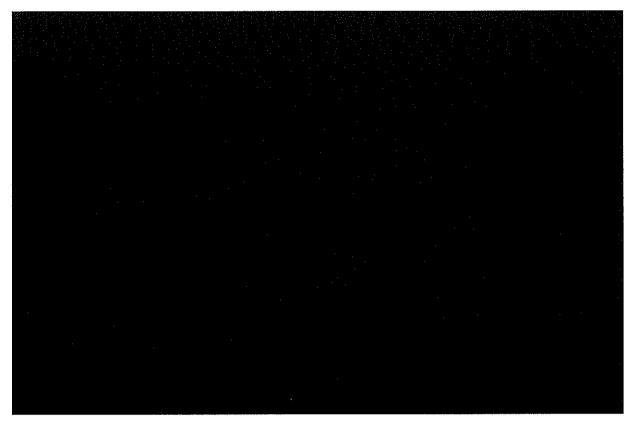
1 - 6 Fig 4 FCS Pitch Channel



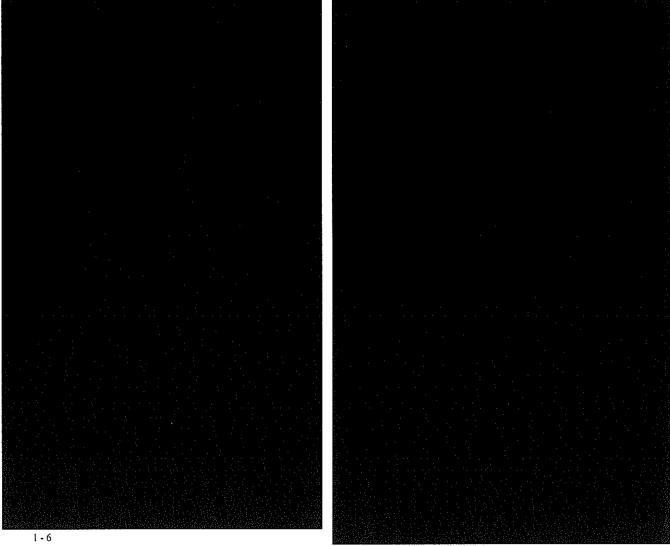


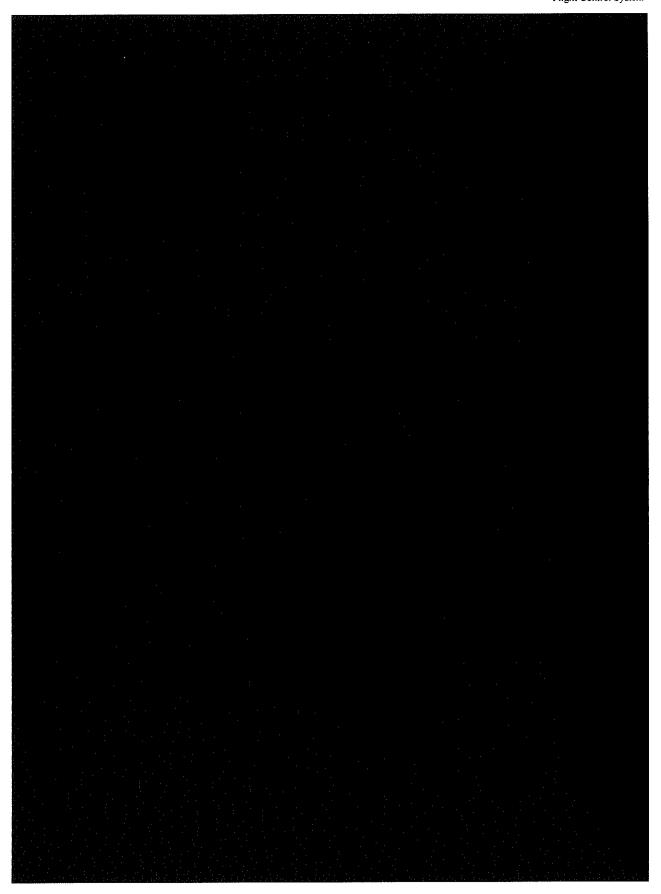
1 - 6 Fig 5 FCS Roll Channel





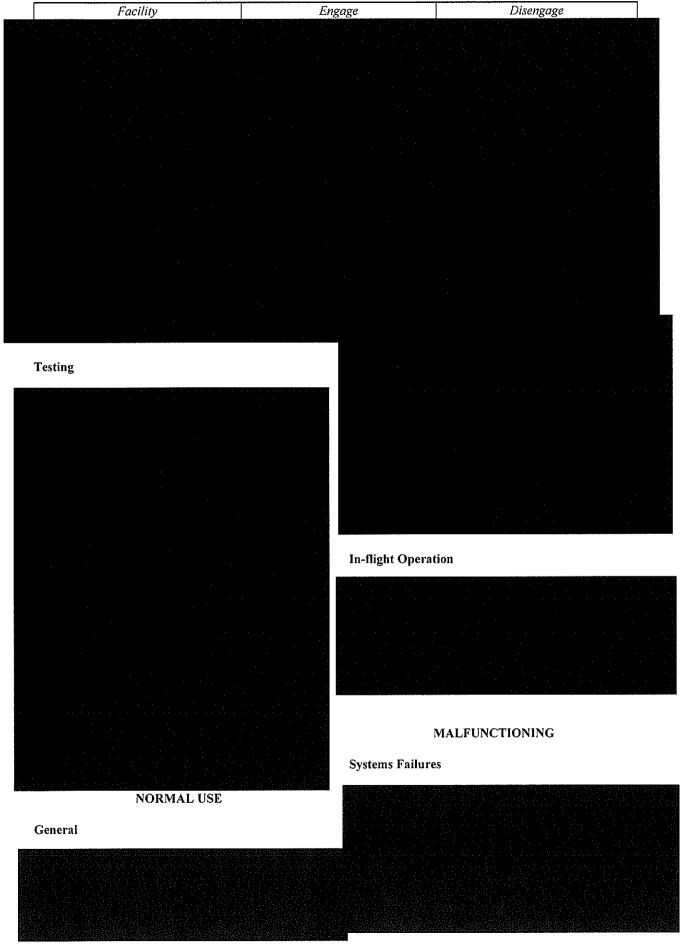
1 - 6 Fig 6 FCS Yaw Channel





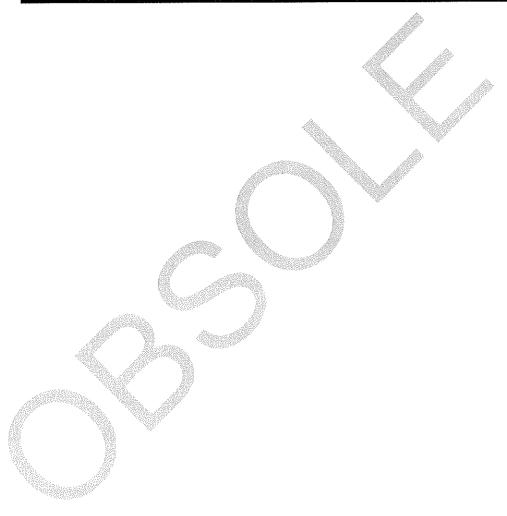
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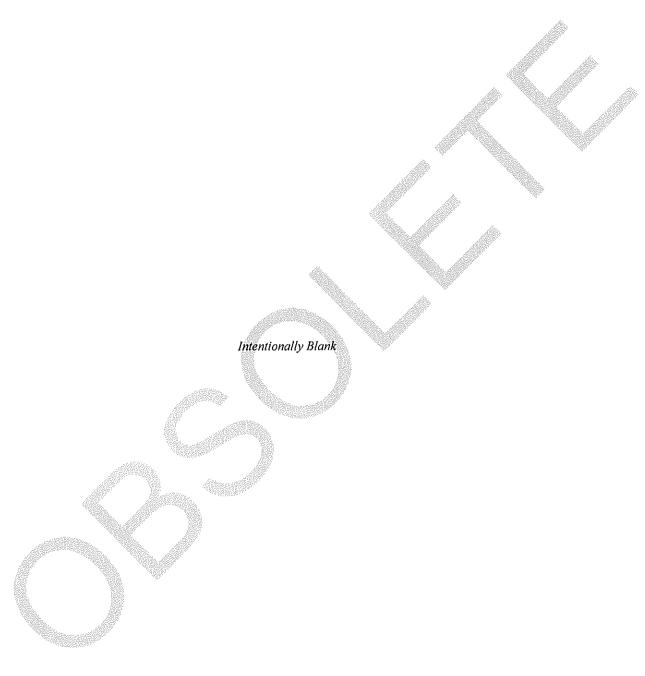
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# PART 1

# **CHAPTER 7 - HYDRAULIC SYSTEMS**

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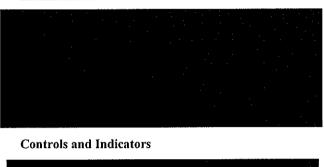
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### GENERAL



Table 1 - Services Supplied from No 3 Hydraulic Sys-

Service	Reference			
Rotor brake.	Part 1, Chap 5.			
Wheel lock release.	Part I, Chap 10.			
Nosewheel castoring.	Part 1, Chap 10.			
Harpoon.	Part 1, Chap 10.			
Hoist.	Part 1, Chap 9.			







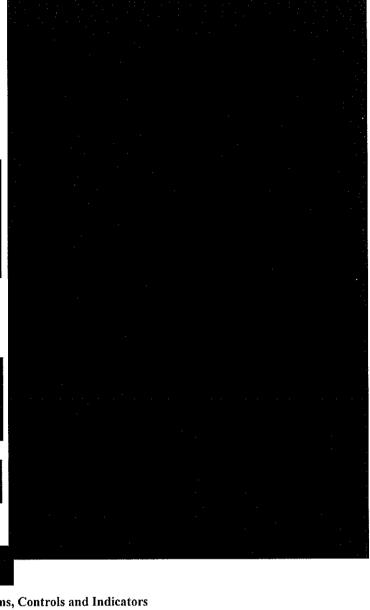
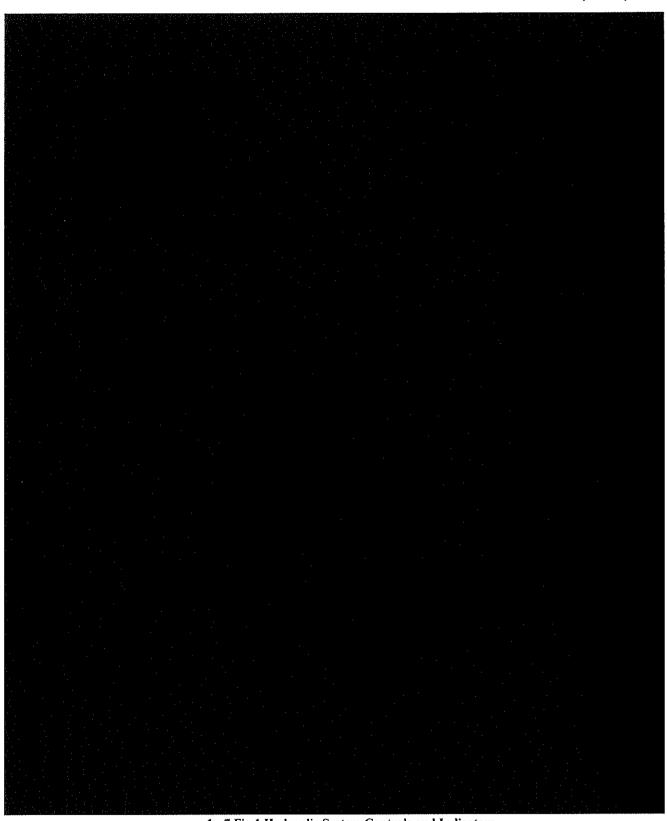
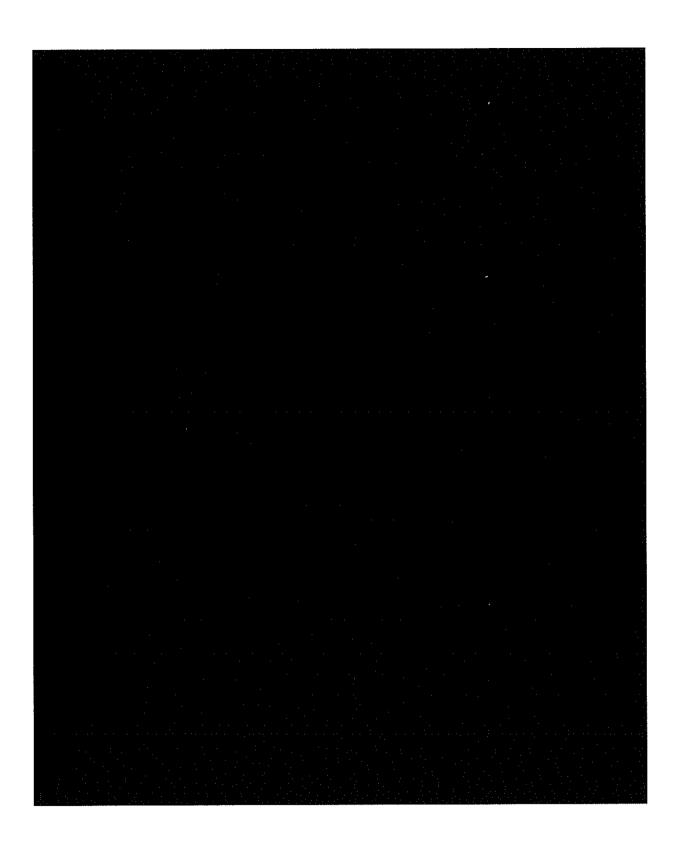


Table 2 - Hydraulic Systems, Controls and Indicators

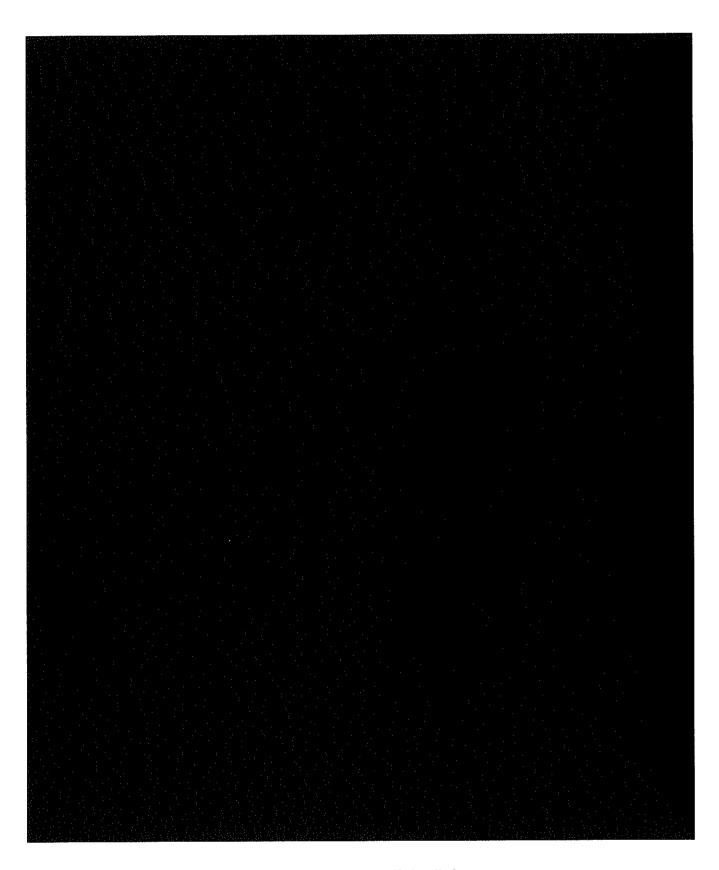
Control/Indicator	Marking	Location				
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Research Control Control (Control Control Cont						



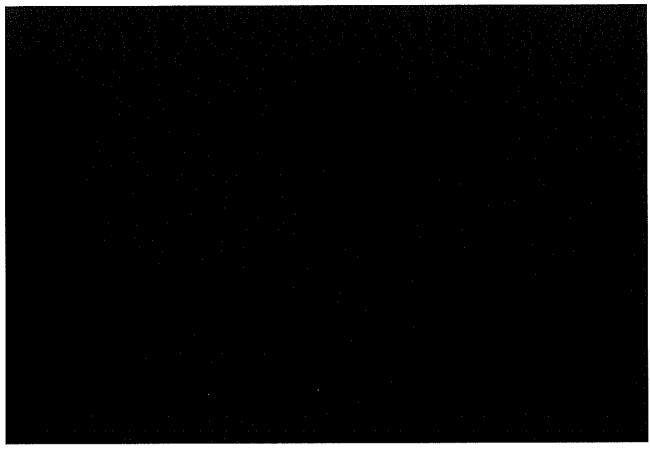
1 - 7 Fig 1 Hydraulic System Controls and Indicators



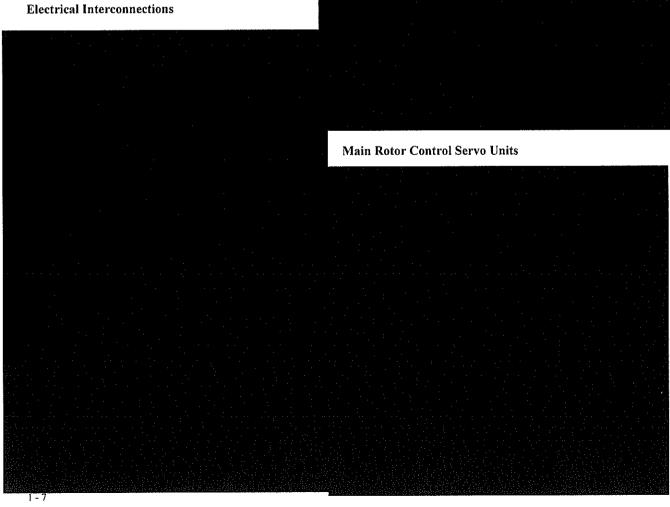
1 - 7 Fig 2 Hydraulic Systems Schematic



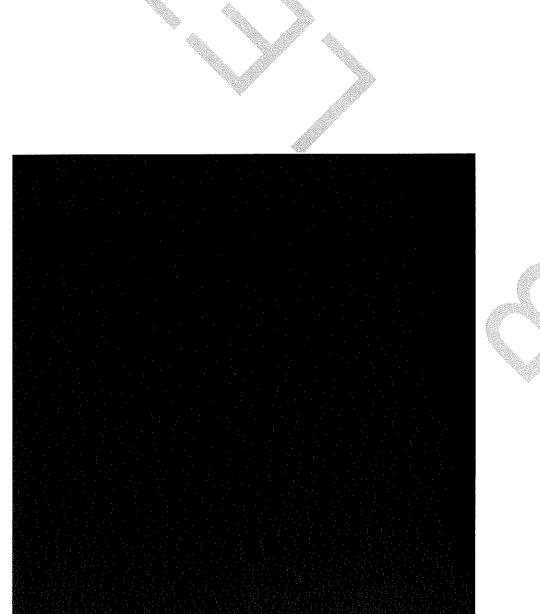
1 - 7 Fig 3 No 1 and 2 Hydraulic Systems

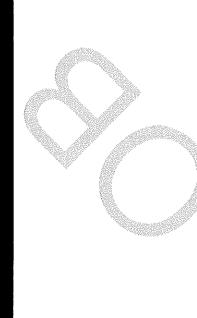


1 - 7 Fig 4 Electrical Interconnections

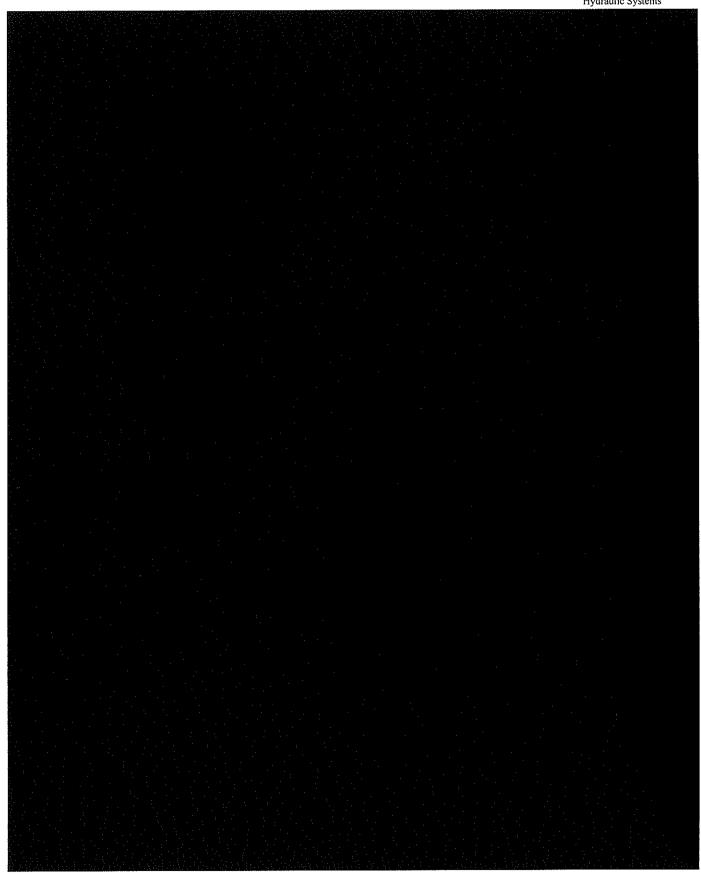


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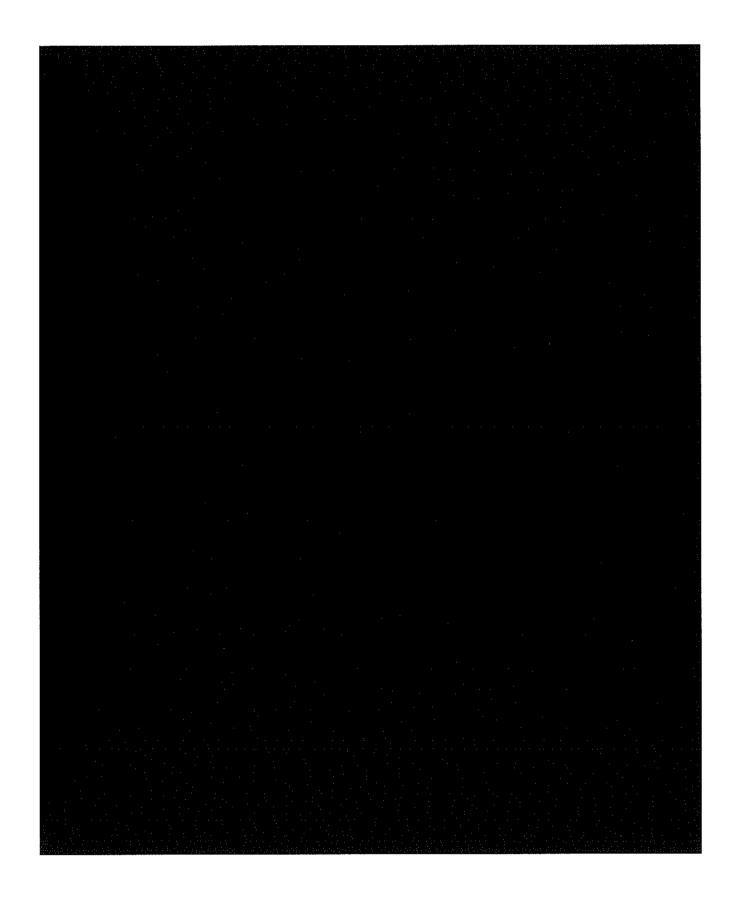




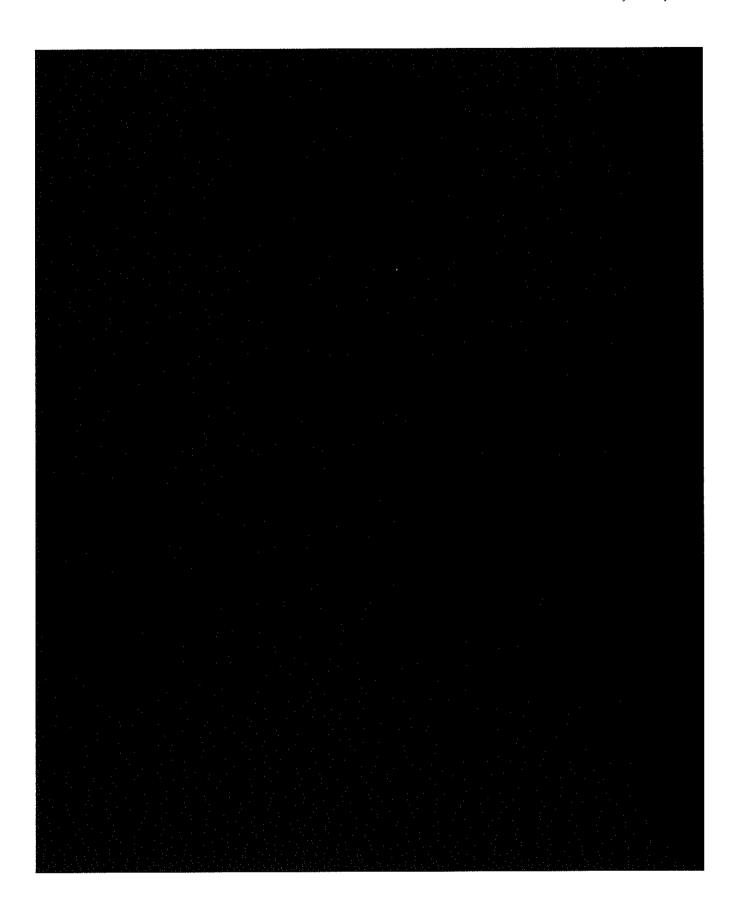
UK RESTRICTED



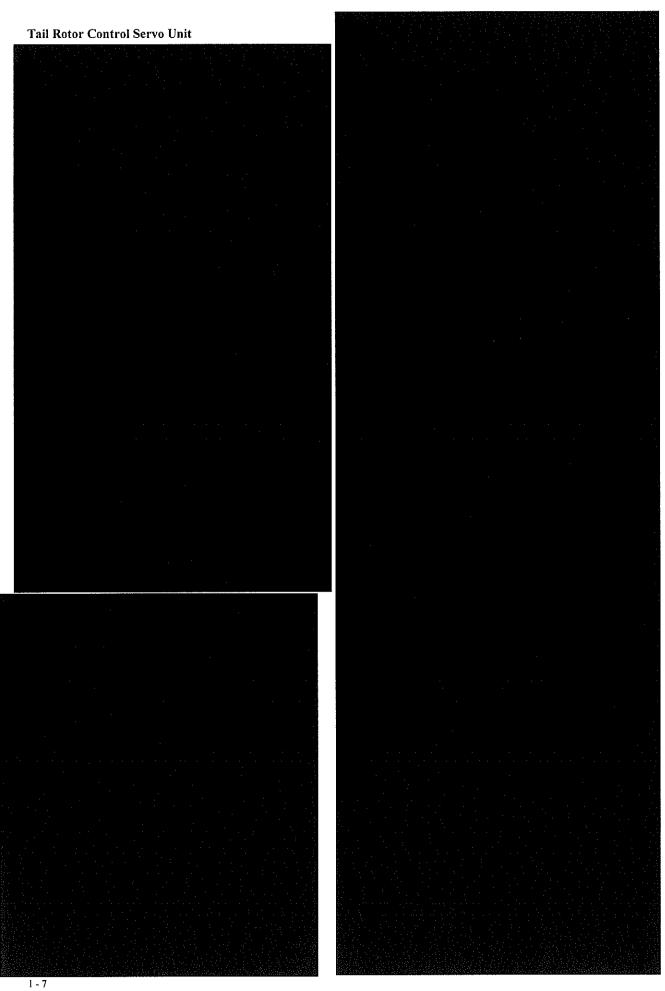
1 - 7 Fig 6 Main Rotor Control Servo Unit, Pilot Controlled Input



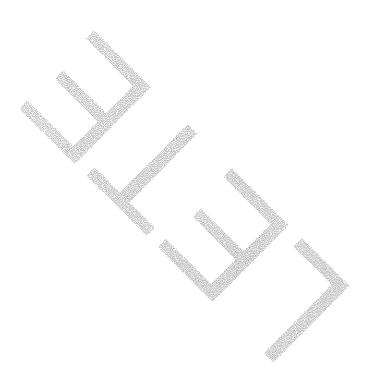
1 - 7 Fig 7 Main Rotor Control Servo Unit, FCS Inputs



1 - 7 Fig 8 Main Rotor Control Servo Unit, Breakout Assembly Activated



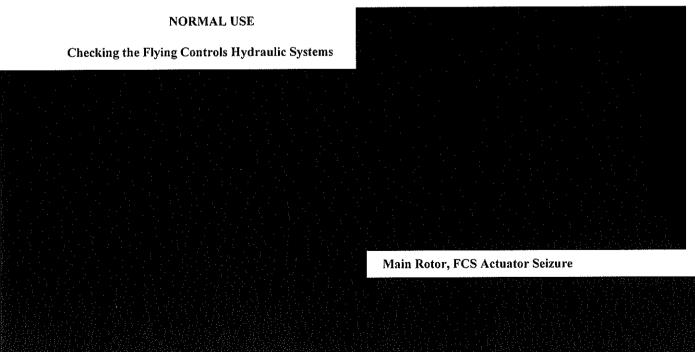
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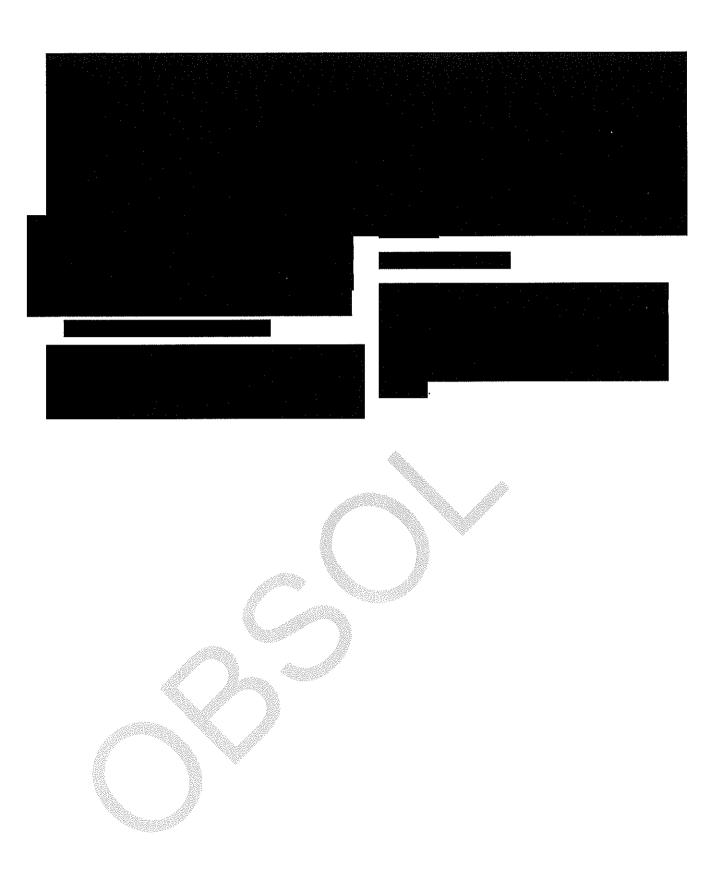
Tail Rotor Control Servo Unit, Manual Control



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### PART 1

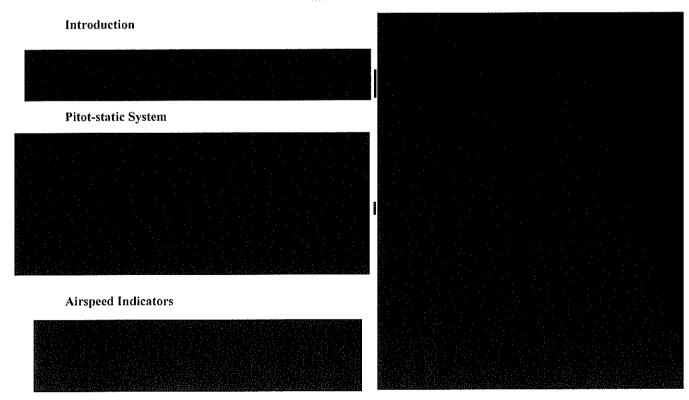
# **CHAPTER 8 - FLIGHT INSTRUMENTS**

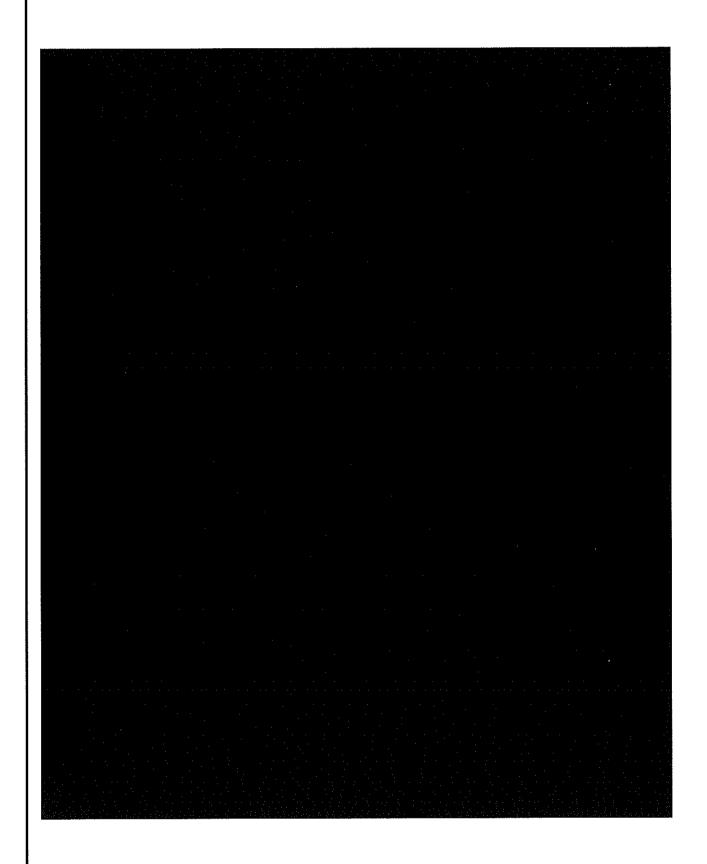
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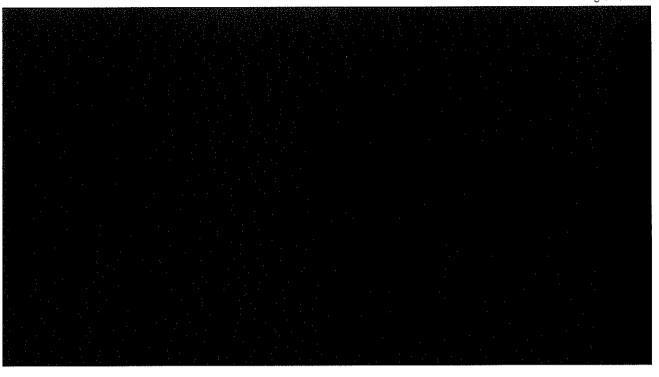
#### **GENERAL**

#### **Barometric Altimeters**

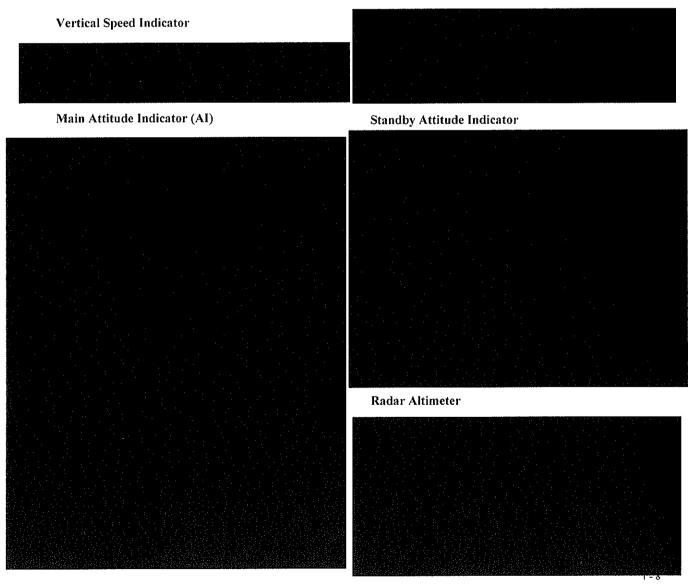


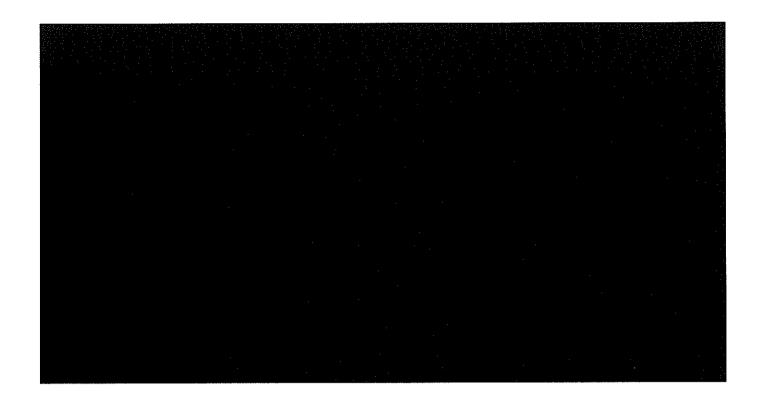


1 - 8 Fig 1 Flight Instruments, General Arrangements

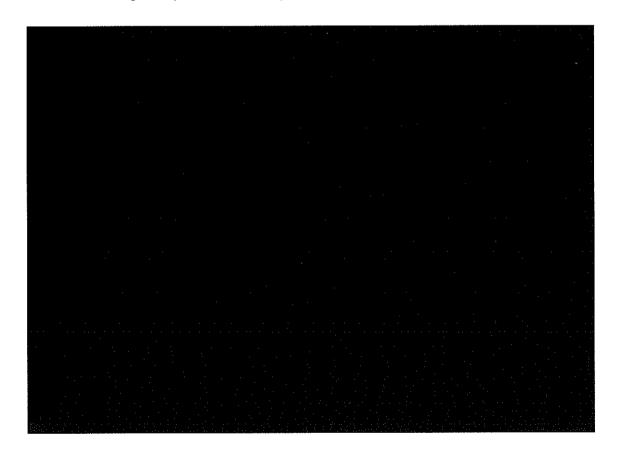


1 - 8 Fig 2 Pitot-static System

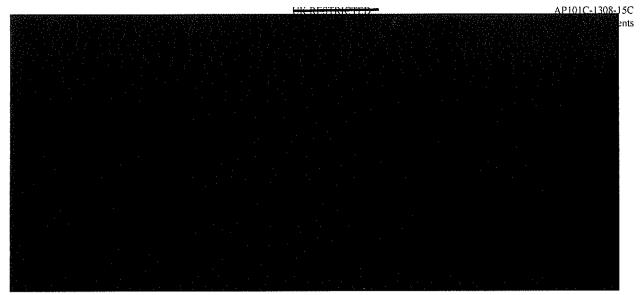




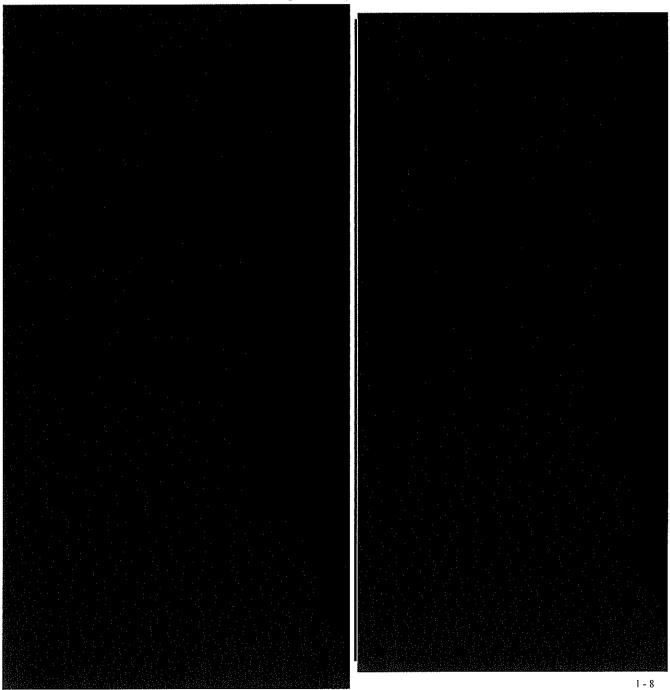
1 - 8 Fig 3 Airspeed and Vertical Speed Indicators and Barometric Altimeter

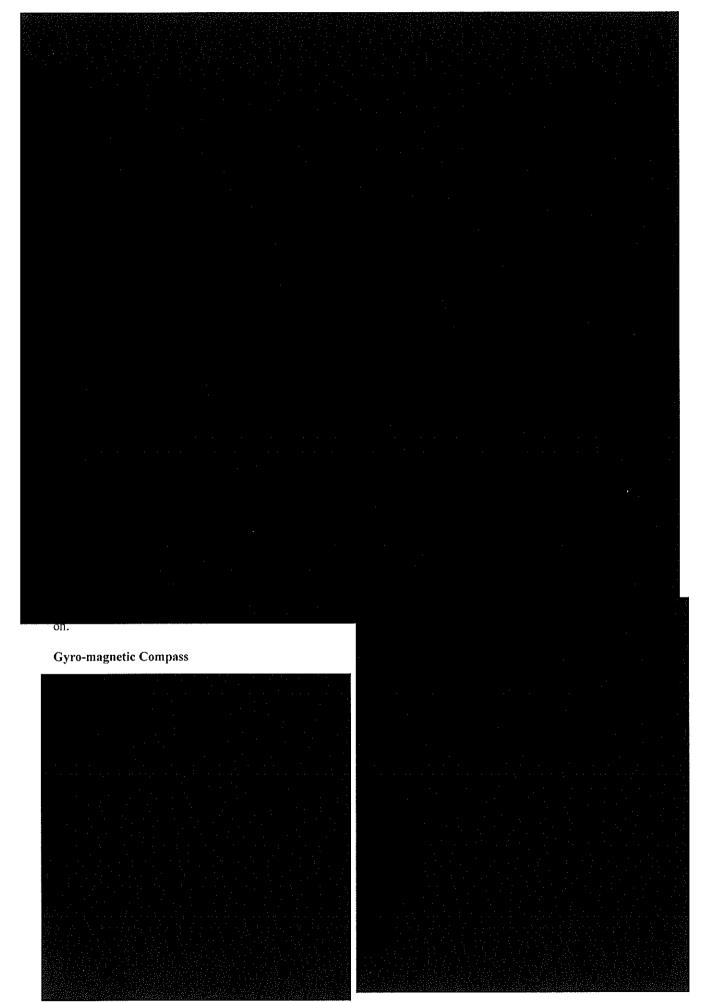


1 - 8 Fig 4 Attitude Indicator and Mode Selector

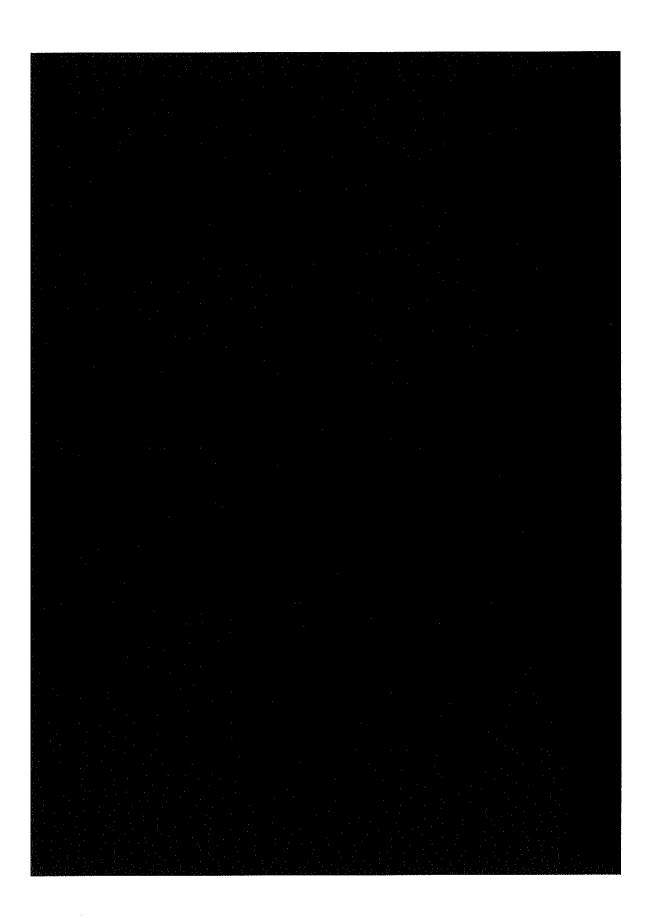


1 - 8 Fig 5 Standby Attitude Indicator

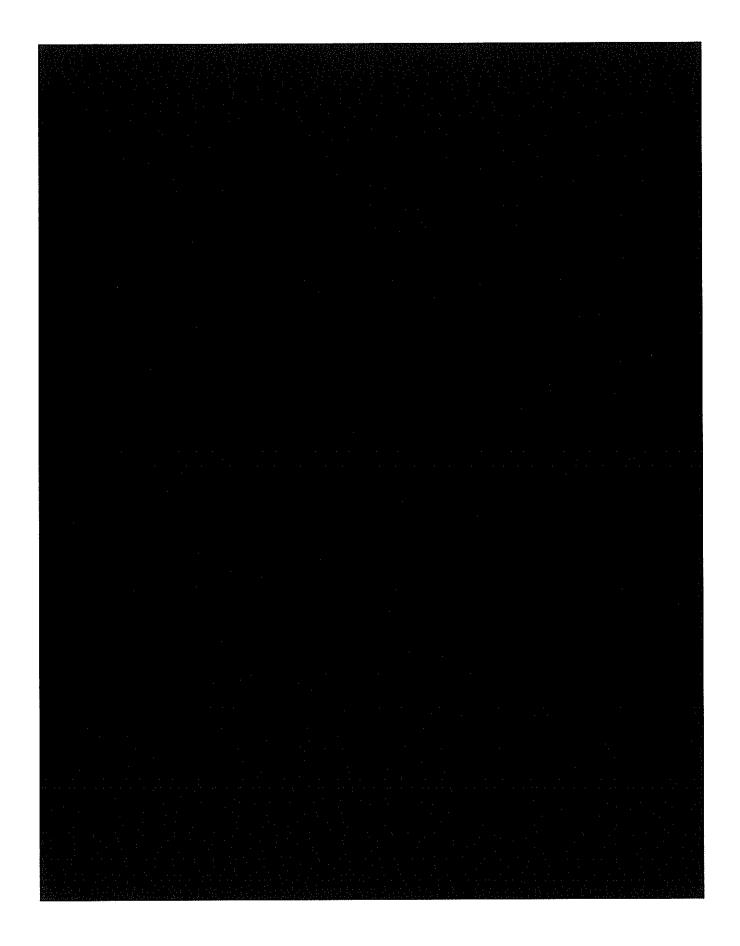




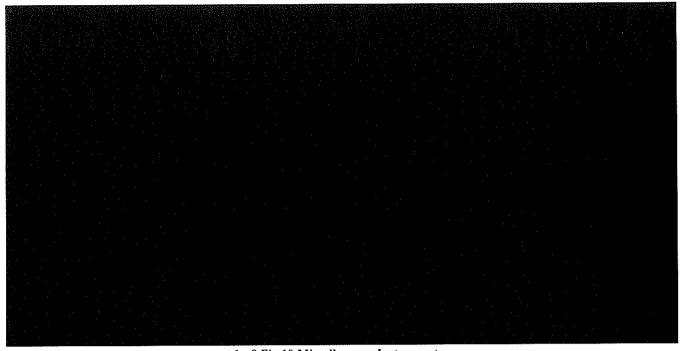
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1 - 8 Fig 8 Radar Altimeter and Low Height Warning System (Post-Mod 792 Part B)



1 - 8 Fig 9 Gyro-magnetic Compass



1 - 8 Fig 10 Miscellaneous Instruments

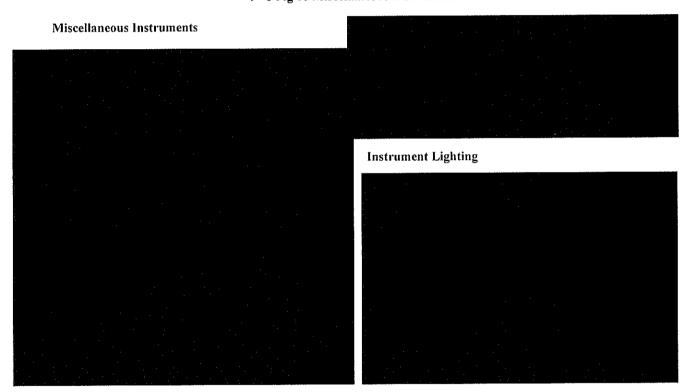
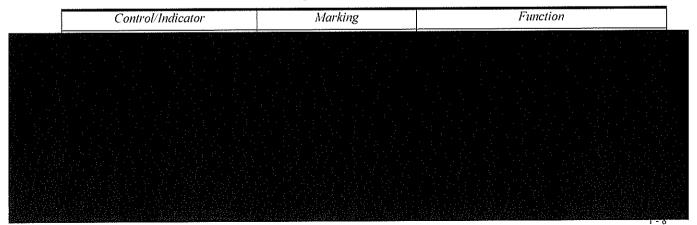


Table 1 - Compass System Controls and Indicators



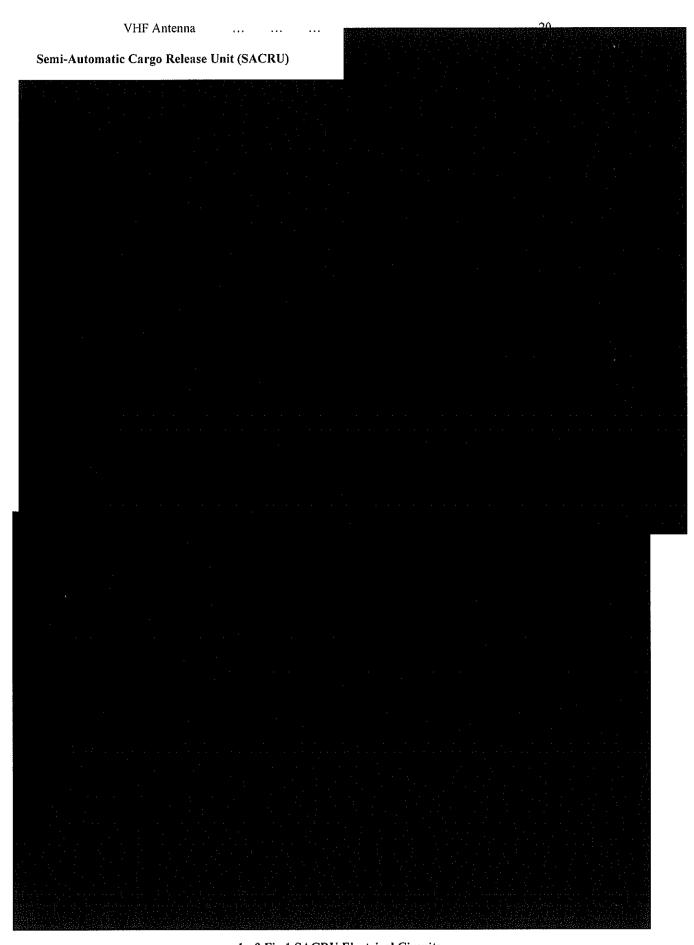
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## PART 1

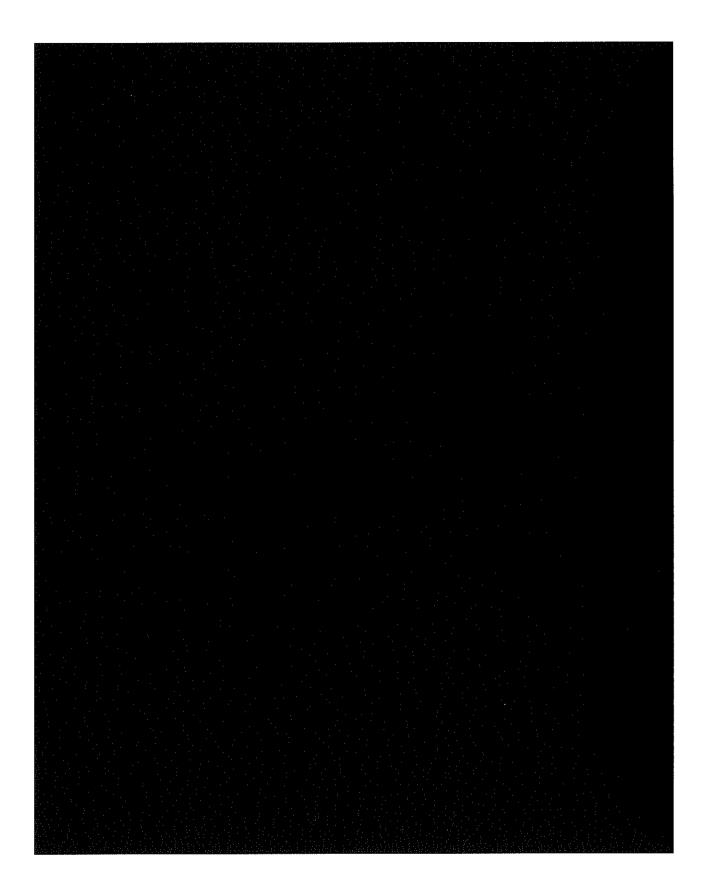
# **CHAPTER 9 - MISCELLANEOUS ROLE EQUIPMENT**

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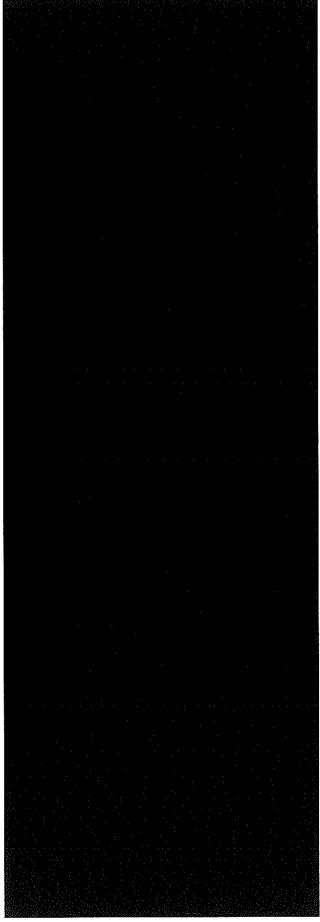


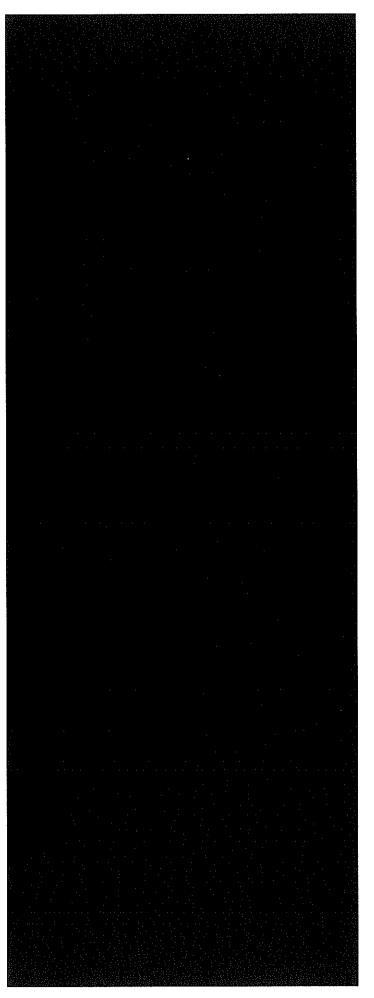
1 - 9 Fig 1 SACRU Electrical Circuits

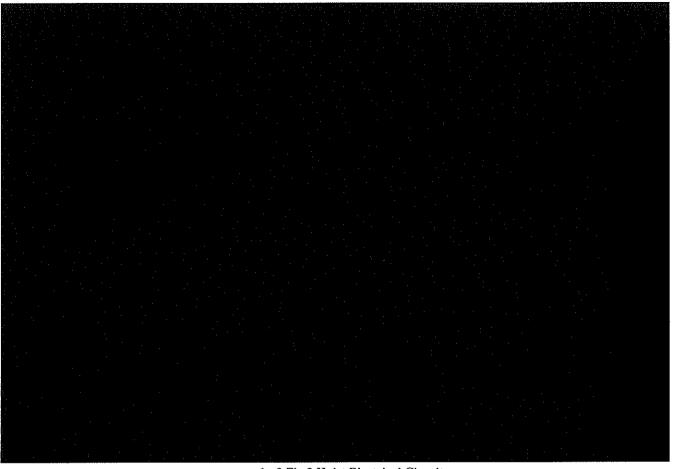


1 - 9 Fig 2 SACRU Controls and Indicators

#### Hoist

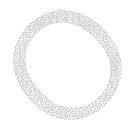


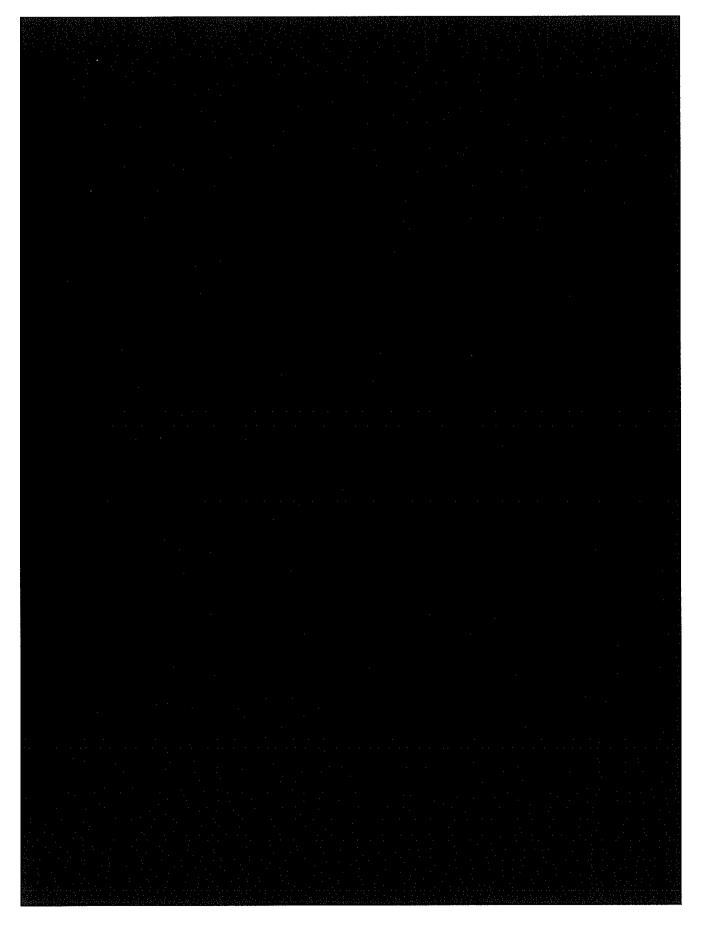




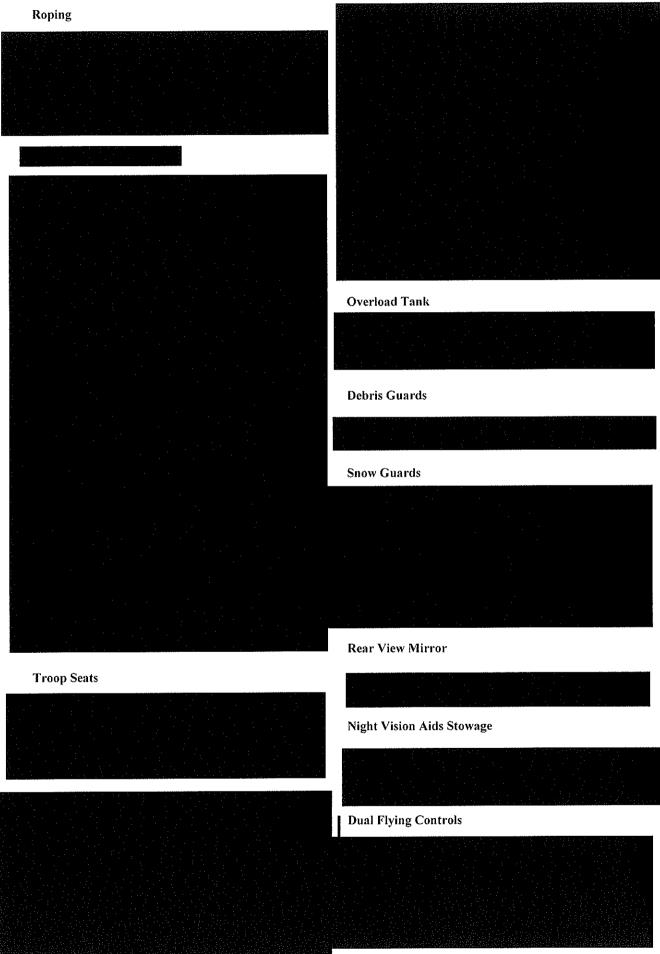
1 - 9 Fig 3 Hoist Electrical Circuits

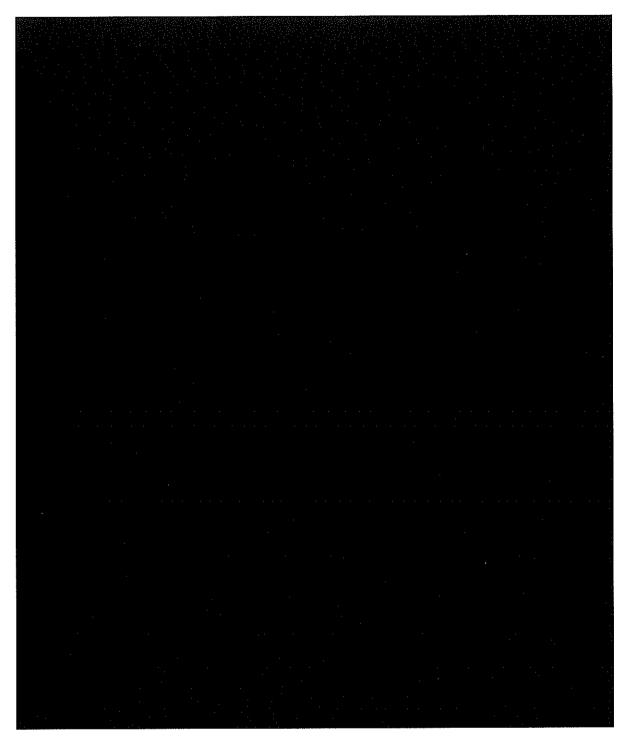






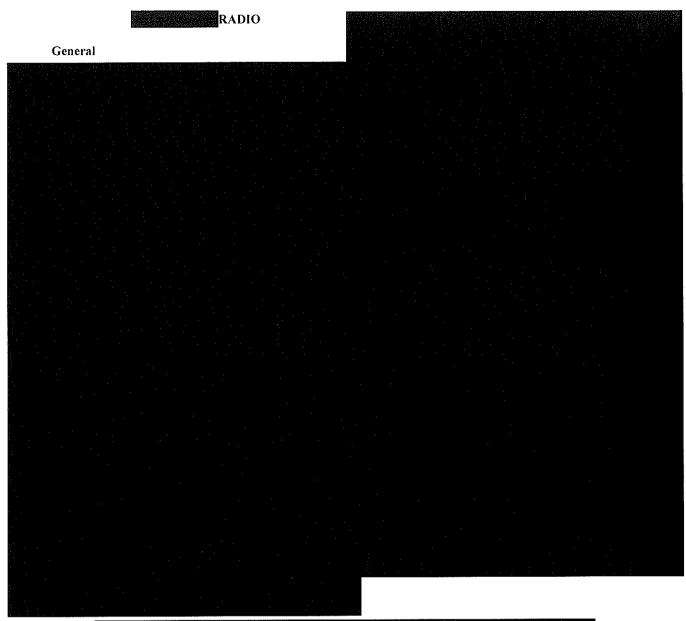
1 - 9 Fig 4 Hoist Controls and Indicators





1 - 9 Fig 5 Inflatable Troop Seats







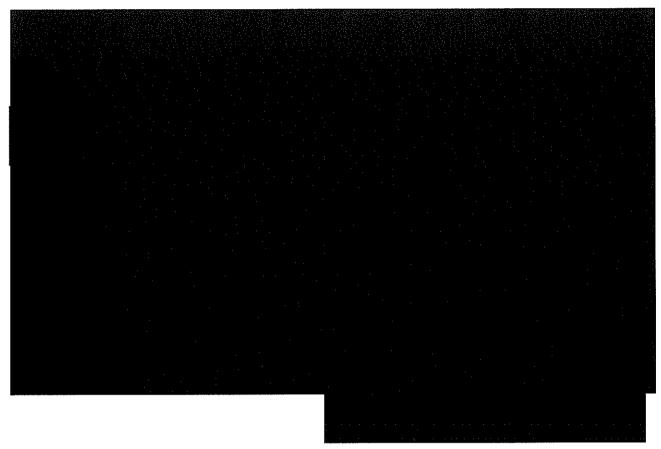
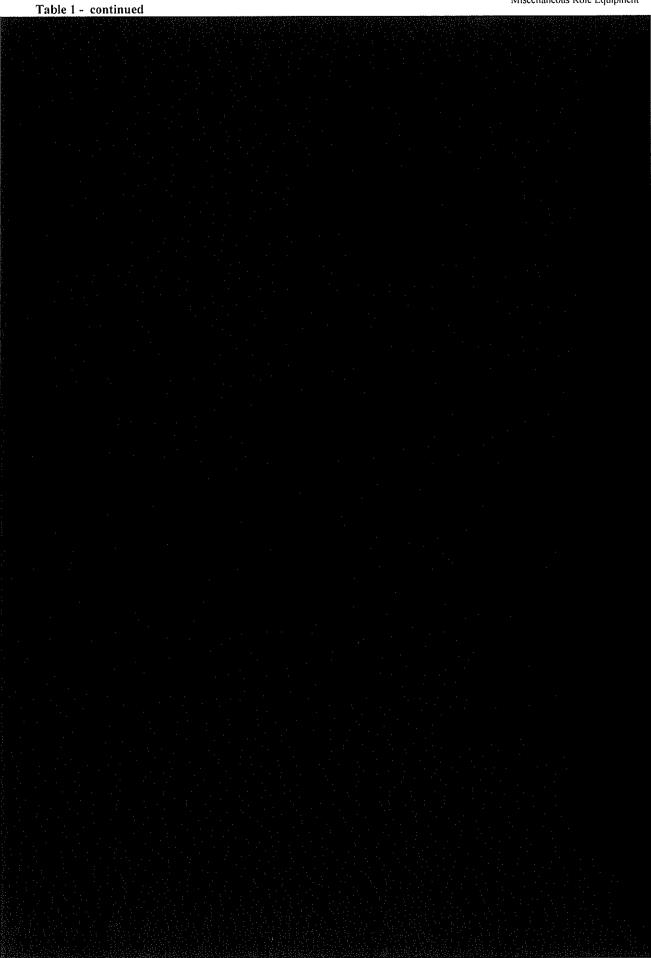


Table 1 - Display Field Entries

Display Fields	Display Entries						



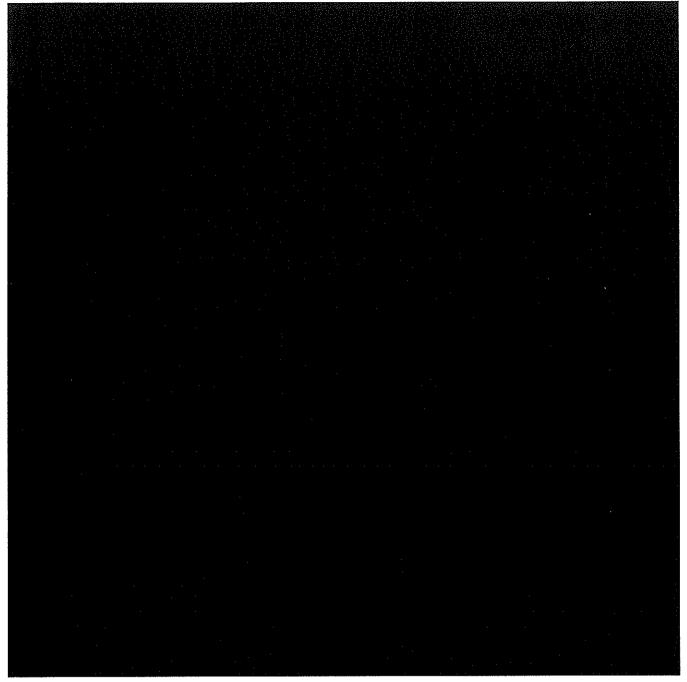
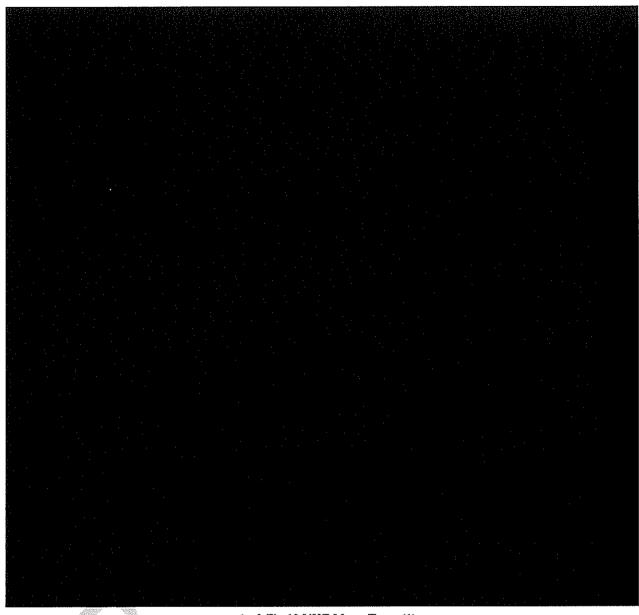
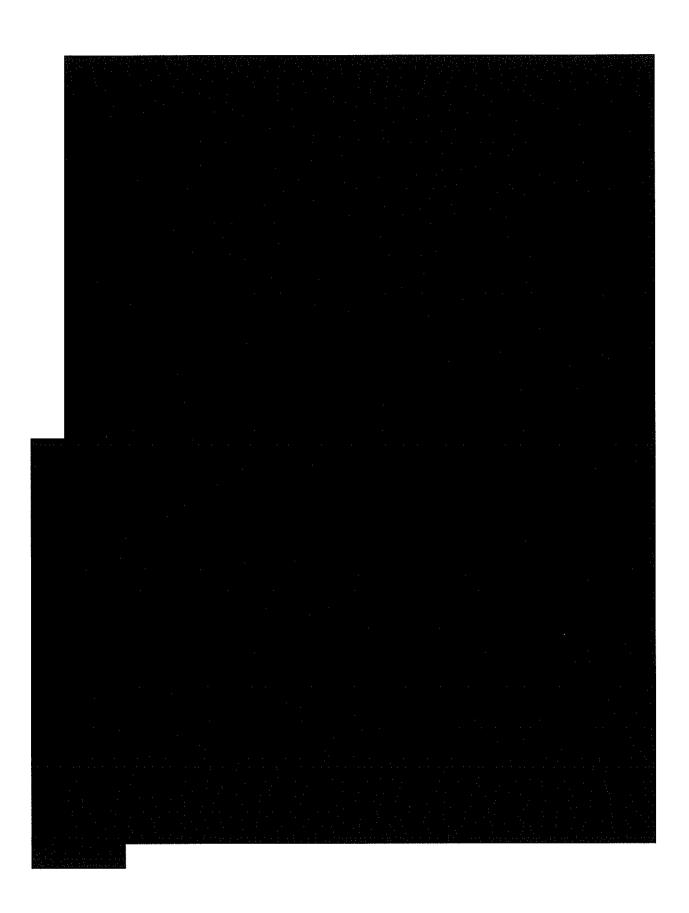


Table 2 - Menu Trees

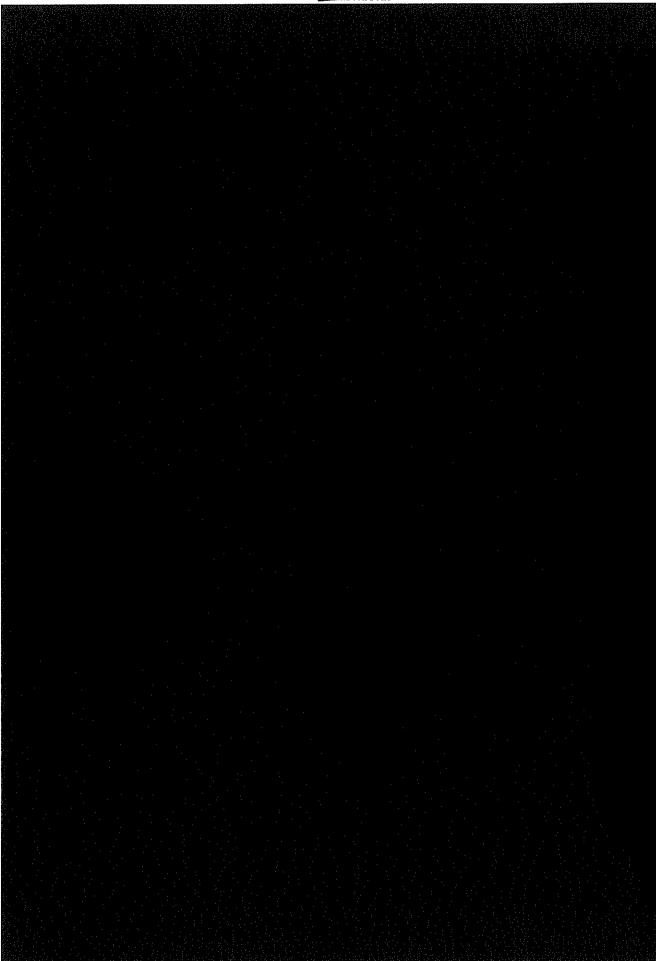
Мепи Кеу	Menu Tree	Function						



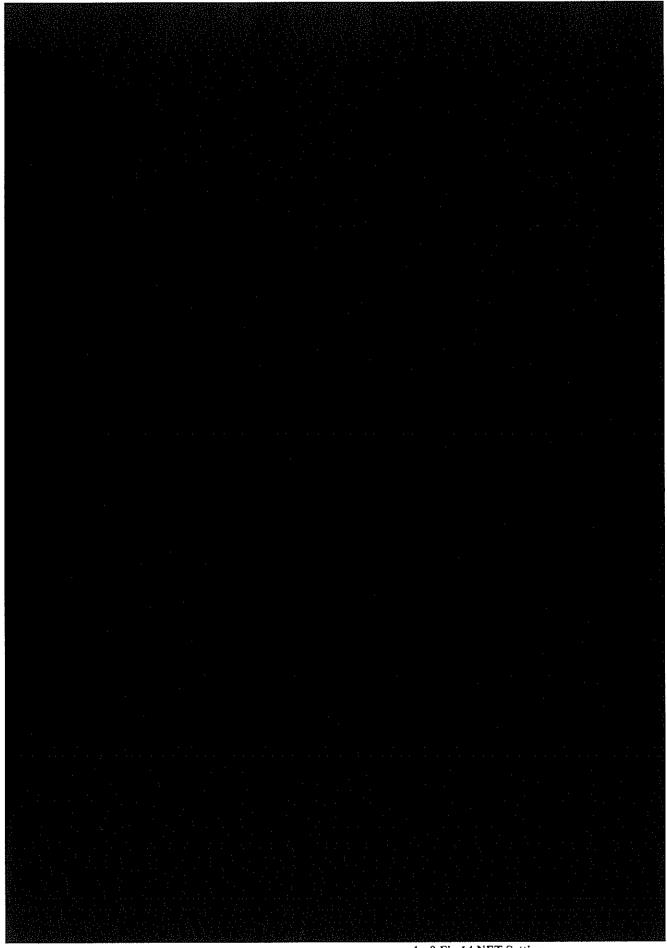
1 - 9 Fig 12 VHF Menu Trees (1)



1 - 9 Fig 13 VHF Menu Trees (2)



AP101C-1308-15C Miscellaneous Role Equipment UK DESTRICTED



1 - 9 Fig 14 NET Setting



Malfunctions

84. HAIL Detect Protocol:

BITE Checks

1-9

Table 4 - Audio Tones, Alarms and Alerts

		no Tones, Atarms and A	
Condition	Frequency Audio (Hz) Level	Duration	Function
		<u> </u>	
ROLE-FIT N	MACHINE GUNS		
		•	
		Branch and Alexander	

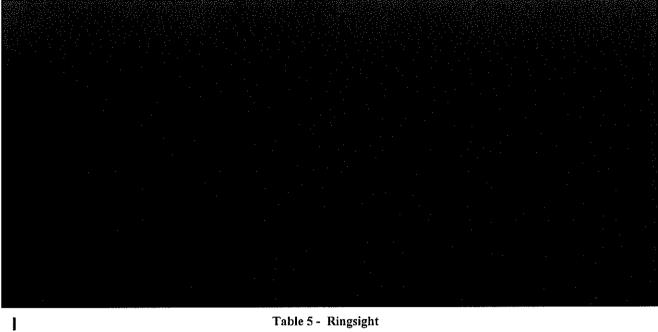
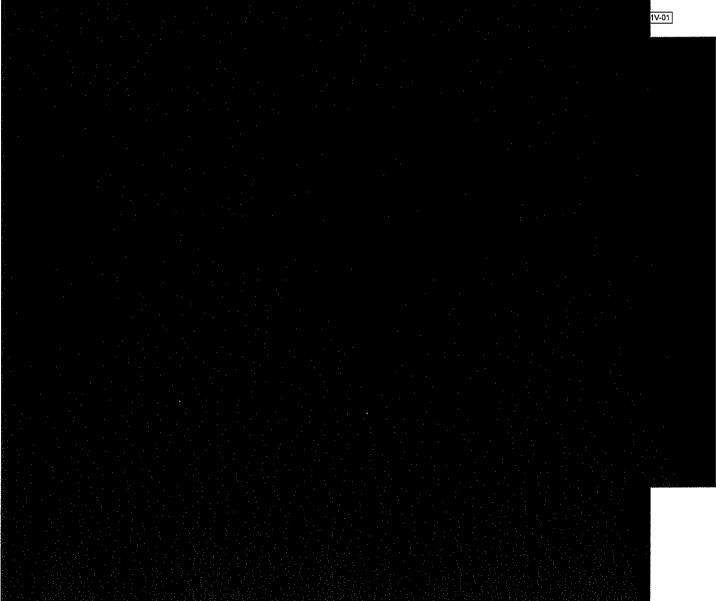
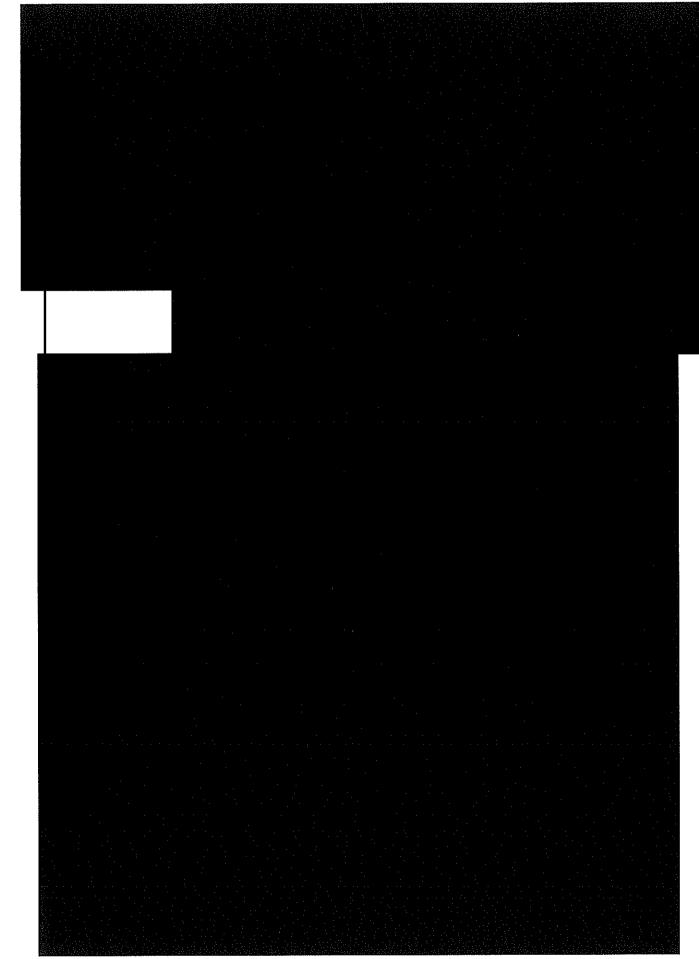


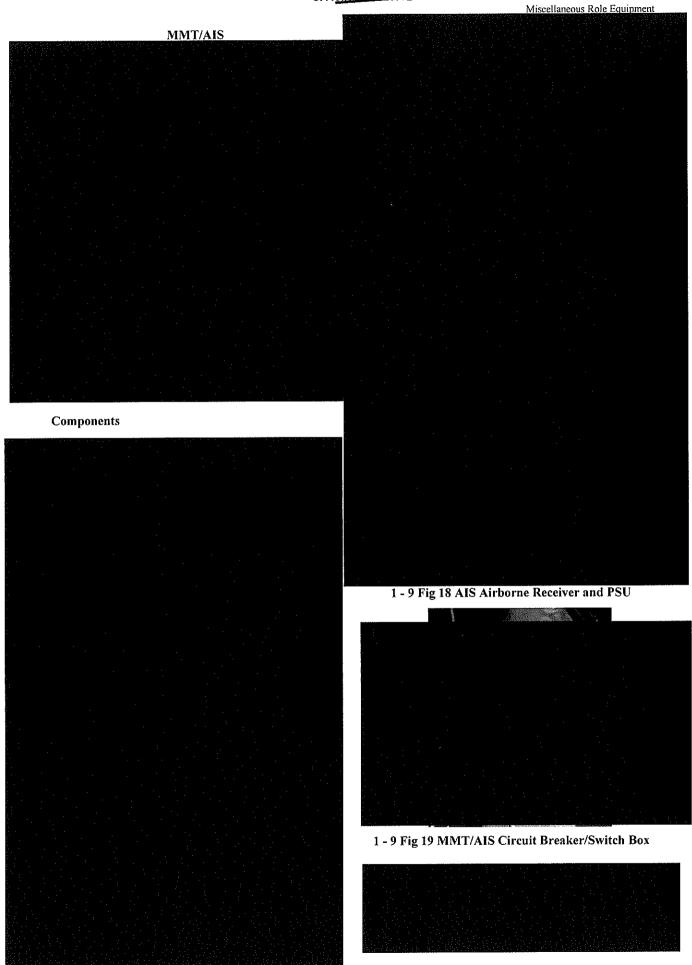
Table 5 - Ringsight

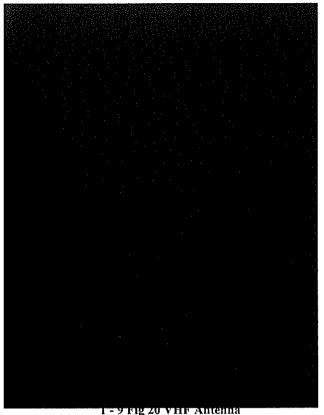
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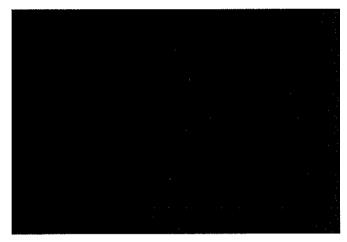




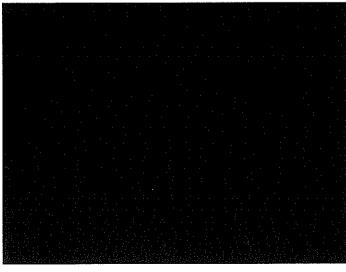


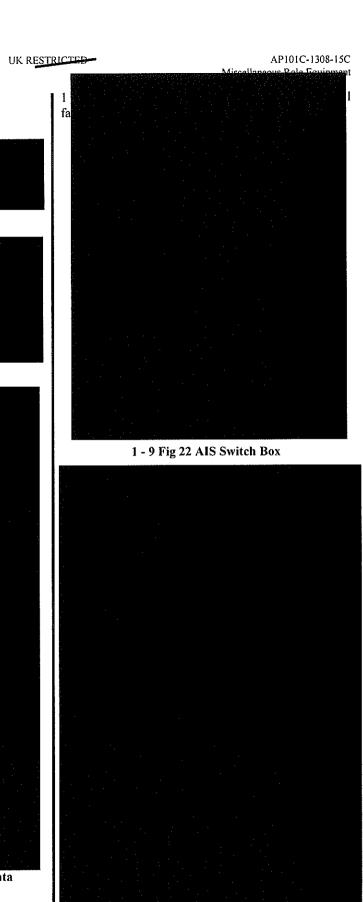


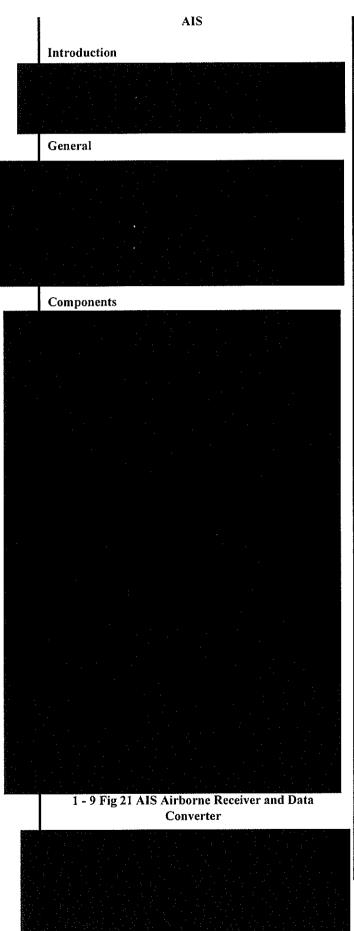


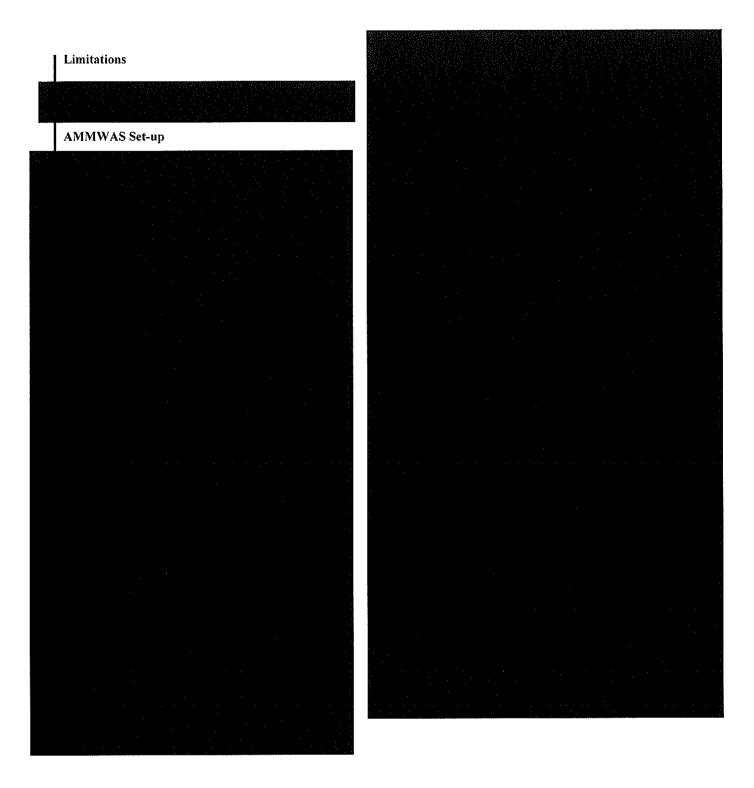


Limitations









## PART 1

# **CHAPTER 10 - GENERAL AND EMERGENCY EQUIPMENT**

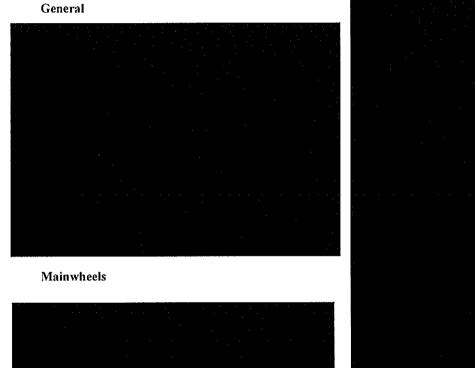
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#### LANDING GEAR AND HARPOON

# Nosewheel Castoring

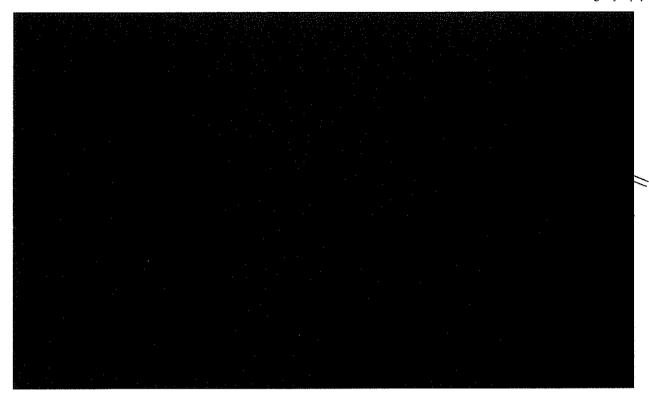




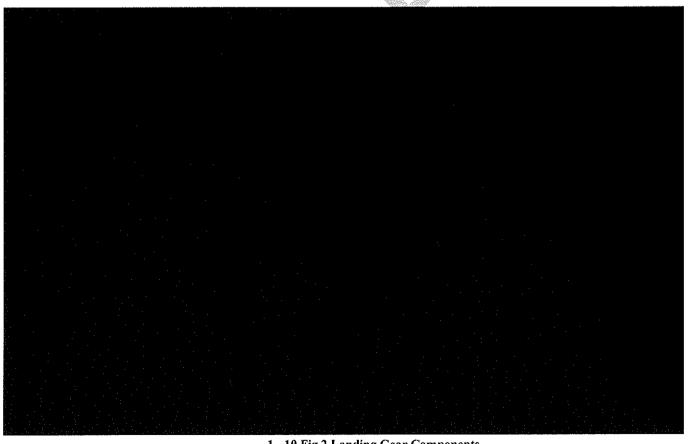


Wheel Locks

Nosewheels

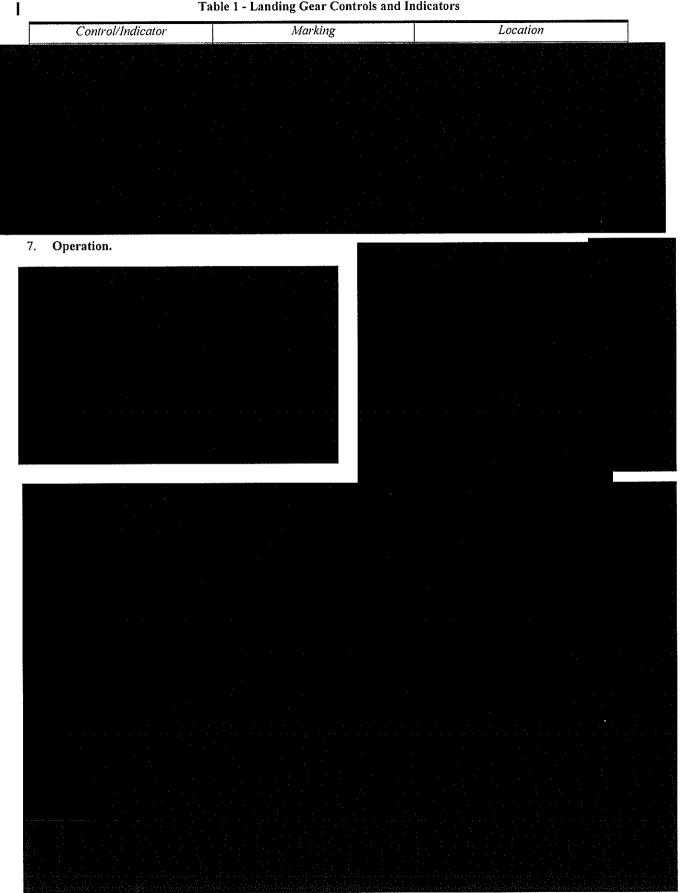


1 - 10 Fig 1 Landing Gear and Harpoon

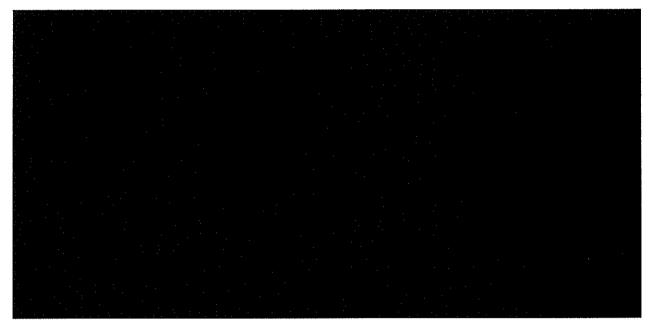


1 - 10 Fig 2 Landing Gear Components

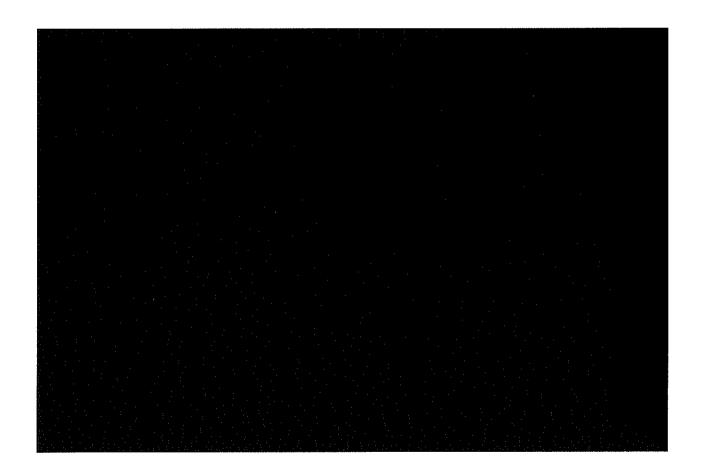
Table 1 - Landing Gear Controls and Indicators



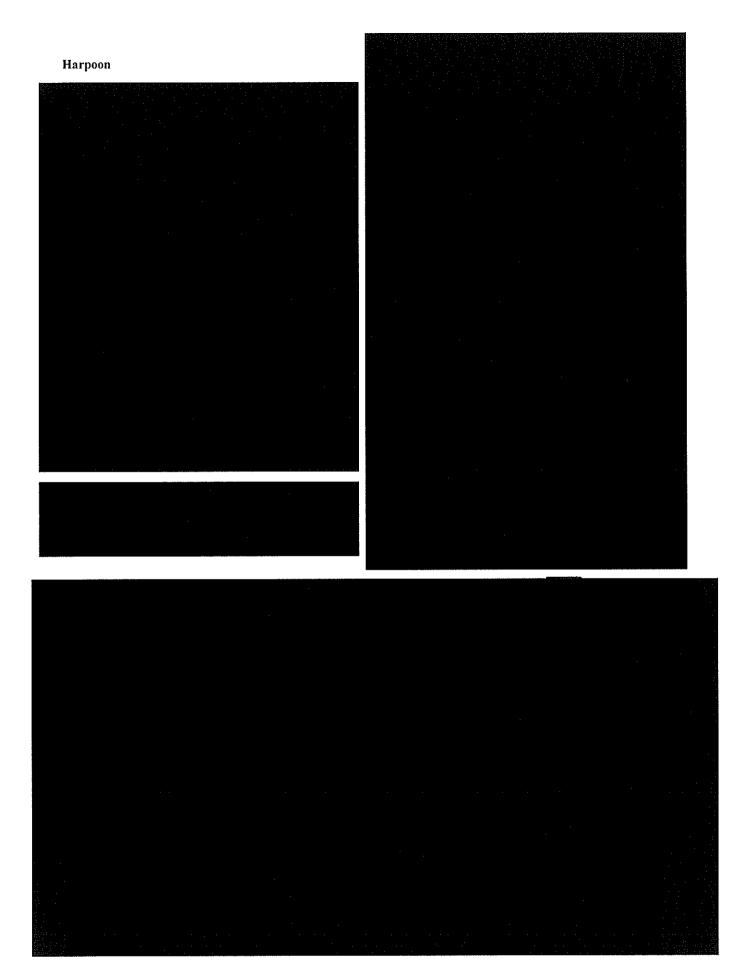
1 - 10 Fig 3 Nosewheel Castoring System



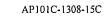
1 - 10 Fig 4 Wheel Lock Operation



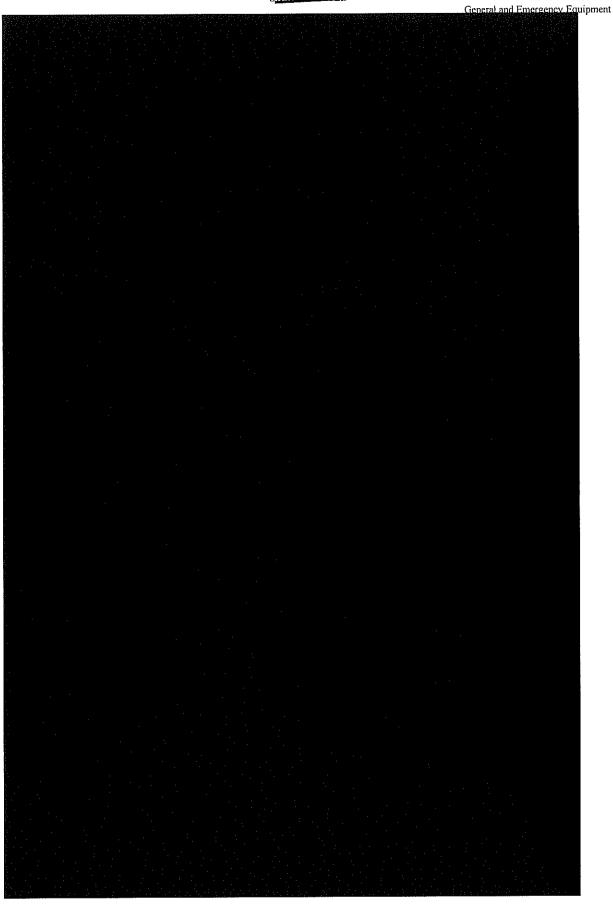
1 - 10 Fig 5 Wheel Lock System



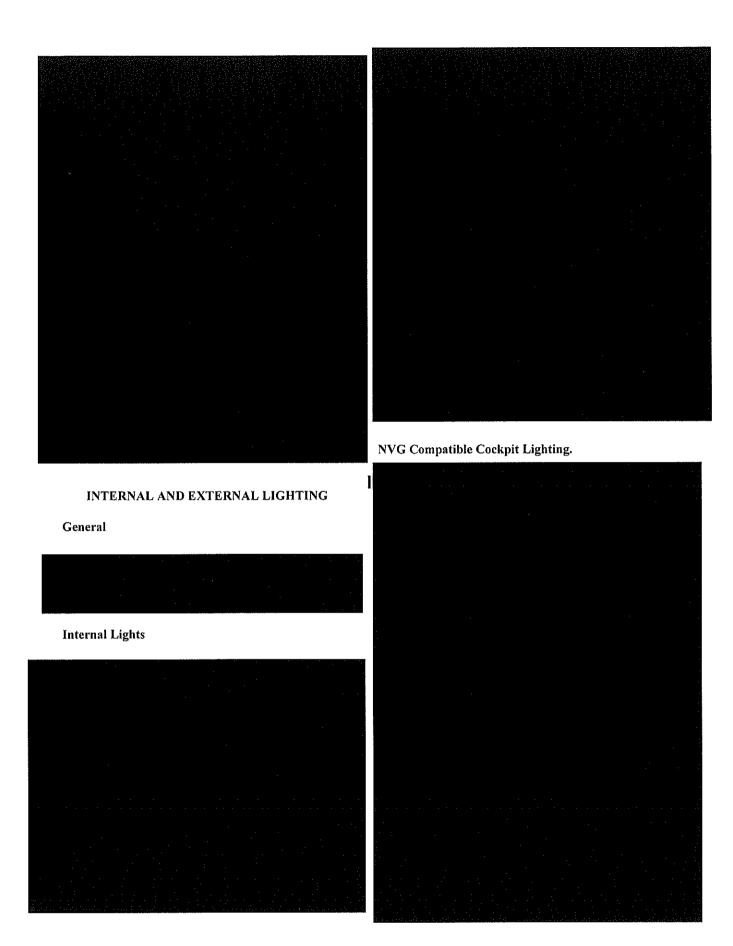
1 - 10 Fig 6 Deck Lock Harpoon System

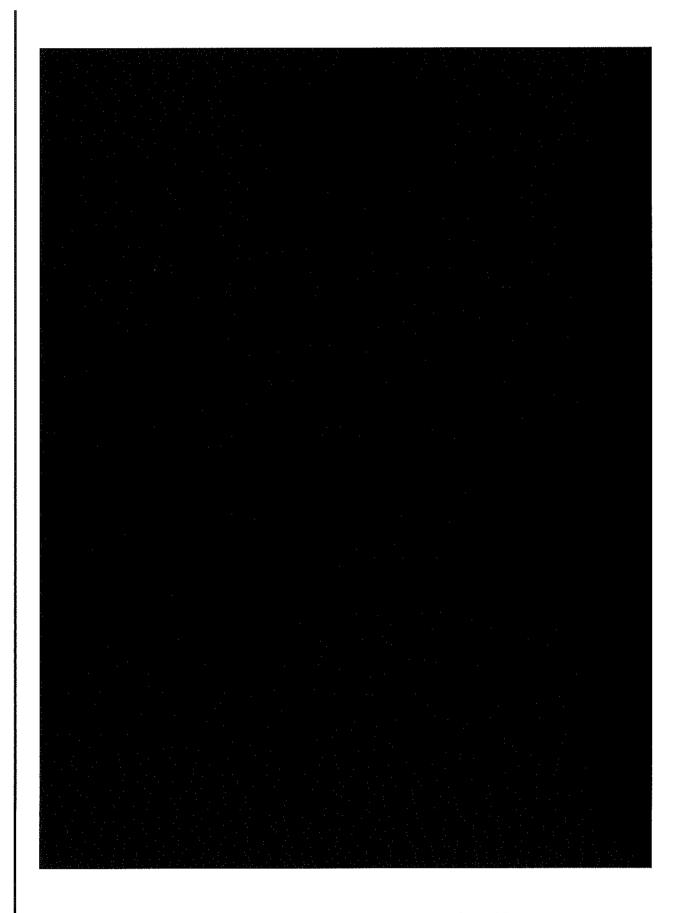




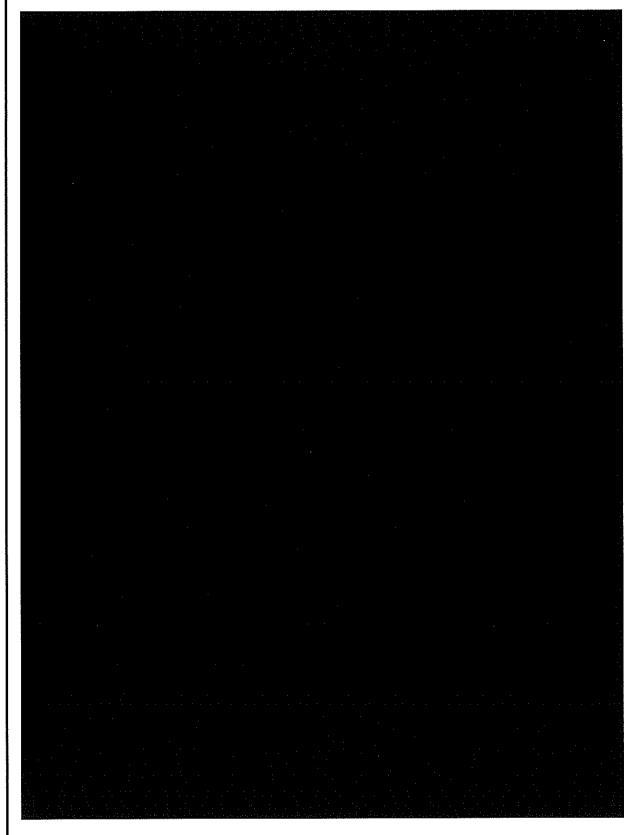


1 - 10 Fig 7 Deck Lock Harpoon

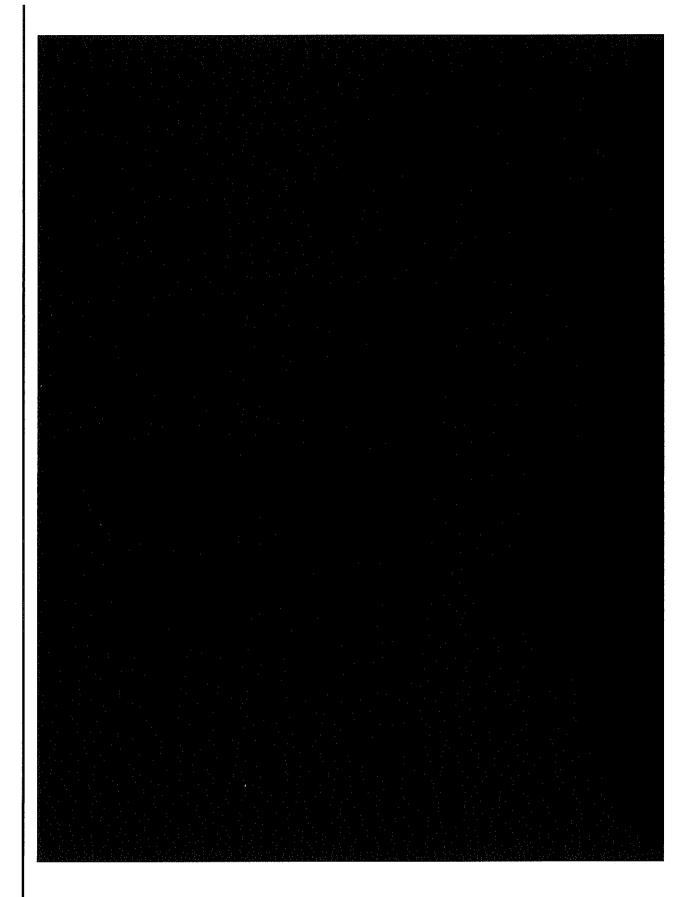




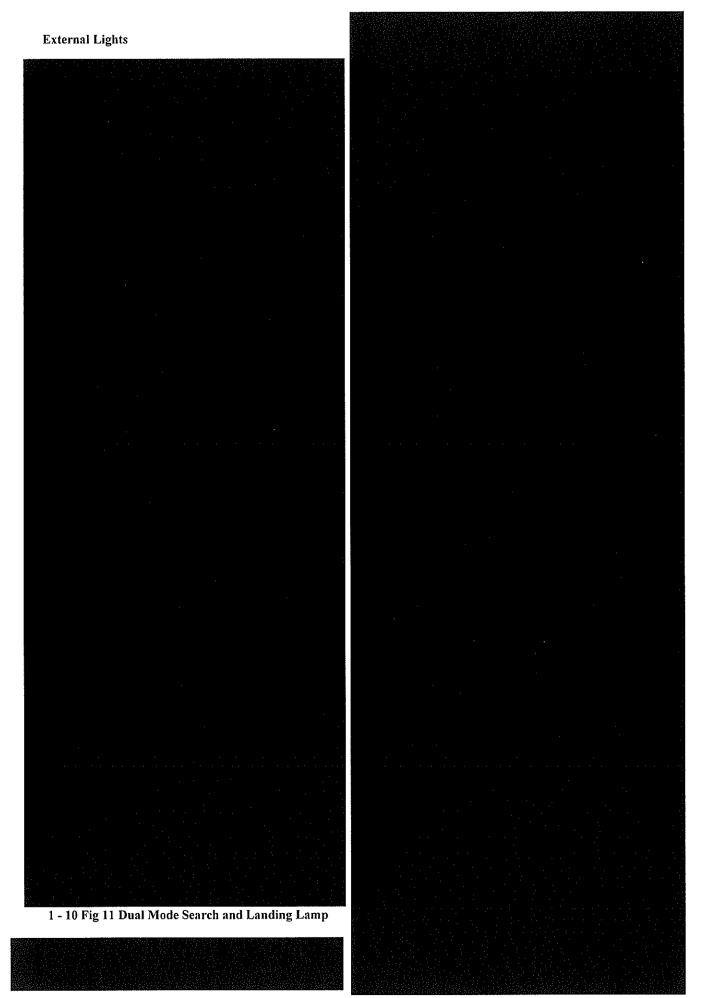
1 - 10 Fig 8 Non-NVG Lighting and Air Conditioning Controls, General Layout



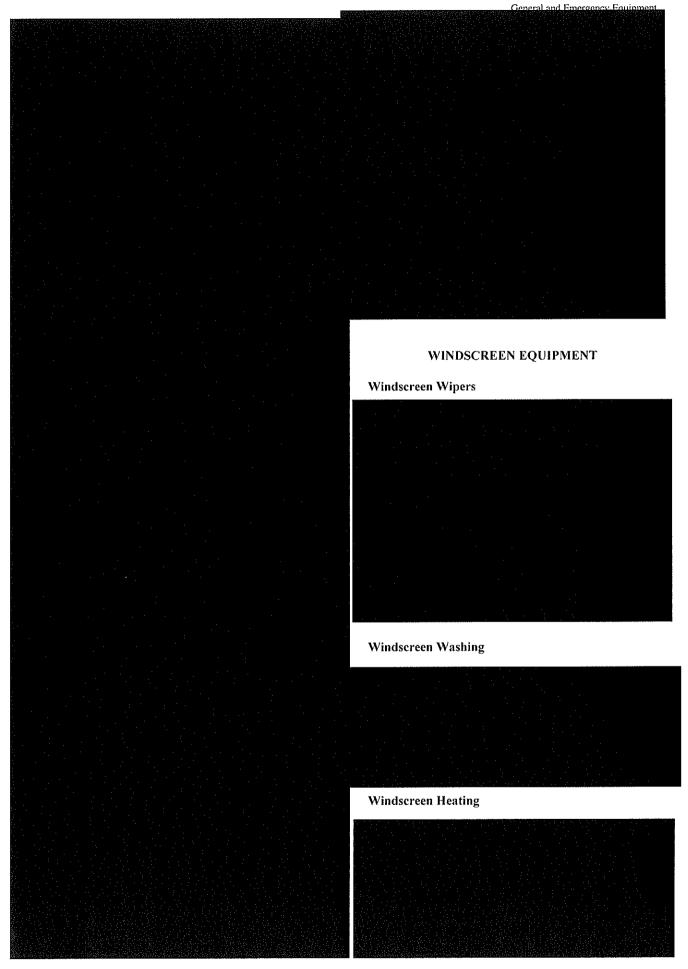
1 - 10 Fig 9 NVG Compatible Cockpit Lighting

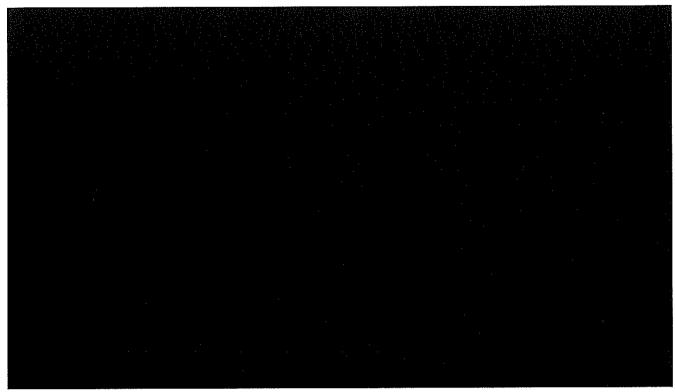


1 - 10 Fig 10 NVG Compatible Cockpit Lighting Filters

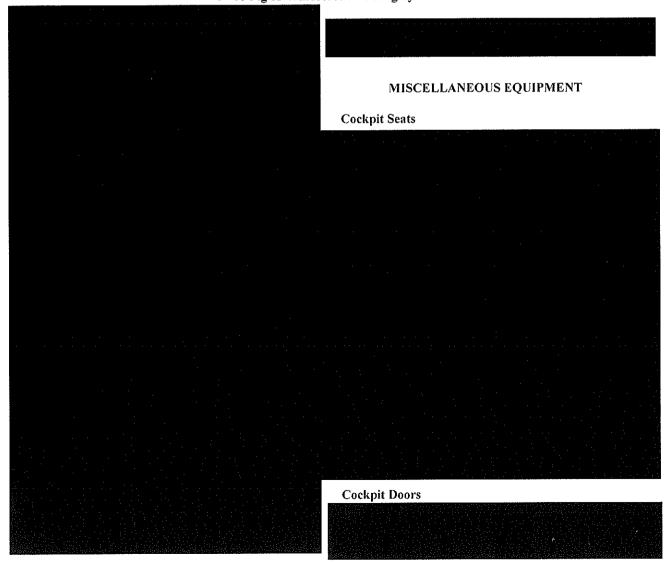


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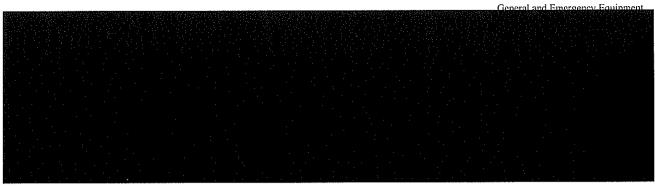


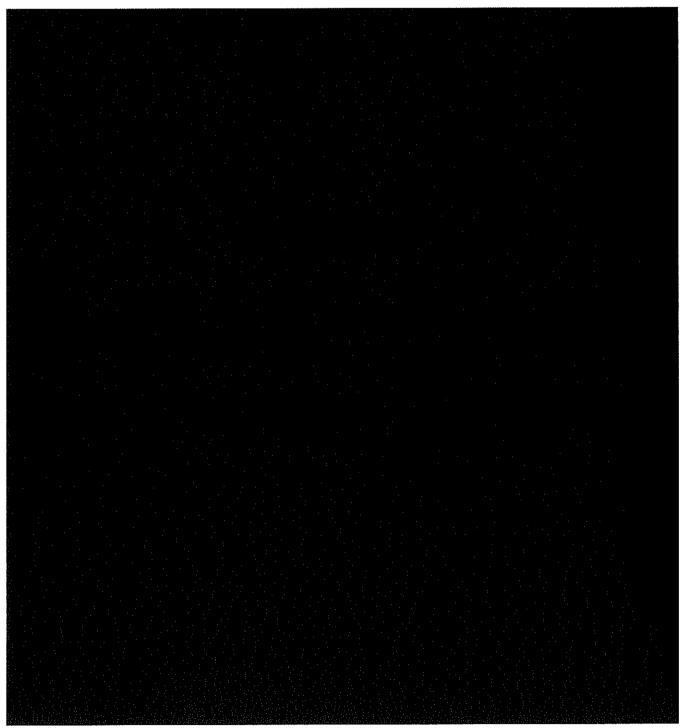
1 - 10 Fig 15 Windscreen Heating System



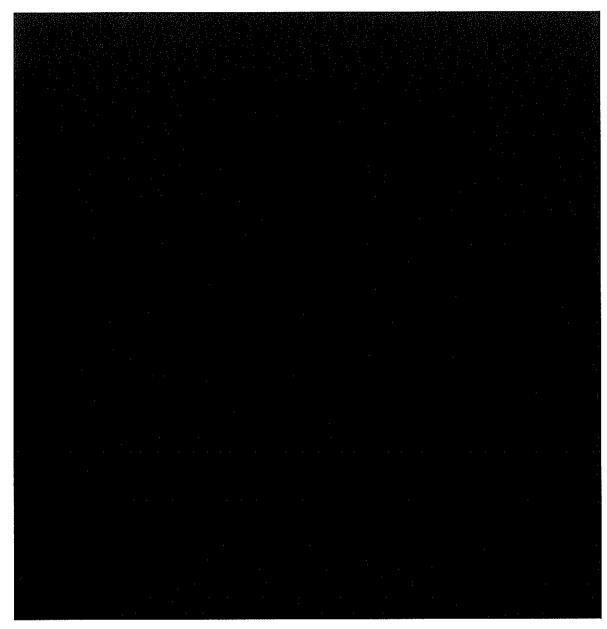
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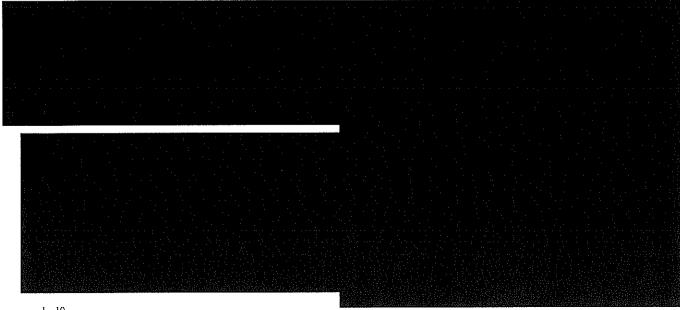
1 - 10 Fig 16 Cockpit Doors



1 - 10 Fig 17 Cabin Doors

#### Cabin Doors

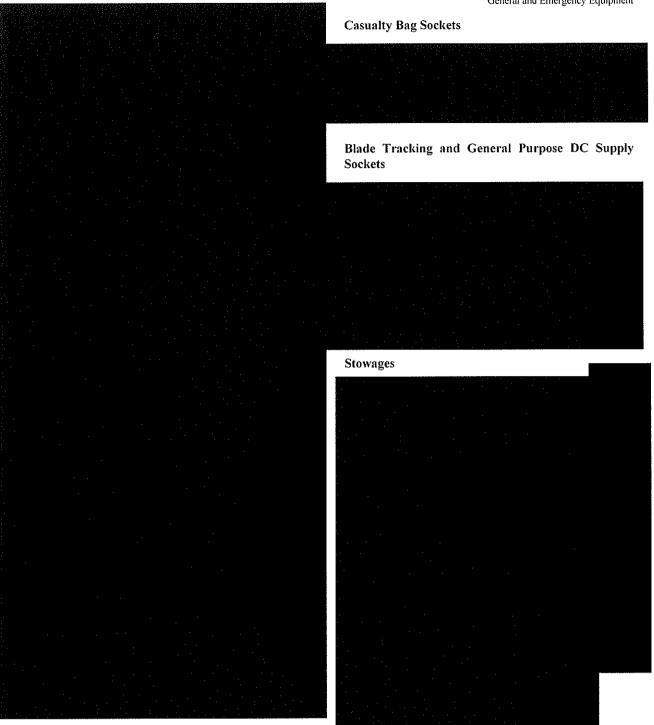
### Heating and Ventilation

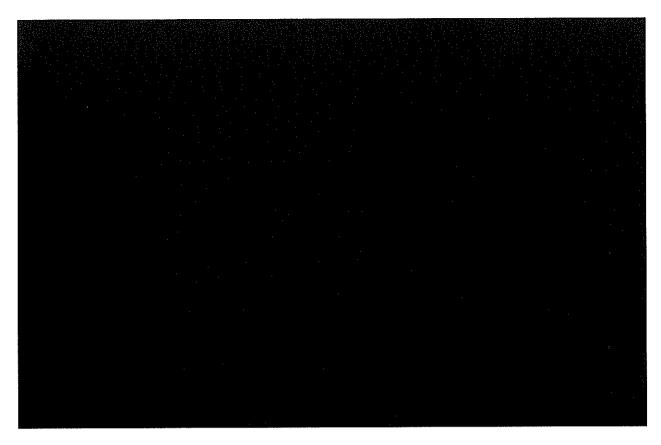


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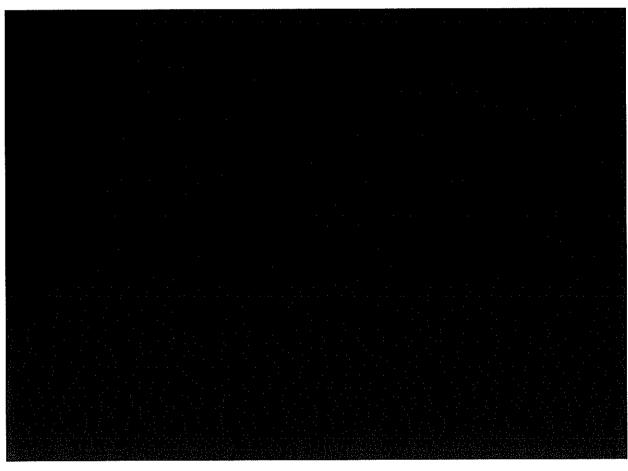
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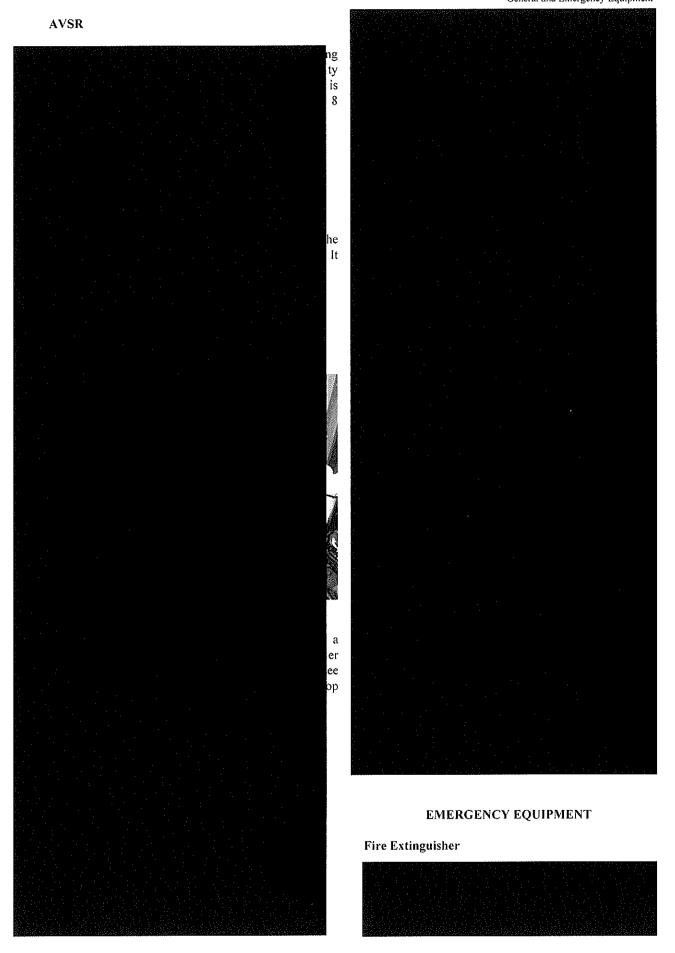


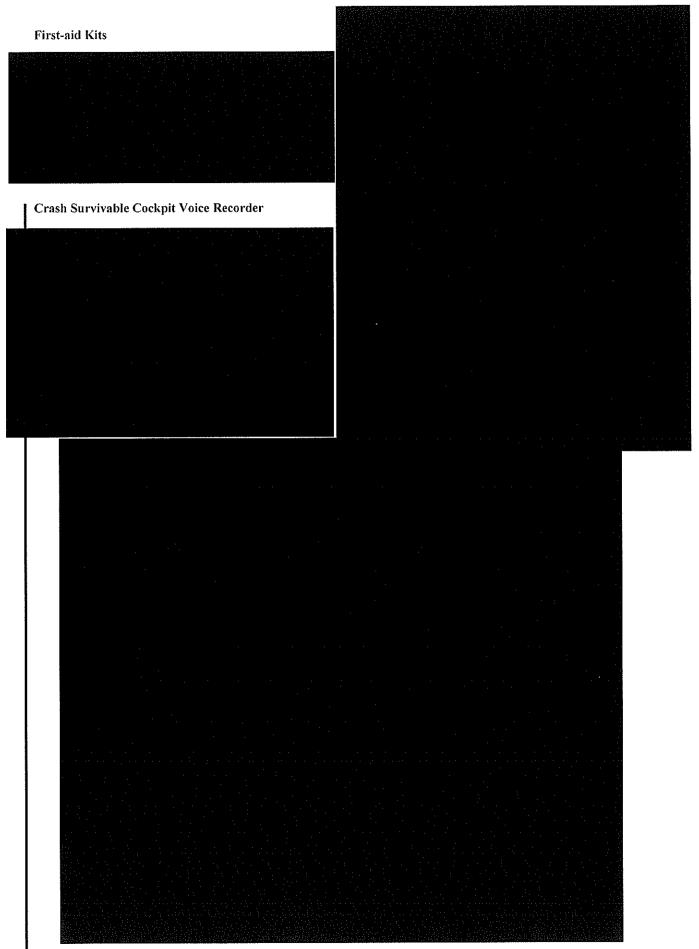


1 - 10 Fig 18 Heating and Ventilation System



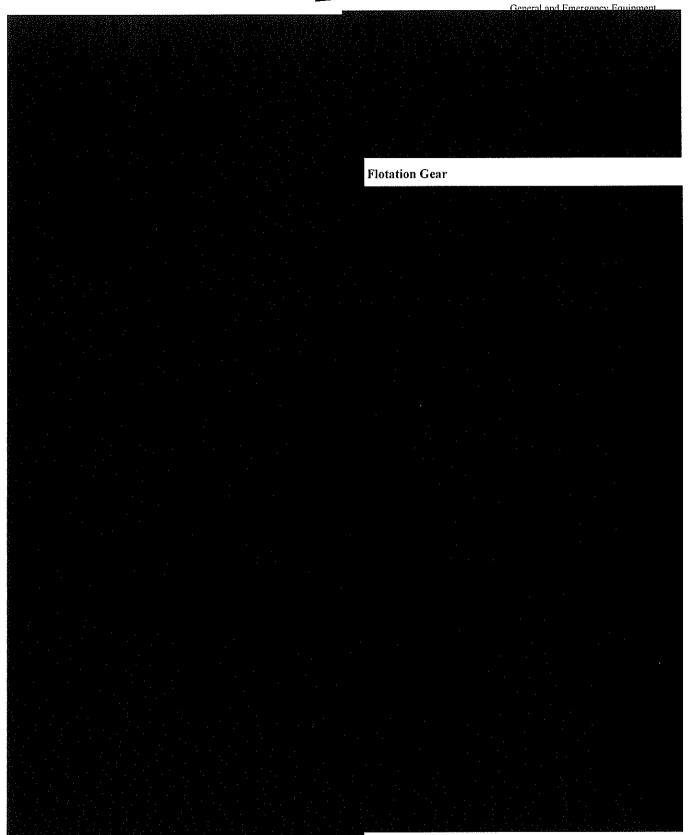
1 - 10 Fig 19 Stowages and Emergency Equipment

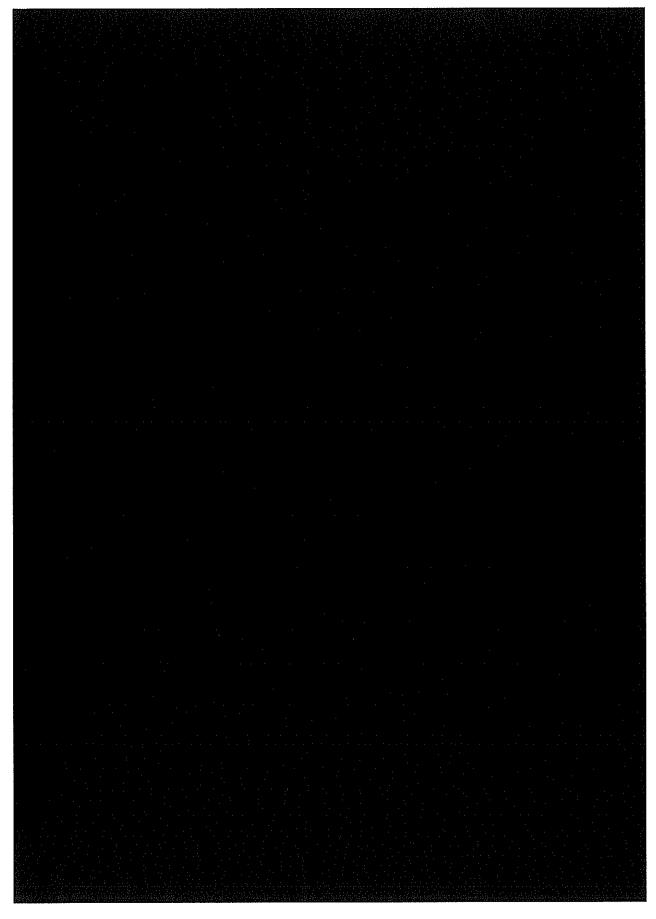




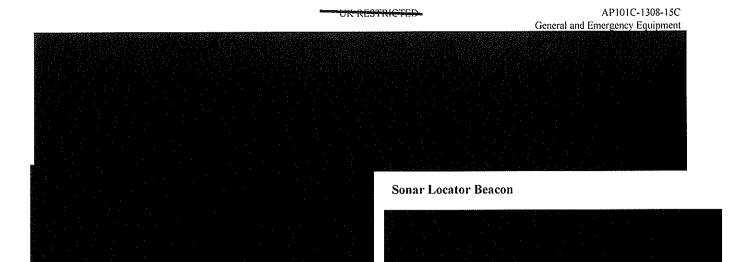
1 - 10 Fig 23 Microphone Box with Controls and Indicator

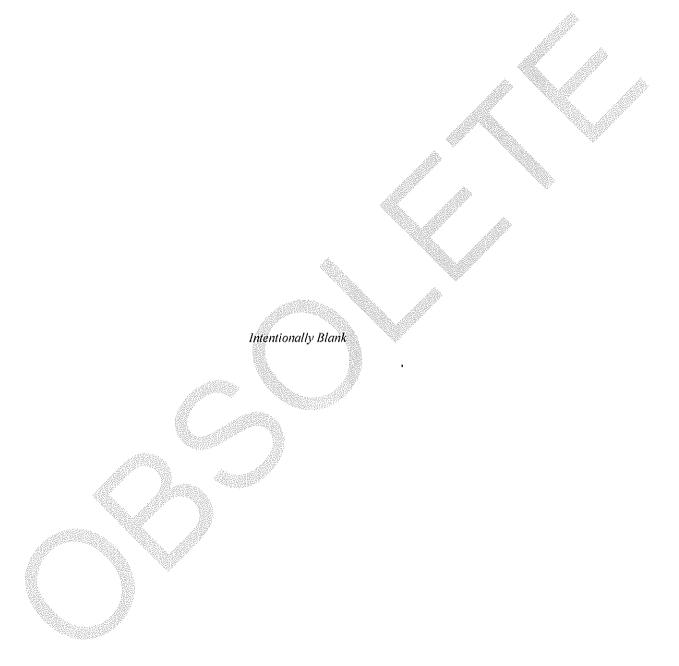
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1 - 10 Fig 25 Flotation Gear Electrical System





### PART 1

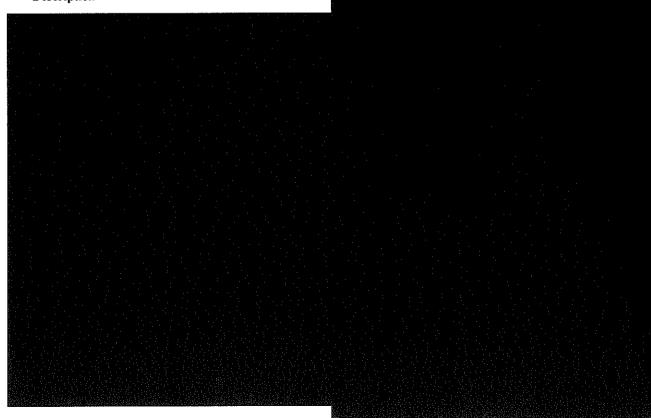
# CHAPTER 11

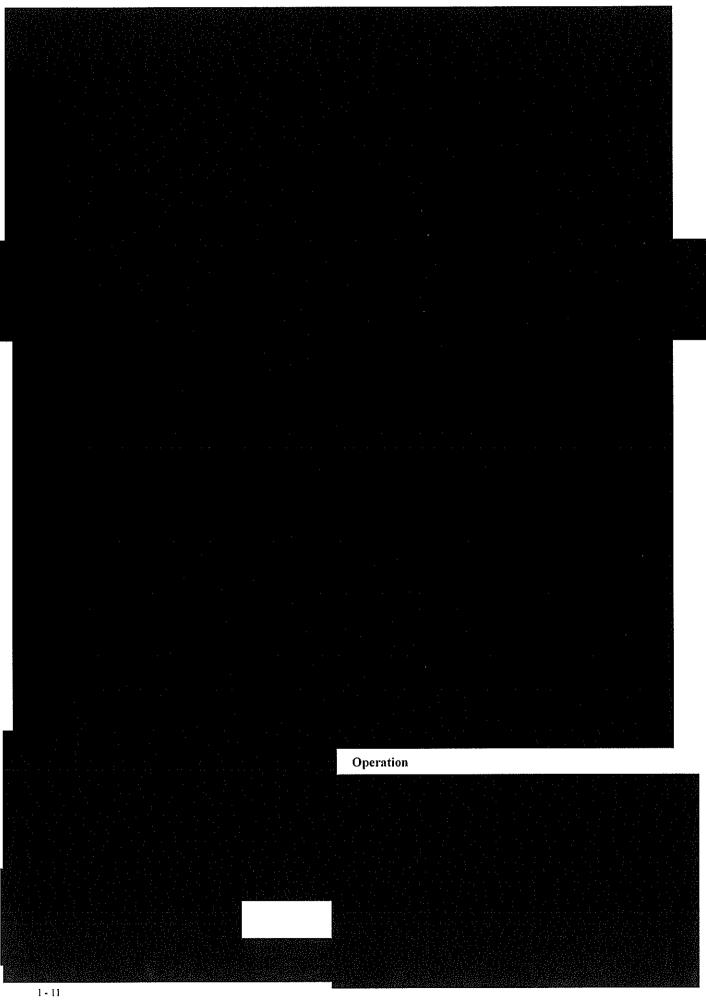
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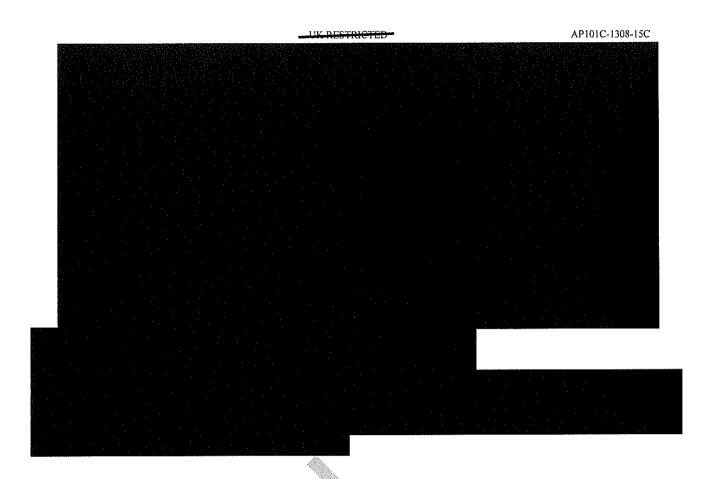
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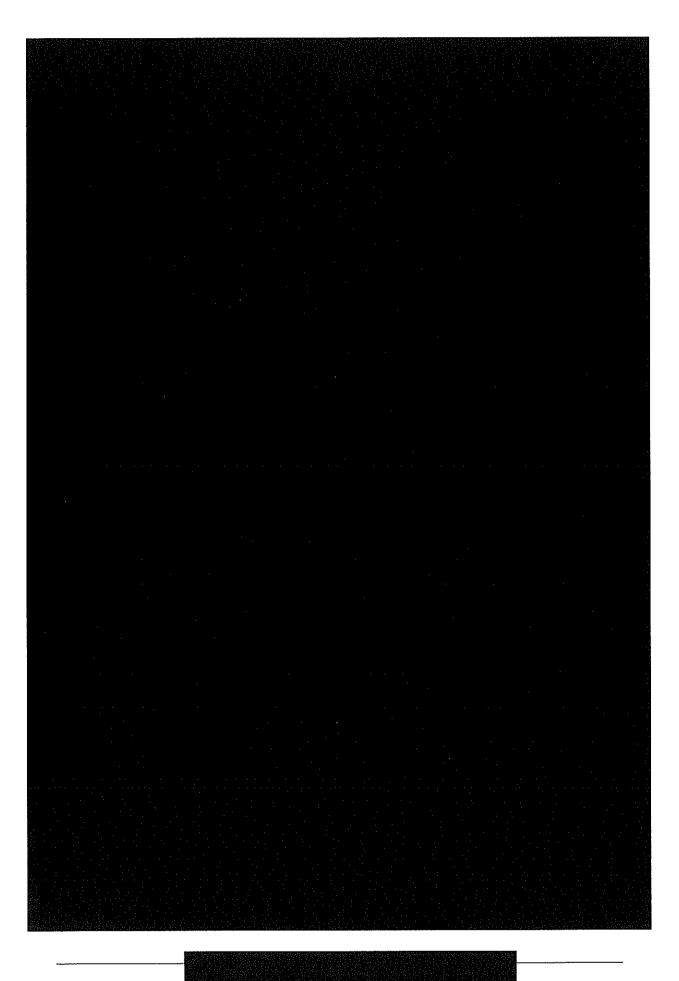
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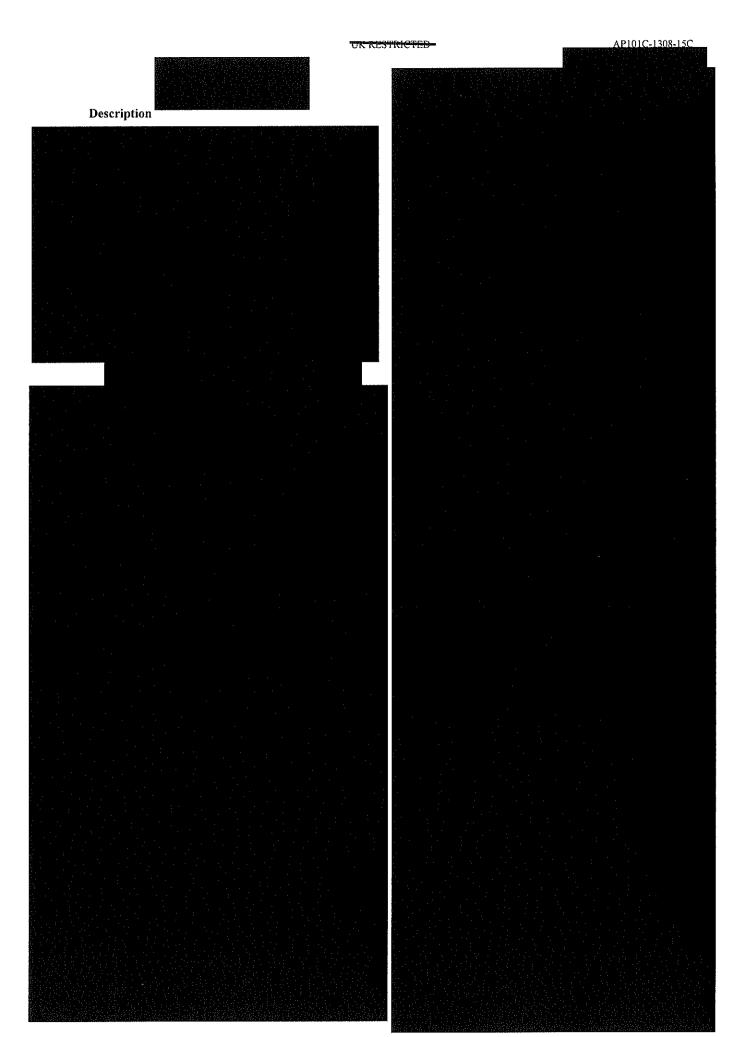
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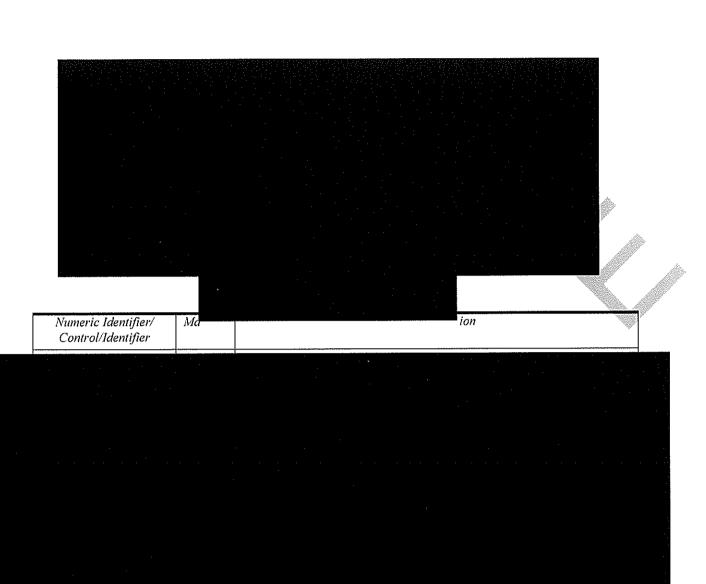


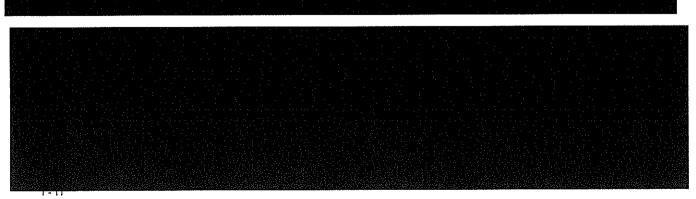




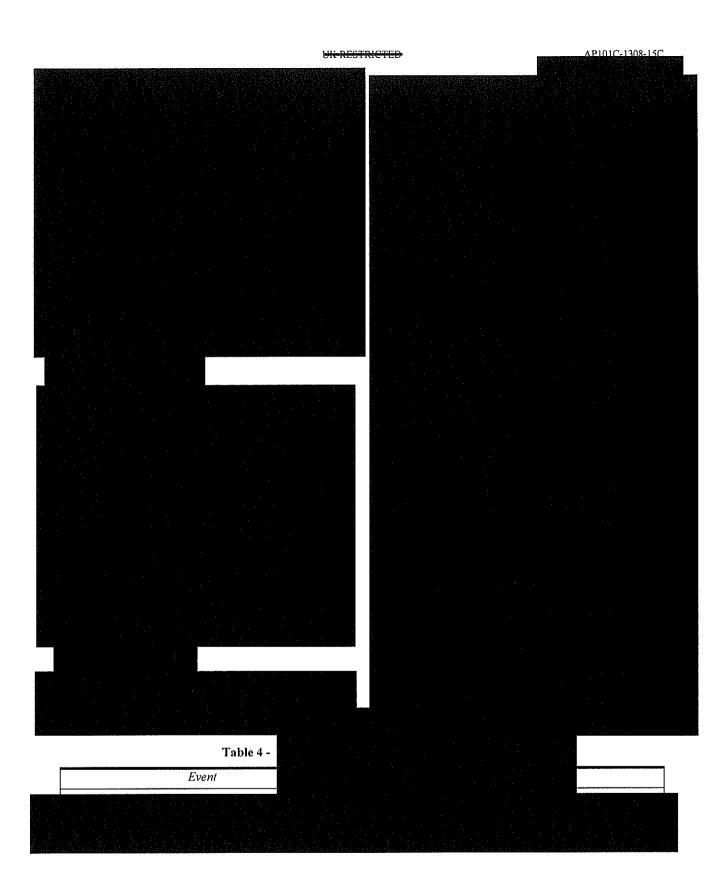


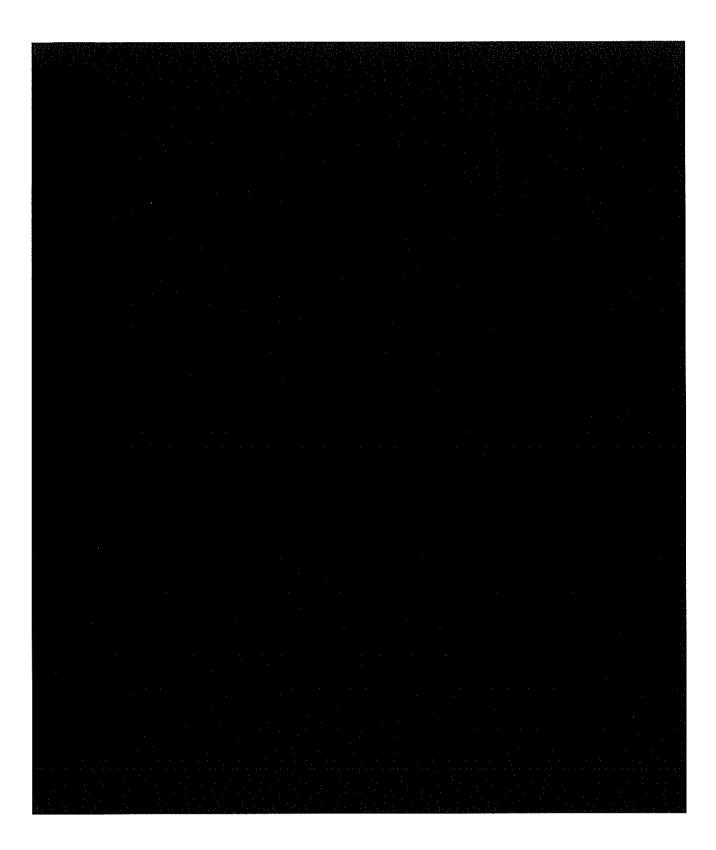


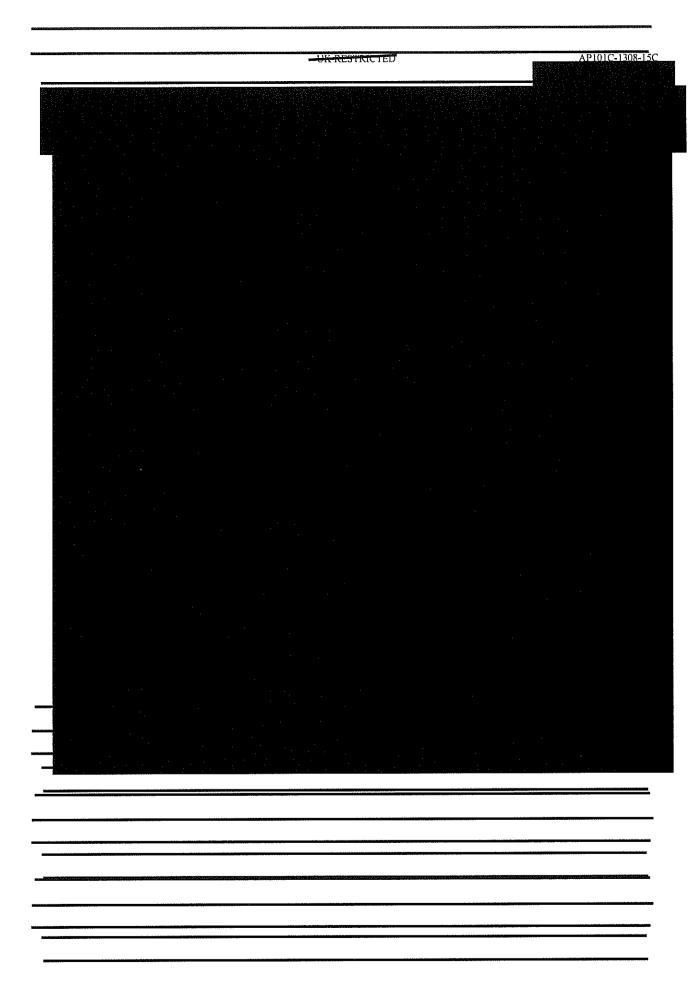


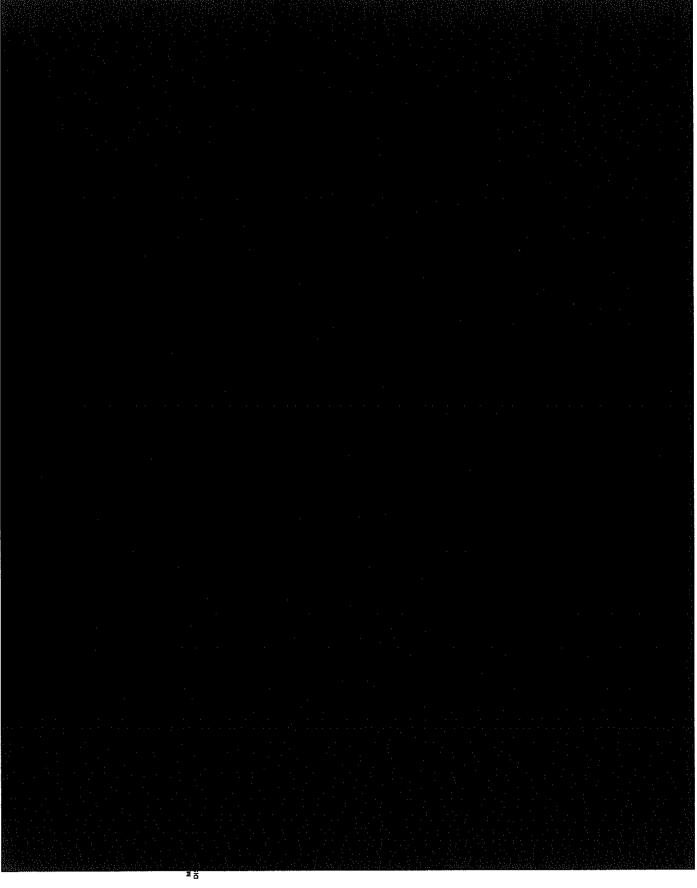


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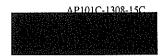


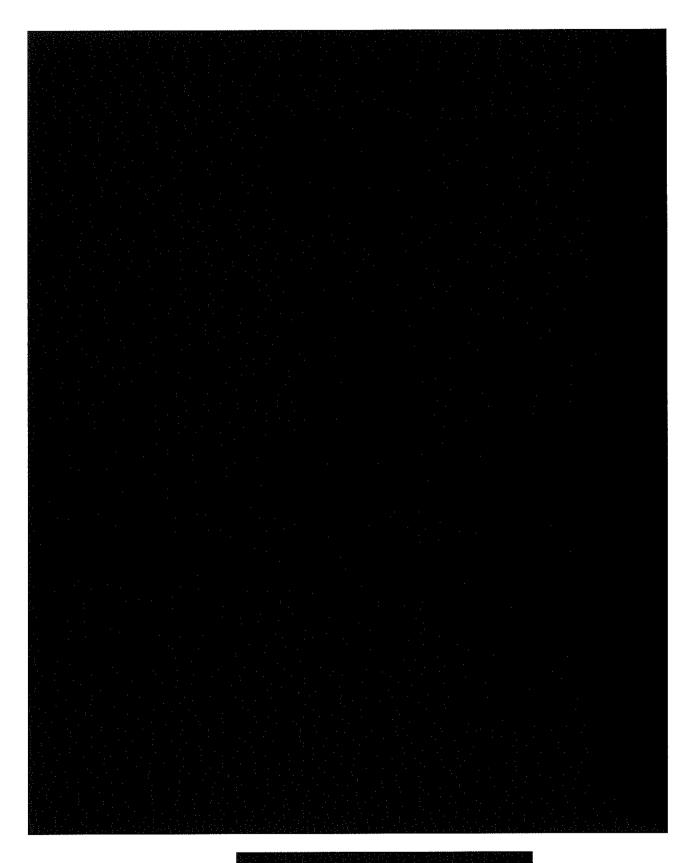




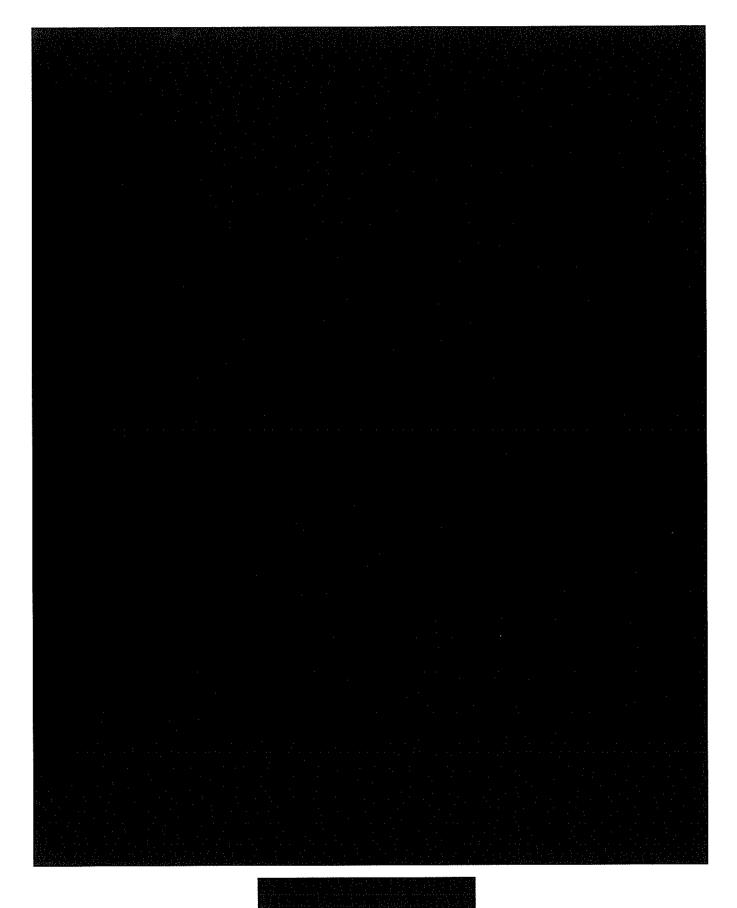


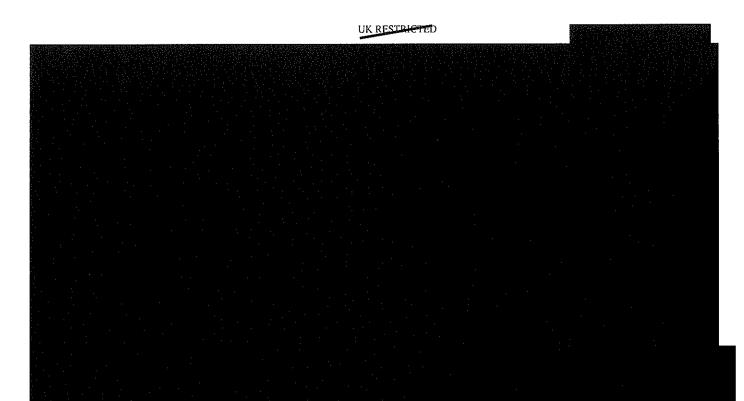
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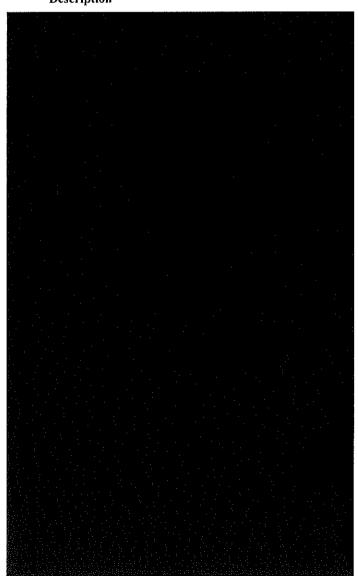


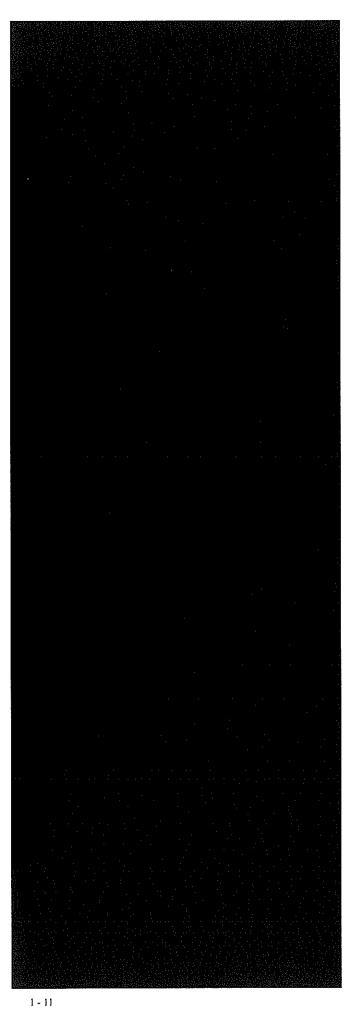
1 - 11 Fig

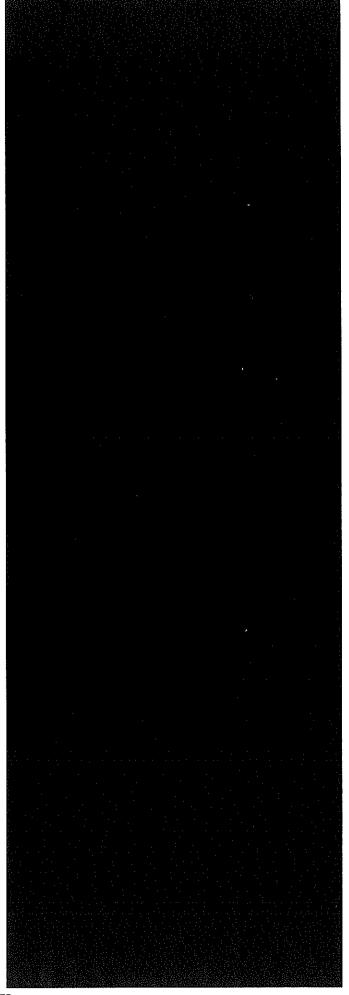


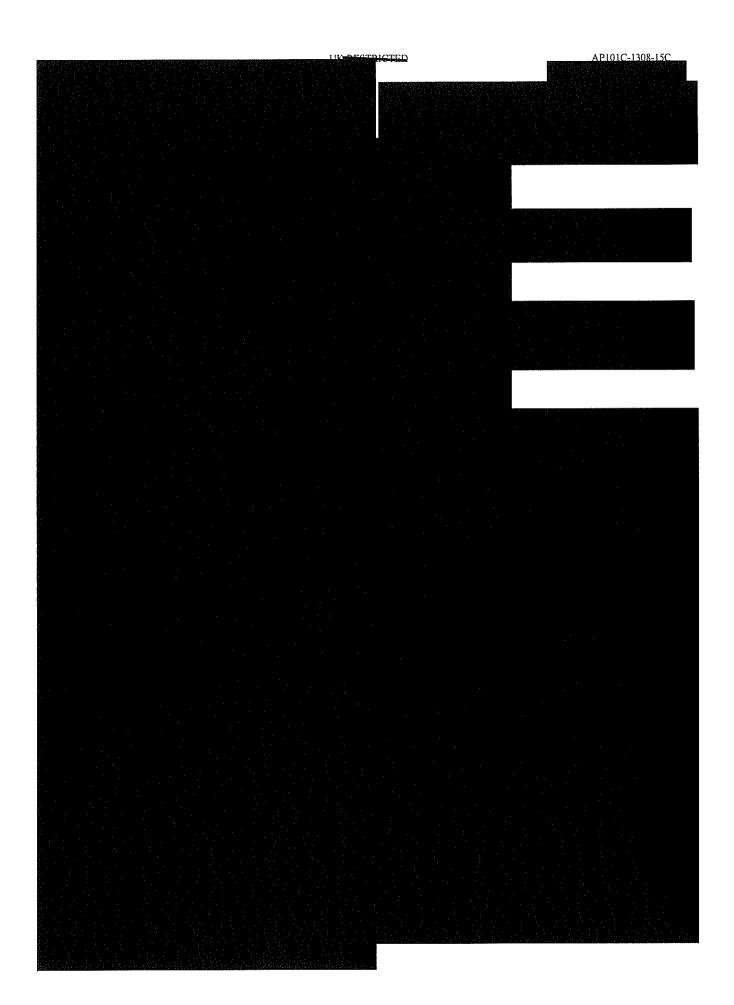


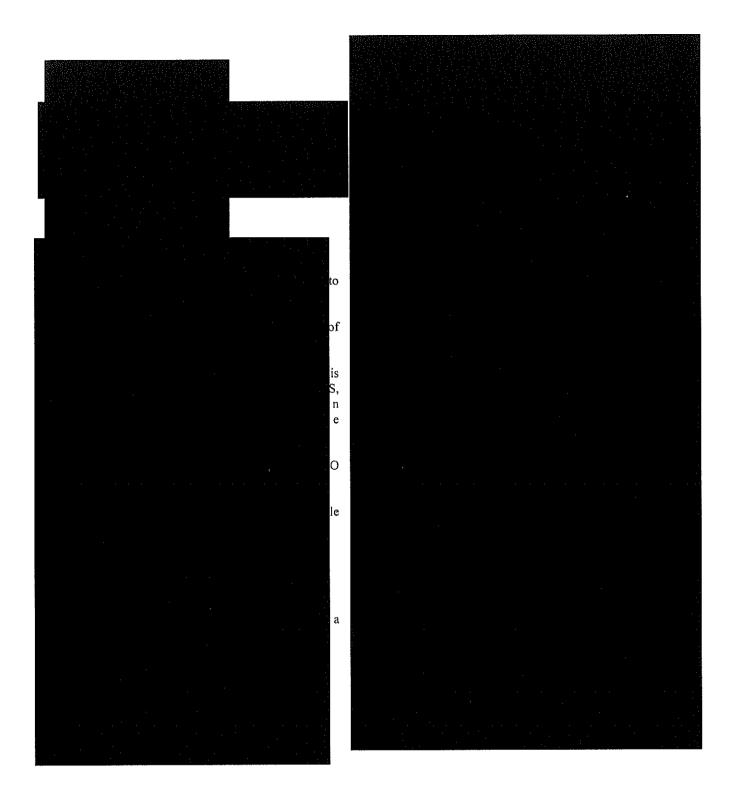
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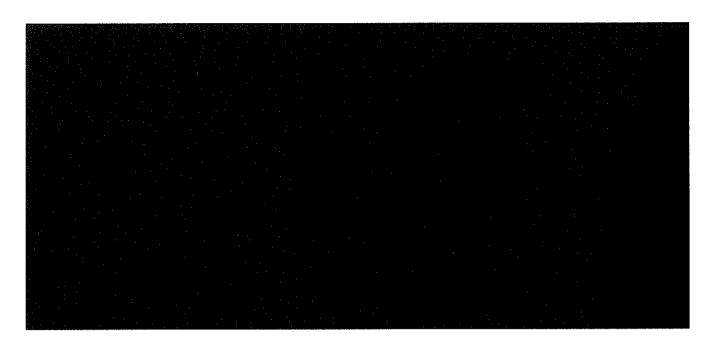




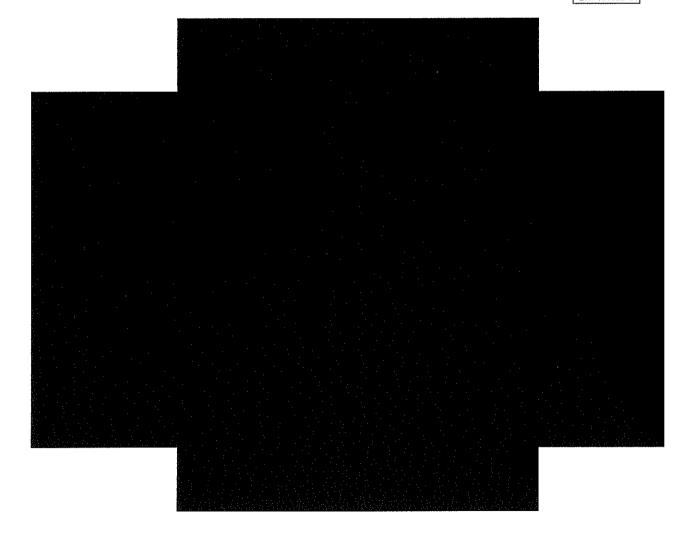


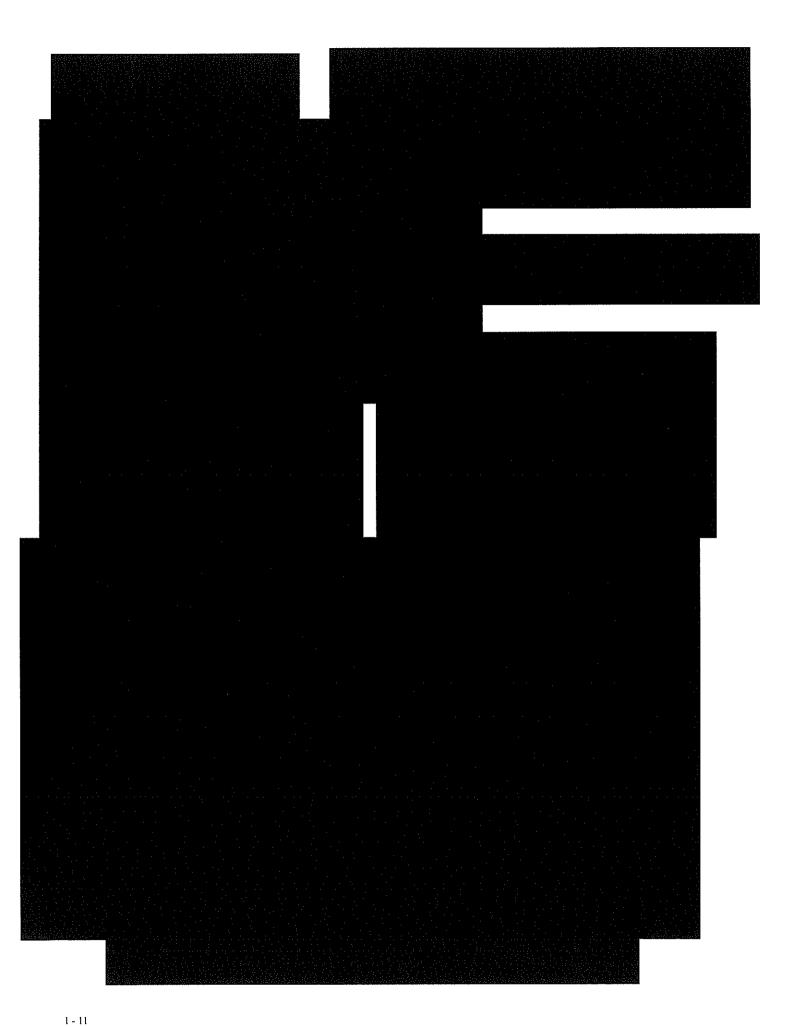


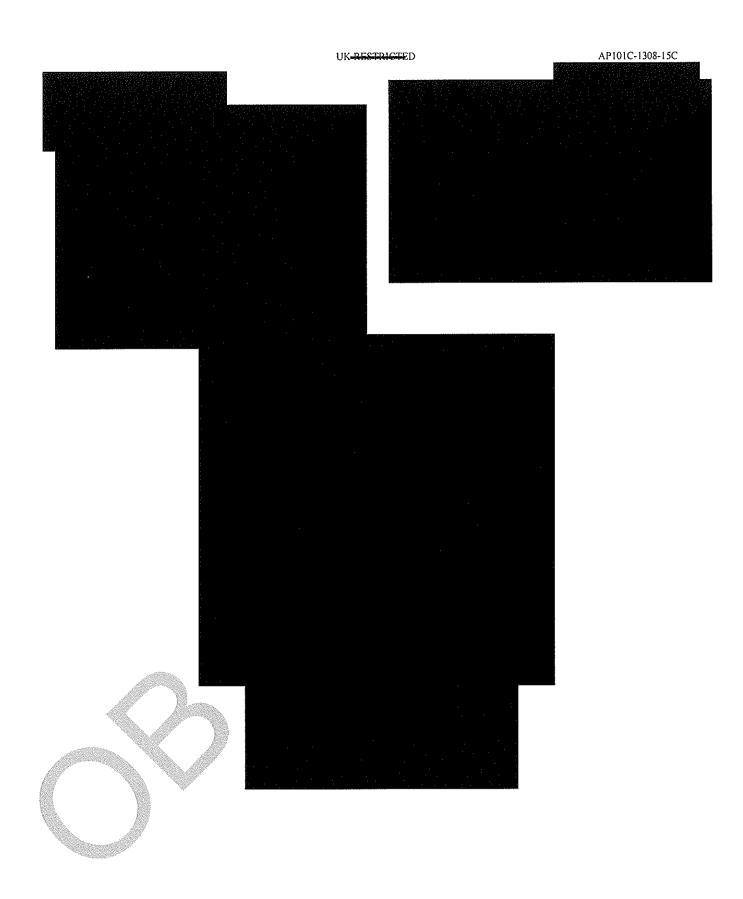


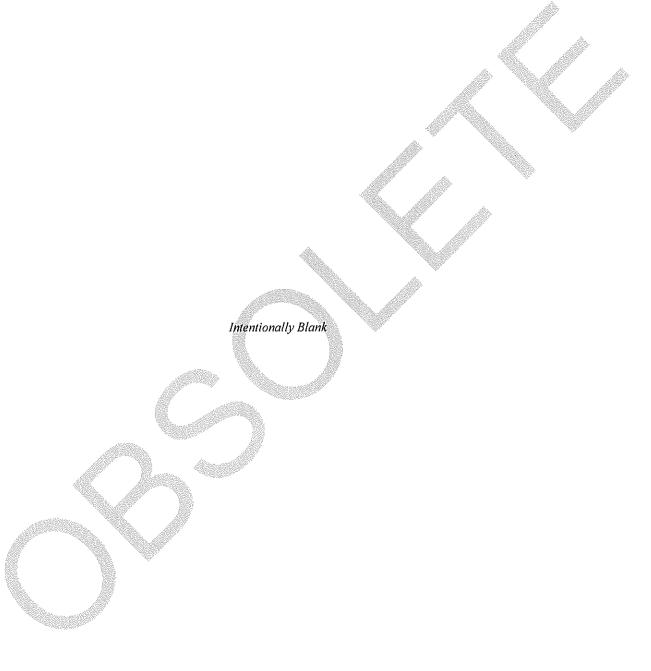


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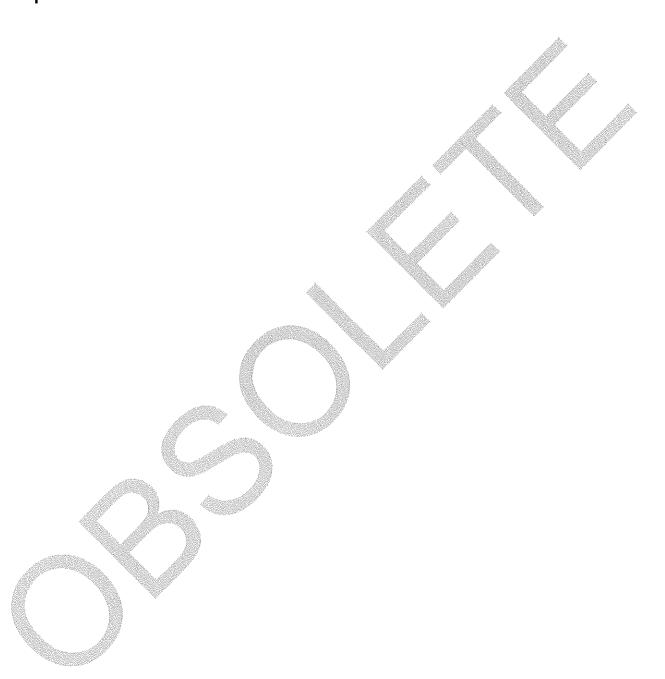
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## **CHAPTER 1 - AIRCRAFT LIMITATIONS**

Refer to the Lynx HMA Mk 8 Release-to-Service for information on aircraft limitations.

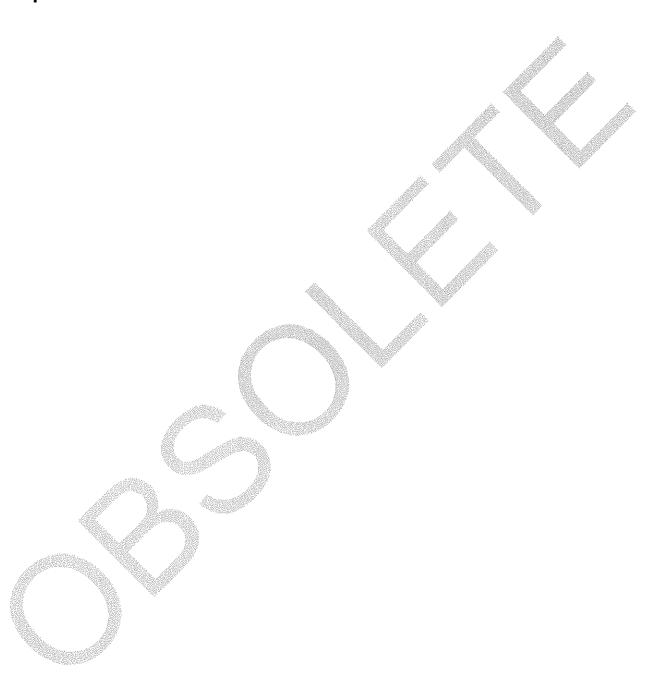
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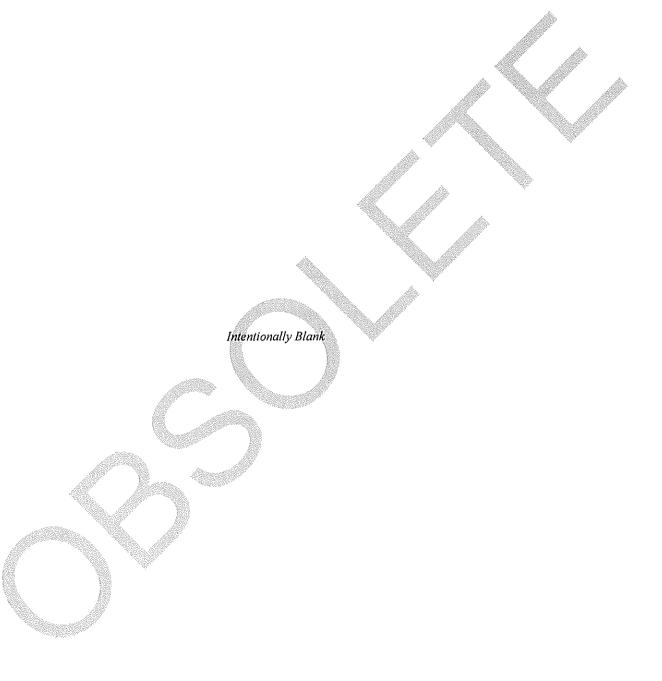


## **CHAPTER 2 - ENGINE, TRANSMISSION AND ROTOR LIMITATIONS**

Refer to the Lynx HMA Mk 8 Release-to-Service for information on engine, transmission and rotor limitations.

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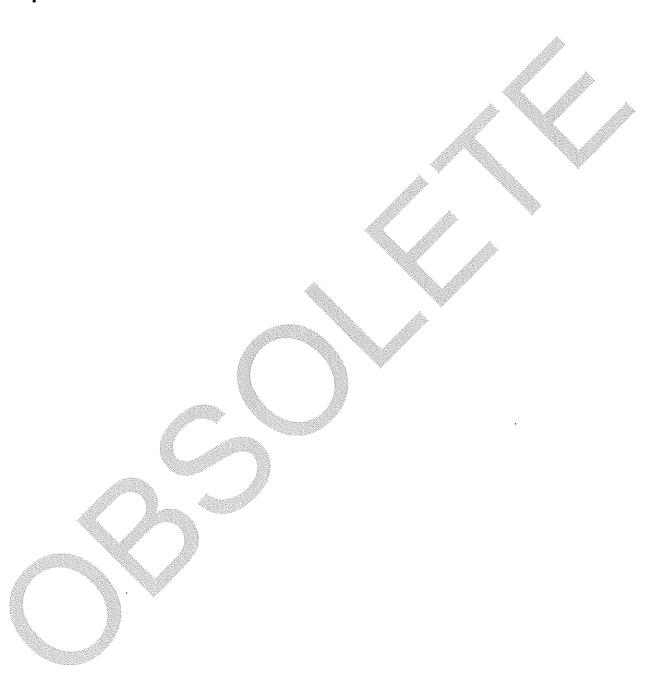
# **CHAPTER 3 - SYSTEMS LIMITATIONS**

Refer to the Lynx HMA Mk 8 Release-to-Service for information on systems limitations.

## **CHAPTER 4 - AVIONICS AND ROLE EQUIPMENT LIMITATIONS**

Refer to the Lynx HMA Mk 8 Release-to-Service for information on role equipment limitations.

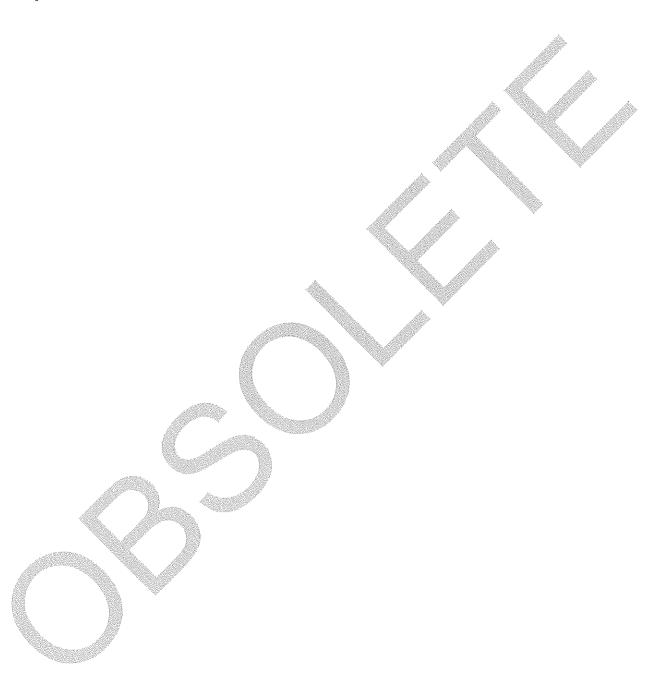
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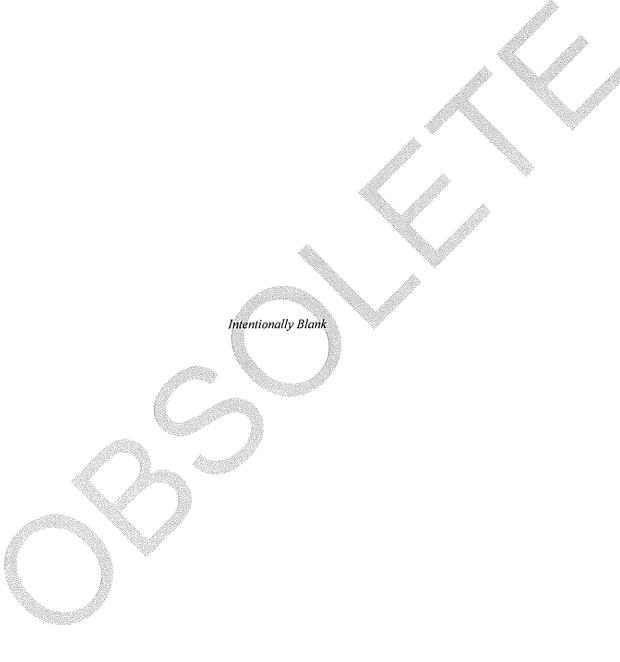


## **CHAPTER 5 - CENTRAL TACTICAL SYSTEM LIMITATIONS**

Refer to the Lynx HMA Mk 8 Release-to-Service for information on central tactical system limitations.







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### **CHAPTER 1 - PREPARATION FOR FLIGHT**

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#### **GENERAL**

#### General

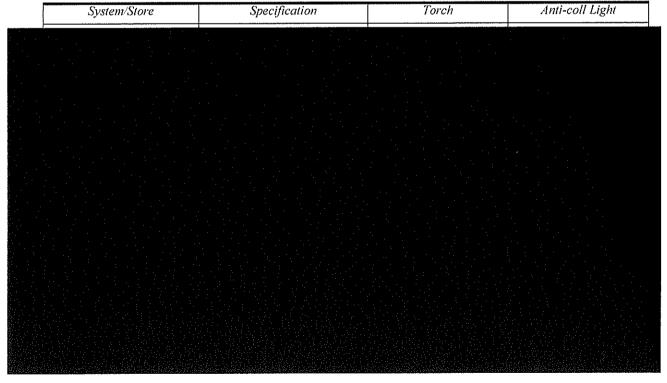
1. The checks referred to in this Part are listed in the FRC and are annotated in **bold print**.

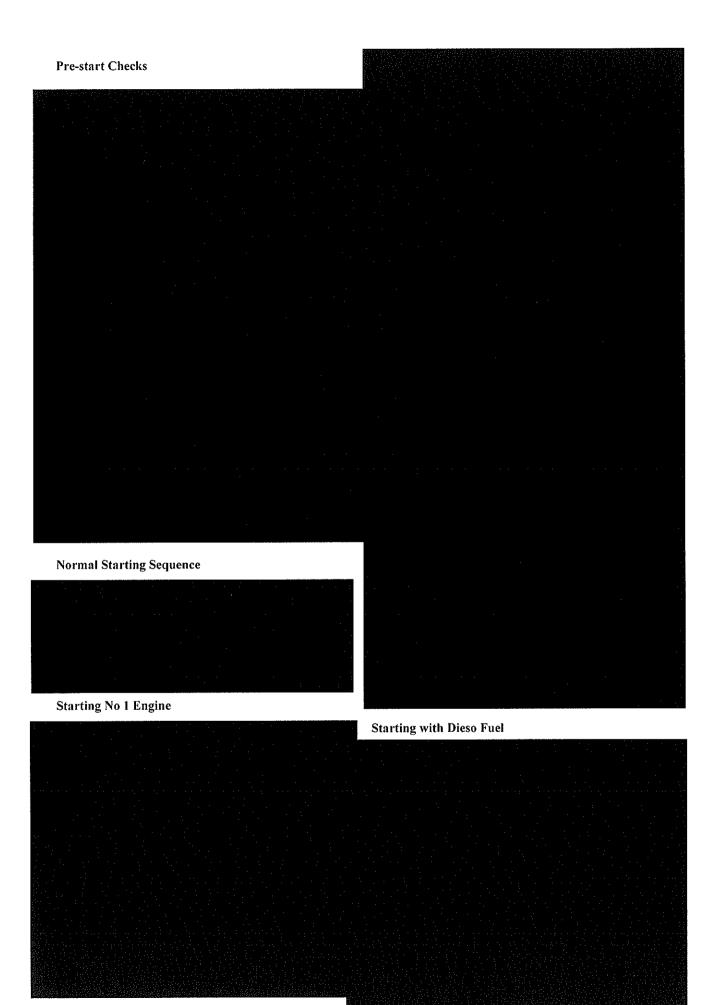
**CAUTION:** When moving to and from the cockpit, take care not to foul the switches on the roof panel with the helmet or visor.

## Night Signals

2. The signals to be used when operating at night are shown in Table 1.

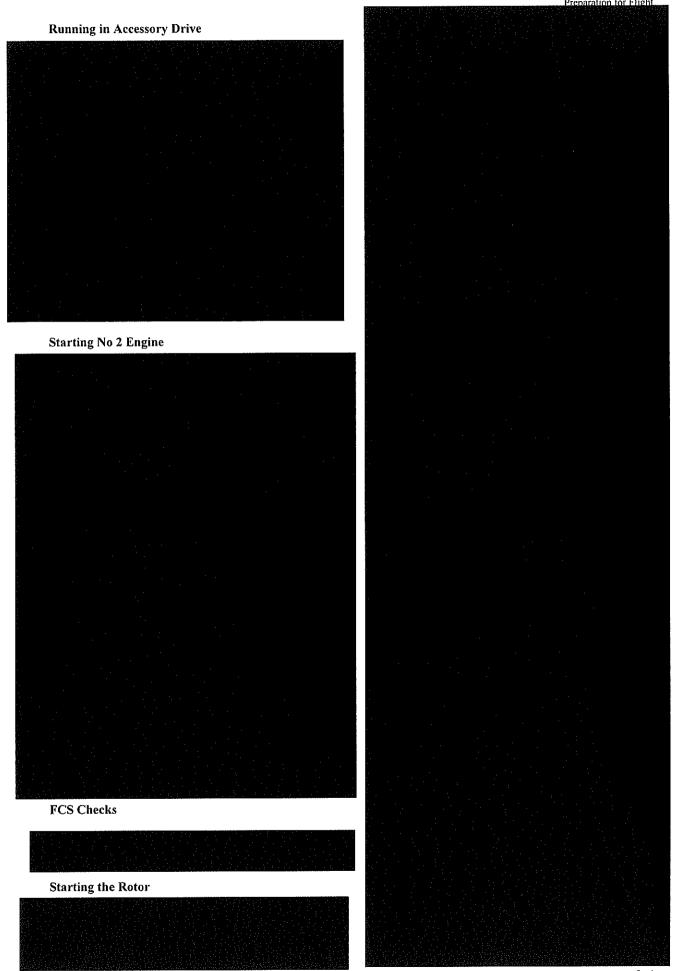
Table 1 - Night Signals

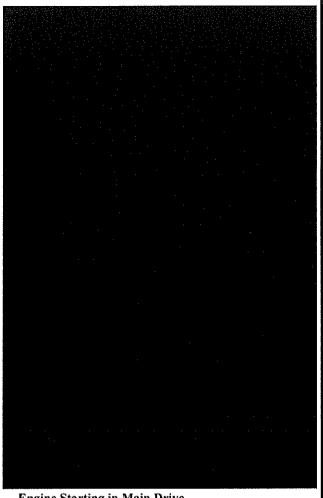




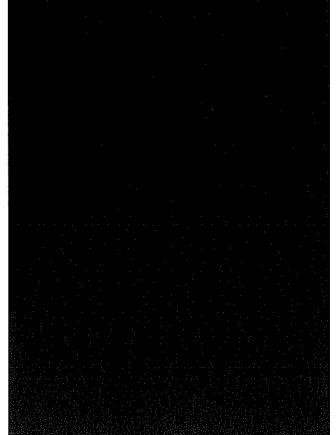






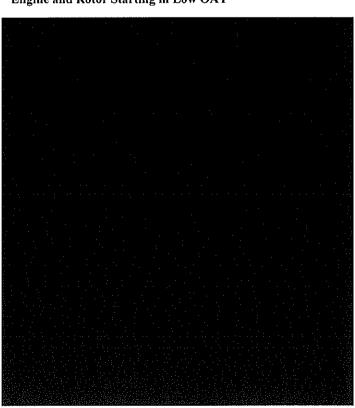


Engine Starting in Main Drive

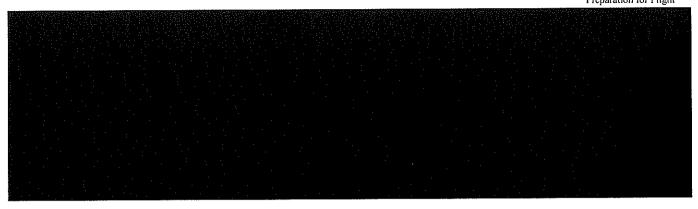


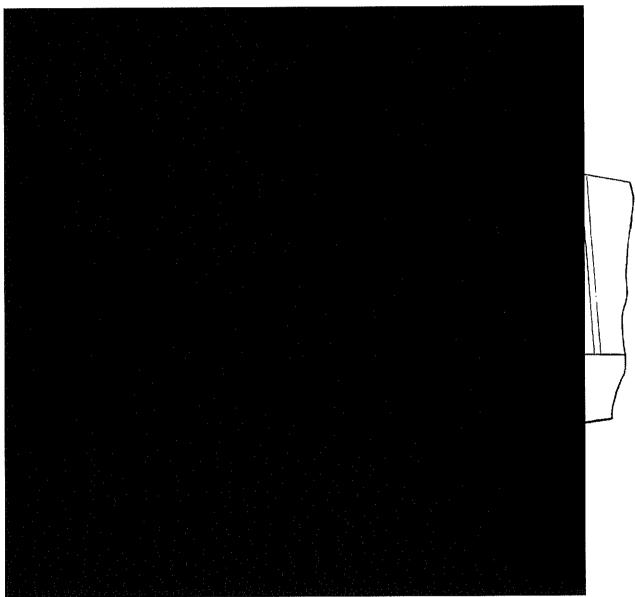
engine.

## Engine and Rotor Starting in Low OAT



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## **CHAPTER 2 - HANDLING IN FLIGHT**

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### **GENERAL**

#### General

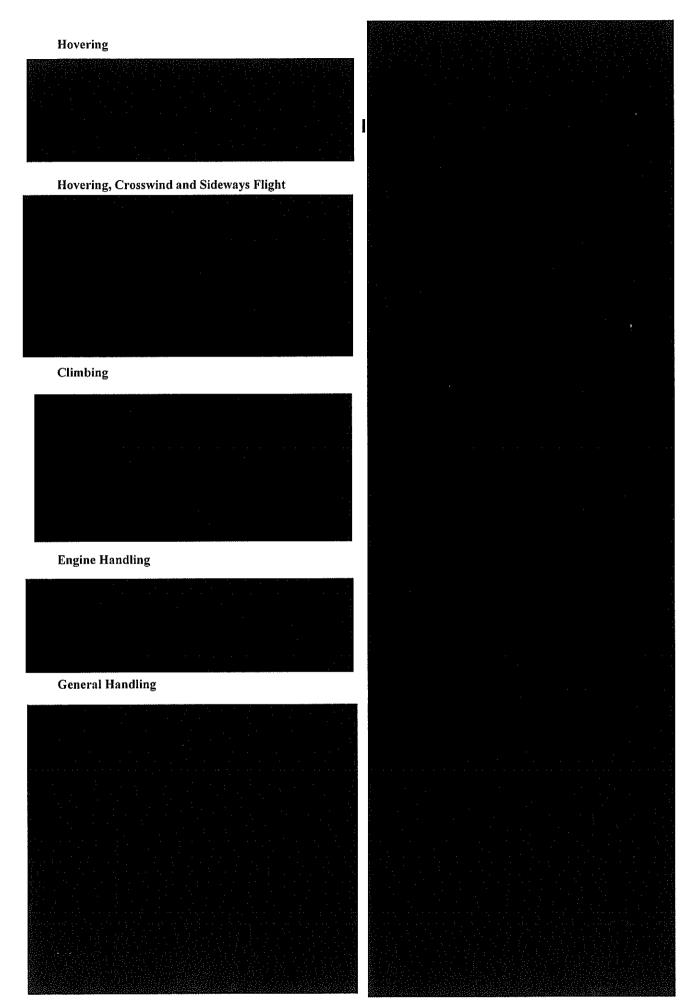


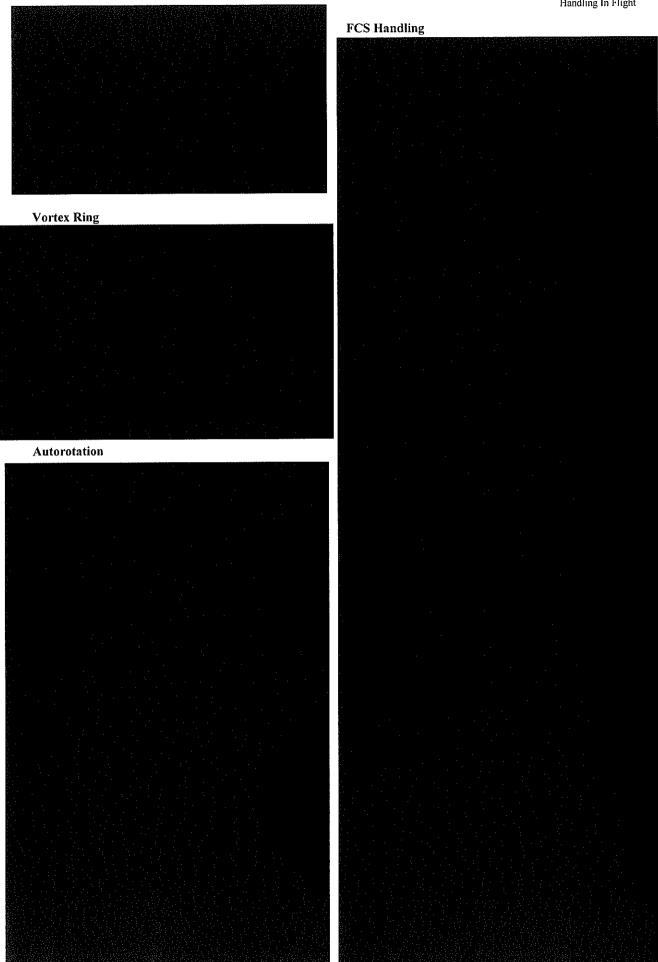


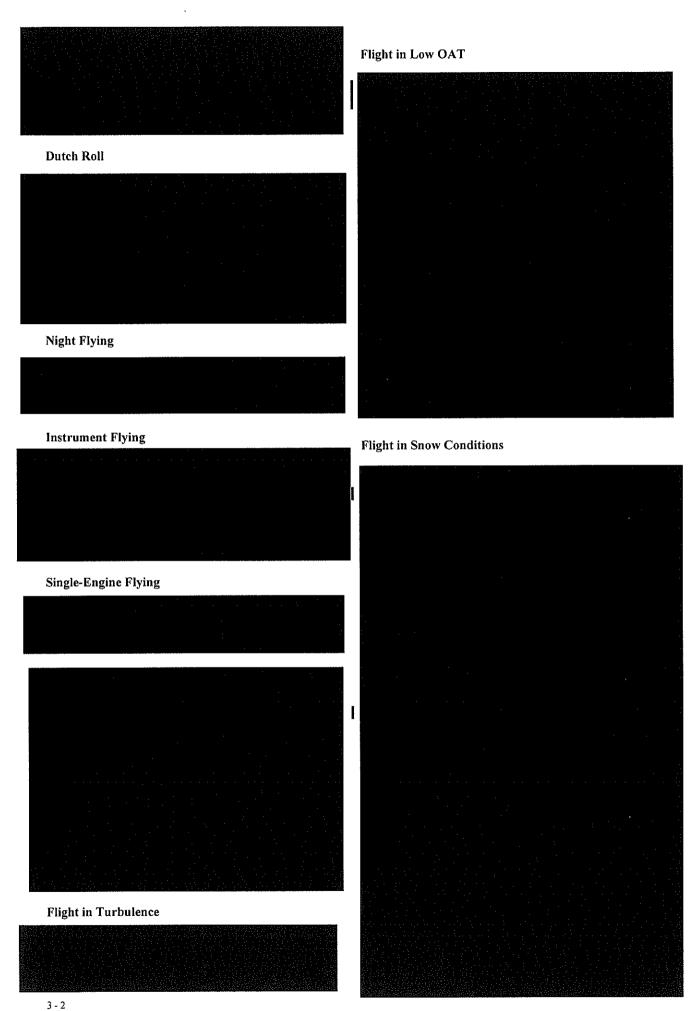


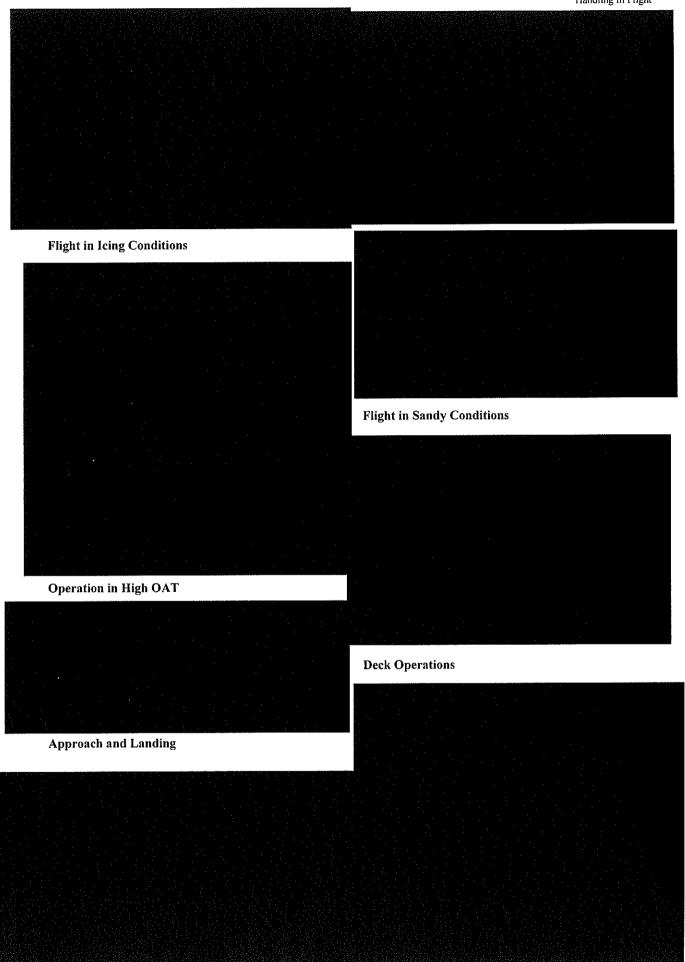


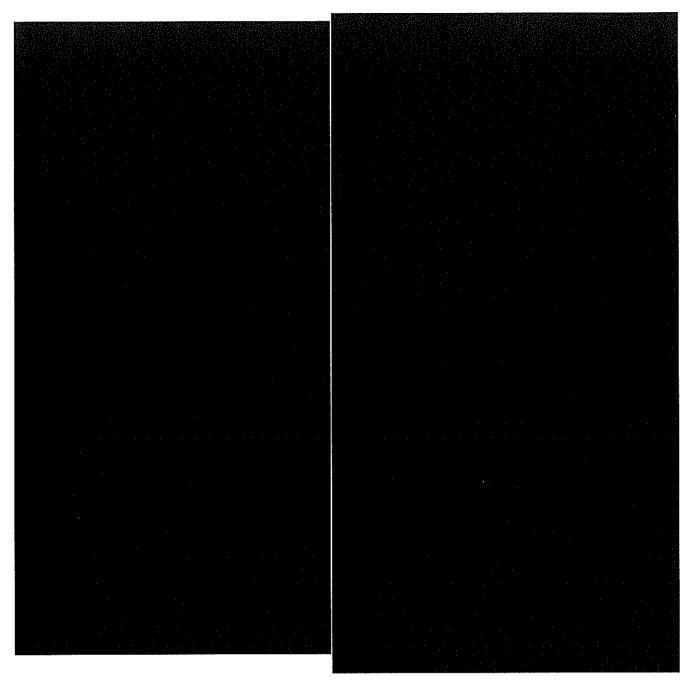
Take-On











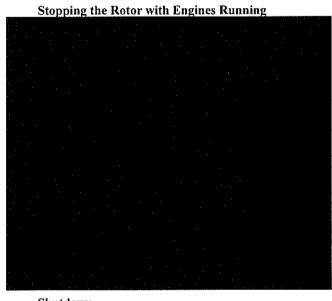


## **CHAPTER 3 - SHUTTING DOWN**

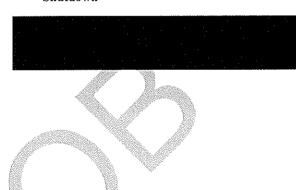
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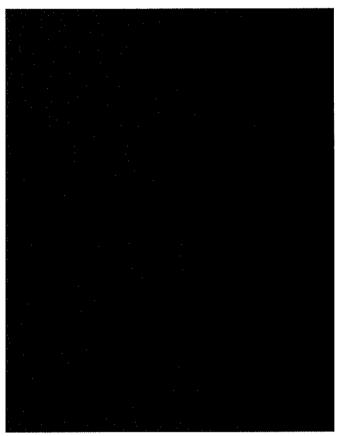
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### **GENERAL**









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## **CHAPTER 1 - ANALYSIS OF FAILURES**

(Completely revised at AL 2)

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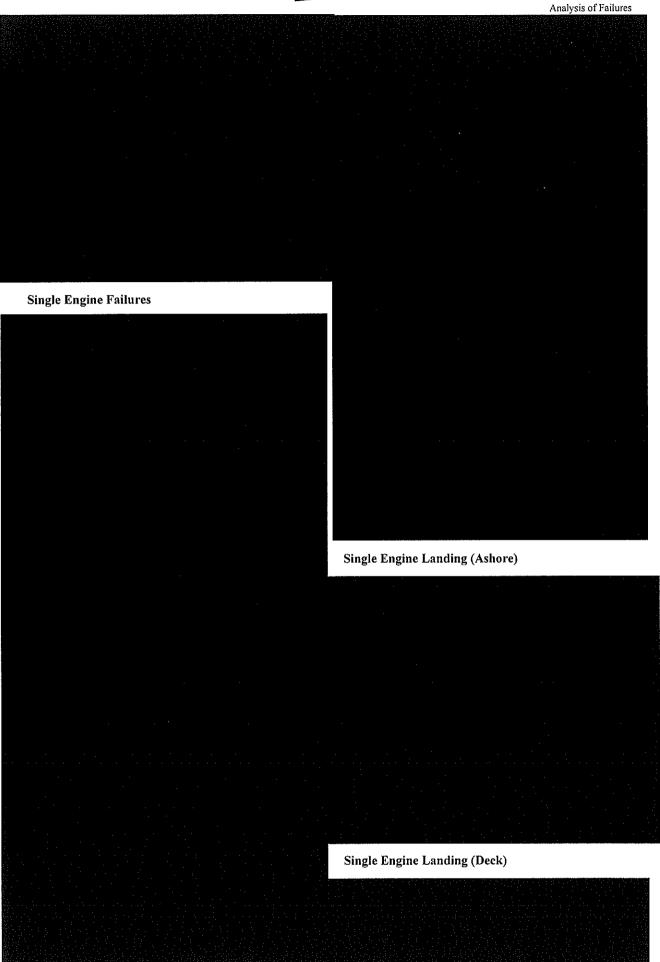
General

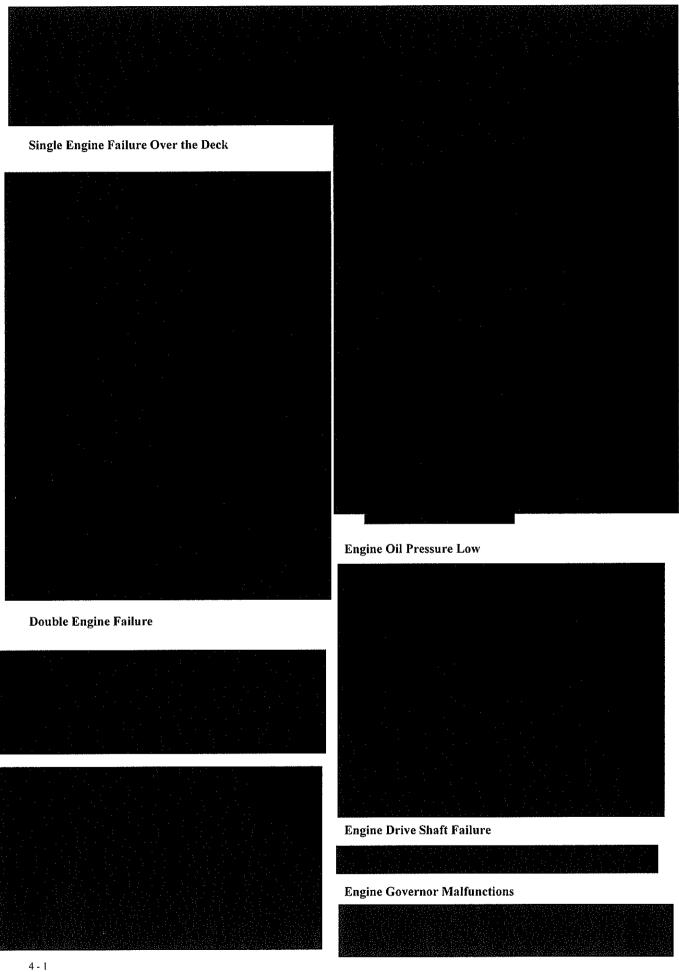
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**Table 1 - Locations of Malfunctioning Procedures** 

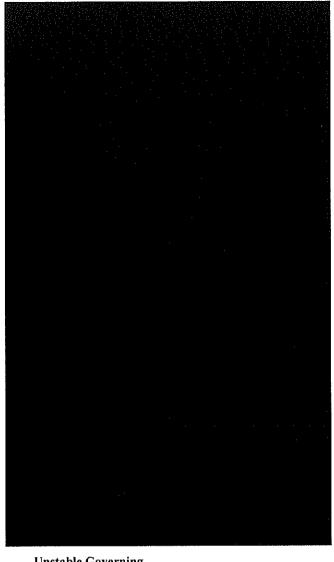
System		Chapter
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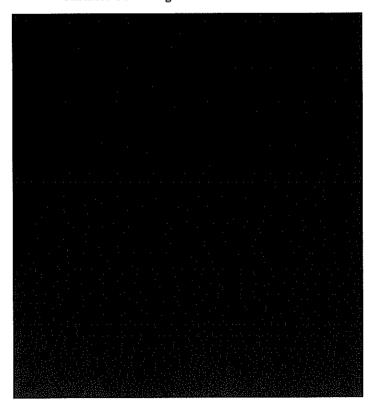


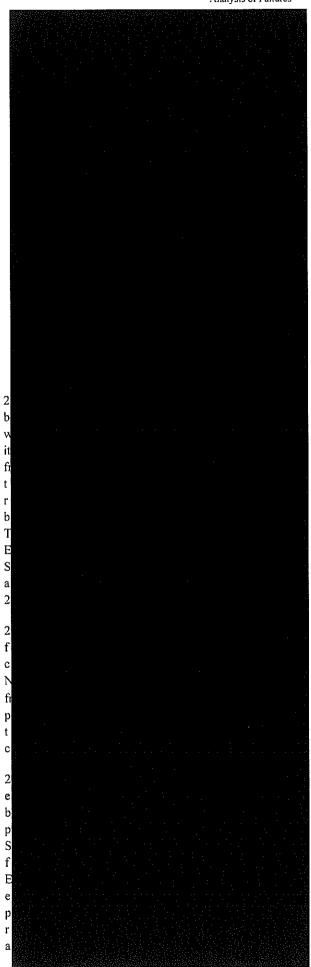


AP101C-1308-15C Analysis of Failures

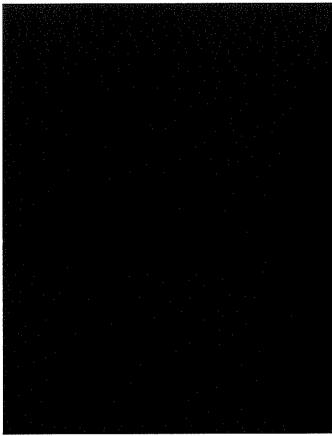


**Unstable Governing** 

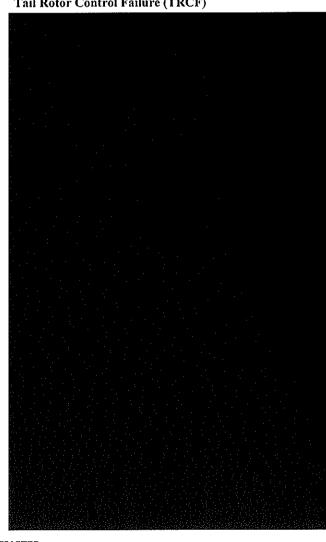


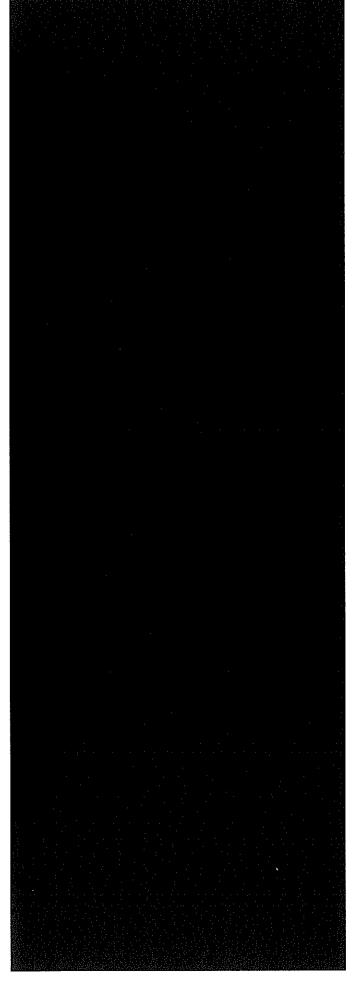




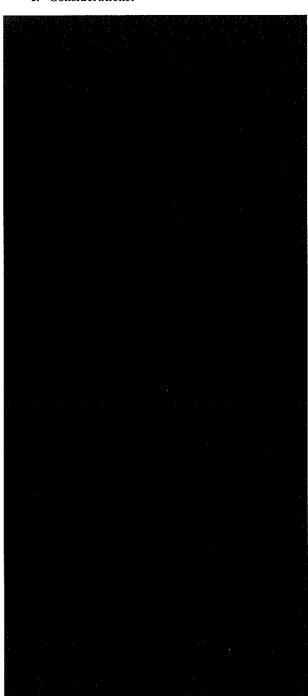


Tail Rotor Control Failure (TRCF)

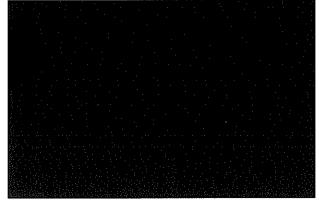




## c. Considerations:

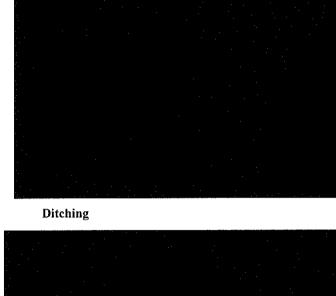


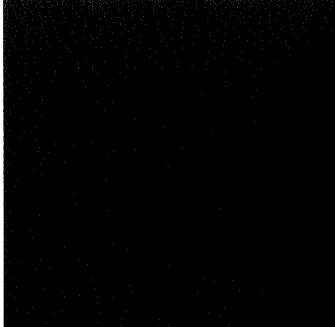
#### b. Considerations:

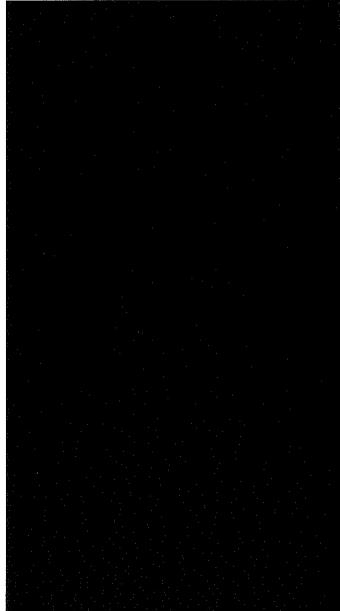


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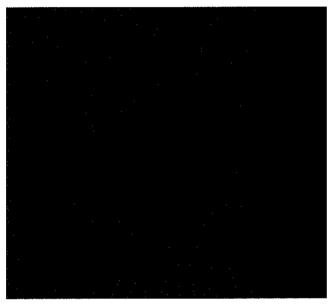








Double Generator Failure



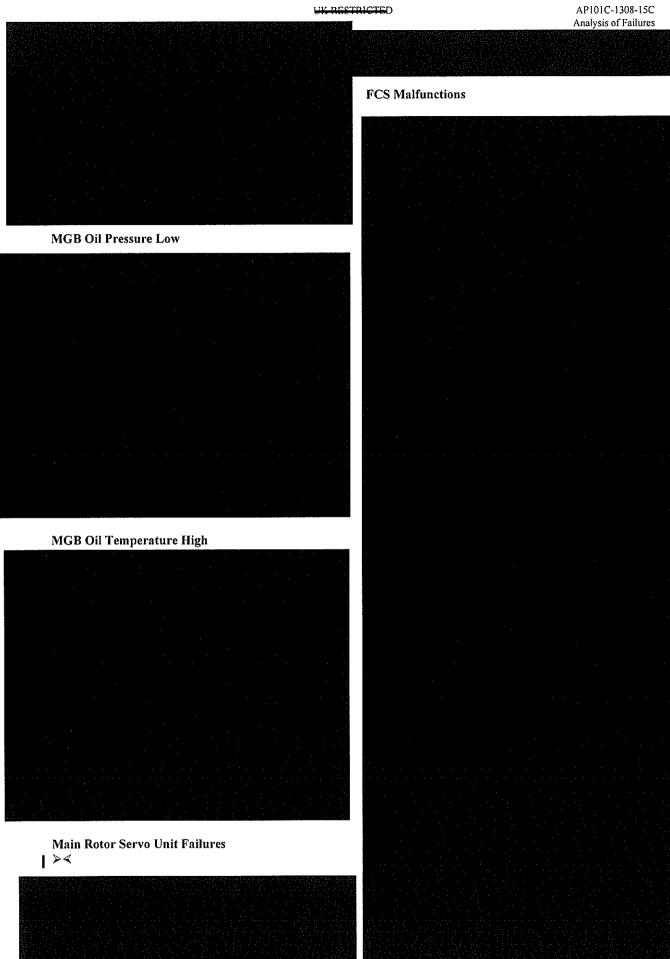
Double Alternator Failure

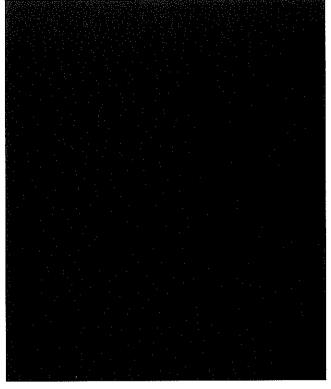


Hydraulic System Failures

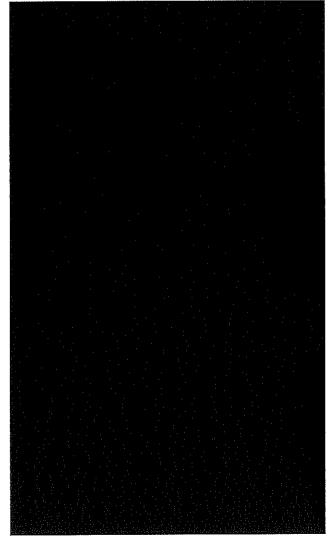


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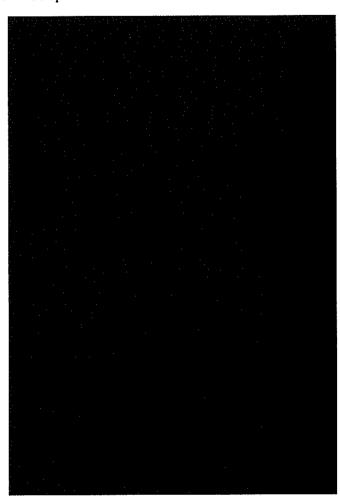




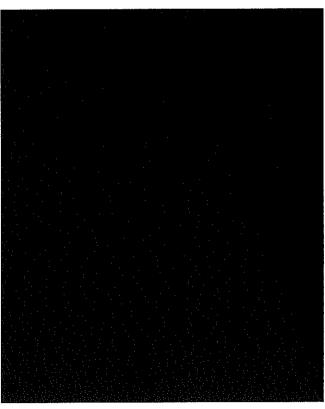
## 57. FCS Actuator Seizure:



# 58. FCS Spool Valve Seizure:



## Feedback Failures

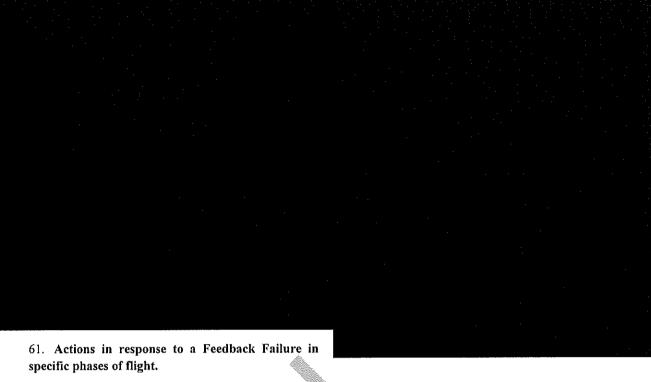


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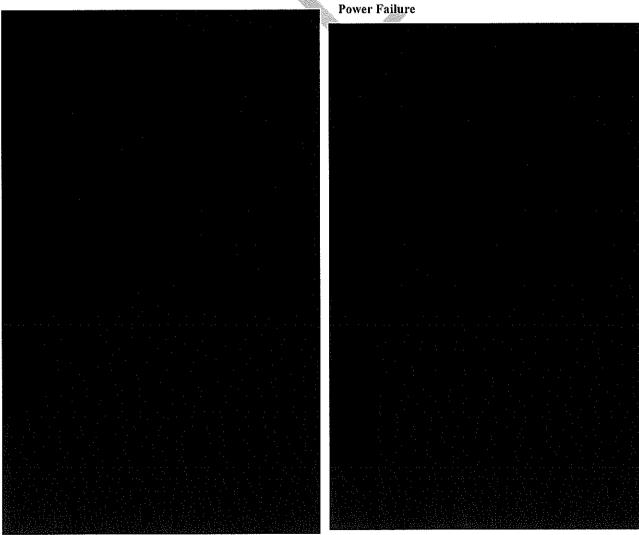


Analysis of Failures

#### **Hold Malfunction**

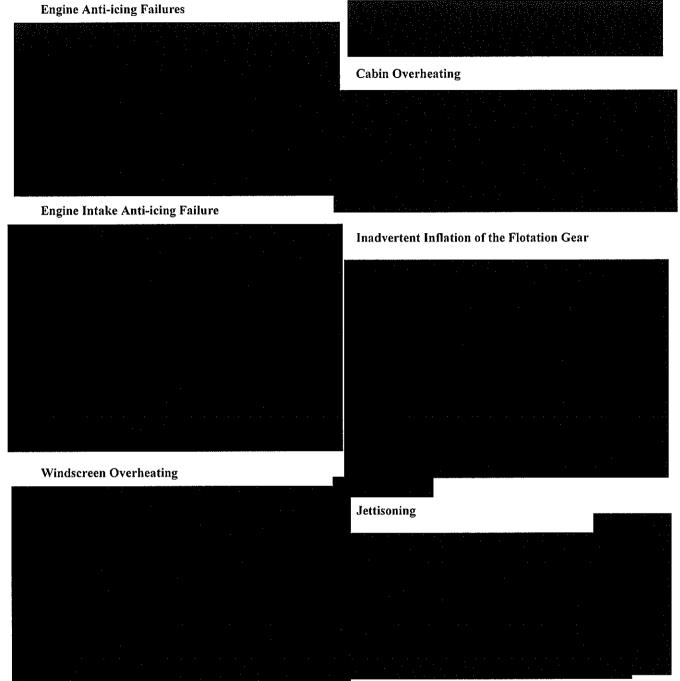


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## Low Fuel Pressure





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# PART 5

	Lis	t of C	hapt	ers			A STATE OF THE PARTY OF THE PAR	d
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COCKPIT ILLUSTRATIONS		50.000	15.175		222	1	1000	

## PART 5

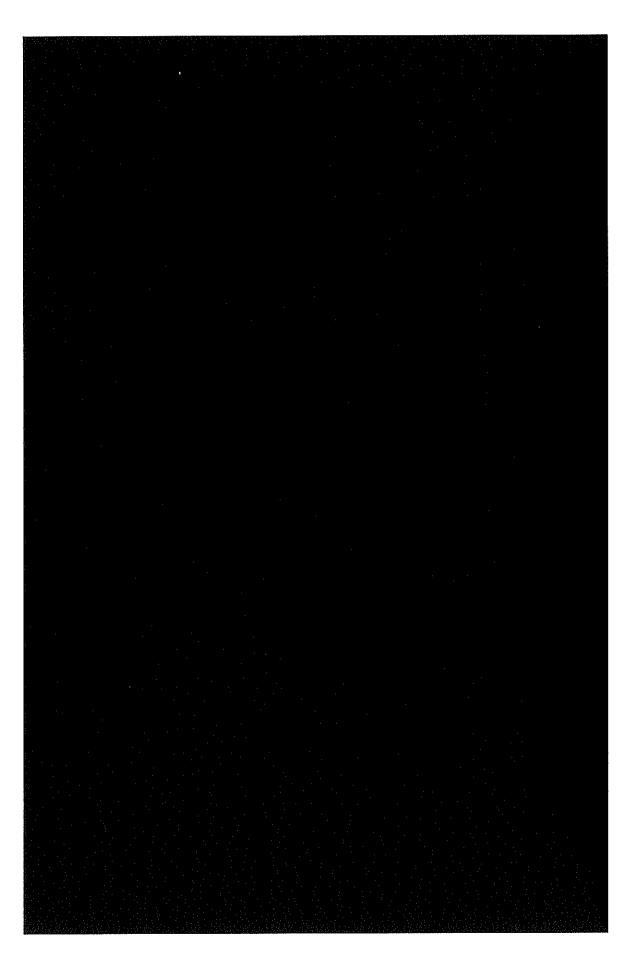
# **CHAPTER 1 - COCKPIT ILLUSTRATIONS**

Illustrations				Fig
Cockpit Controls and Indicators (1)	 	 	•••	1
Cockpit Controls and Indicators (2)	 	 	•••	2
Cockpit Controls and Indicators (3)	 	 		3
Cockpit Controls and Indicators (4)	 	 		4
Cockpit Controls and Indicators (5)	 	 		<b>∞</b> 5′

Key to Fig 1 - Cockpit Controls and Indicators (1)

		Referei	nce
		Topic (	Chapter
1	Cockpit fresh air control	15C	10
2	Adjustable louvre	15C	10
3	Observer's foot operated P to T switches	15D	5
4	Louvre control	15C	10
5	Punkah louvre	15C	10
6	Map stowage	15C	10
7	Not used	15D	5
8	OAT gauge	15C	8
9	Port door jettison handle	15C	10
10	Standby compass	15C	8
11	Stbd door jettison handle	15C	10
12	Signal cartridges	15C	10
13	Signal pistol	15C	10
14	Pilot's yaw pedal adjuster	15C	5
15	Map stowage	15C	10
16	Adjustable louvre	15C	10
		284.	

		Reference
		Figure
De	tails	
A	Cockpit instrument panel	2
В	Cyclic stick handgrip	2
C	Collective lever handgrip	2
D	Cockpit centre console	3
E	Cockpit roof panel	4
F	Engine control quadrant	4
G	Cold start switch panel	4
Н	Ground servicing switch panel	4
Ι	Emergency services safety break	5
J	Control frame switch box	5
K	Blade tracking and general purpose DC supply	5
	sockets	



5-1 Fig 1 Cockpit Controls and Indicators (1)

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Indicators
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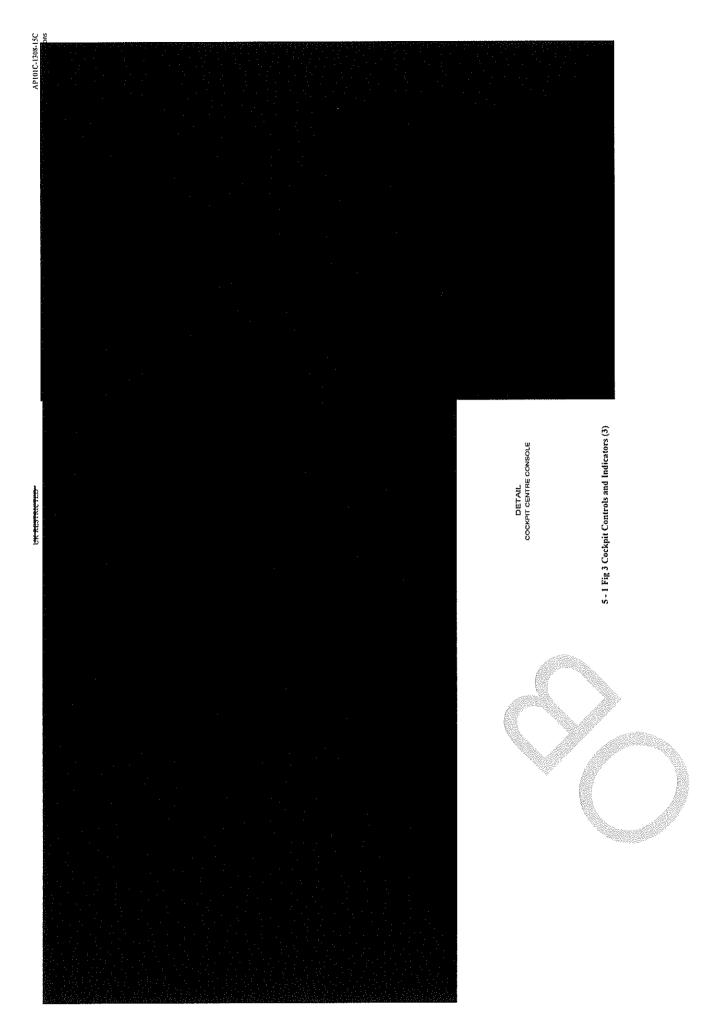
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		vejerence			361	3713	
	Торкс	Chapter			Topic	Chapter	
Detail A							
Observer's Instrument Panel			33	Torquemeter	15C	47	
l Blank		h	34	Standby attitude indicator	15C	00	
2 Floodlight and dimmer	15C	∞	35	Heading indicator	15C	œ	
3 Barometric altimeter	15C	œ	36	Attitude indicator	15C	00	
4 Airspeed indicator	15C	∞	37	Amber attention getter and cancel	15C	7	
5 Heading indicator	15C	<b>∞</b>	38	Red attention getter and test	15C	2	
<ol> <li>Radar display and control indica- tor</li> </ol>	15D	9	39	Radar altimeter	15C	00	
7 Blank		,	40	Barometric altimeter	15C	00	
8 Blank			41	Vertical speed indicator	150	90	
9 Dísplay control unit	15D	2	42	Jettison mode selector	15C	6	
			43	Harpoon engaged light	15C	10	
Main Instrument Panel							
10 Not used	15D	7	Det	Detail B			
<ol> <li>Tactical situation display - graphics</li> </ol>	15D	4	Ç	Cyclic Stick Handgrip			
12 ESM control indicator	15D	7	4	External cargo release button	150	5,9	
13 Central warning panel	15C	7		(guarded)			
14 Clock	15C	8	45	Blank		•	
15 No 1 engine Nu tachometer	150	ব	46	CWP cancel button	15C	2,5	
16 No 2 engine Nu tachometer	15C	4	47	CAC cutout button	15C	5,6	
17 No 1 engine T6 indicator	15C	4	48	AFCS cutout button	15C	5,6	
18 No 2 engine T6 indicator	15C	4	49	Press to transmit button	15C	5	
19 No I engine oil pressure	15C	*3*	20	Cyclic stick trim switch	15C	5	
20 No 2 engine oil pressure	15C	77					
21 No I engine oil temperature	15C	47			100		
22 No 2 engine oil temperature	15C	4	Det	Detail C			
23 Fuel contents selector	15C	3	Pilo	Pilot's Collective Lever Handgrip		i i	
24 Fuel contents gauge	15C	Э	51	Jettison button (guarded)	15C	5.9	
25 Hydraulic pressure gauges (3 off)	15C	7	52	Landing light switch	ISC	5, 10	
26 Master Armament Safety Switch	15D	6	53	Landing light directional control	2	5, 10	
(MASS)			ķ	Harpoon engage button	150	5, 10	
27 Transmission oil temperature	15C	5	55	Emergency, flotation button	15C	5, 10	
28 Transmission oil pressure	150	S		(guarded)	les.		
29 MGB oil cooler shutter override	150	5	56	Harpoon release button	15C	'n	
swítch			57	Press to mute button	15C	'n	
30 Ng and Mf triple tachometer	15C	4,5	58	Collective channel release button	15C	5,6	
31 Red attention getter and cancel	150	7	59	Rescue hoist control switch	15C	5,9	
32 Airspeed indicator	15C	∞	09	Hydraulic servo selector switch	15C	5,7	

AP101C-1308-15C Cockpit Illustrations

D. C. C. C. C.	vererence	•	

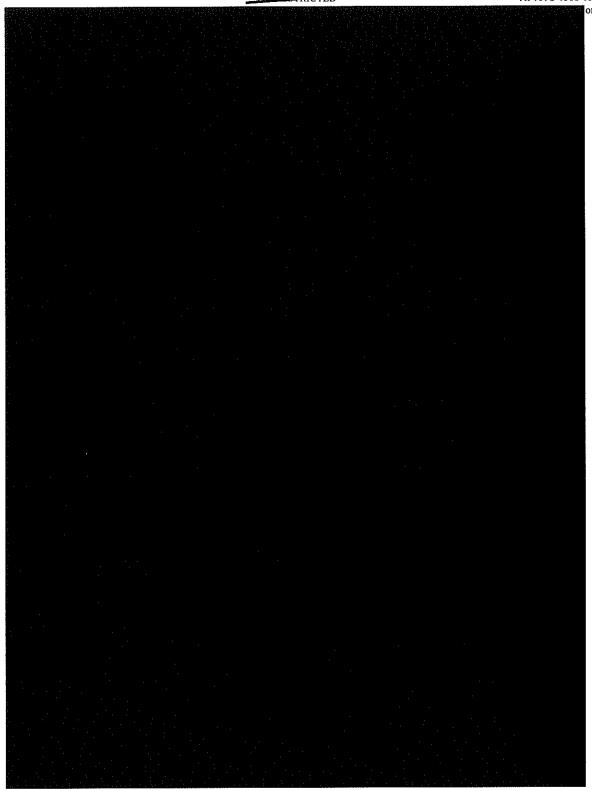
					6														
Chapter			7	2	5, 9, 10	90	5	5	00	9	2	3	•	10	δ	5	9	9	2,9
Topic			15D	15D	15C	15D	15D	15D	15D	15D	15D	15D	150	150	15D	15D	15C	15C	15D
	Detail D	Cockpit Centre Console	Observer's control display navigation unit	Pilot's control display navigation unit	Miscellancous switch panel	Ground speed and drift indicator	Observer's communications control panel	Observer's remote control unit	Auxiliary control panel	IFF/SSR control unit	Data transfer device receptacle unit	Master erase switch panel	Compass controller	Cockpit lighting control panel	Pilot's communication control panel	Pilot's remote control unit	AFC engage controller	AFC test controller	Stores release panel
	Ď	S	-	7	3	7	8	9	7	8	6	2	Ξ	12	13	7	15	16	17

GITTO GETT G. III



		2	
	Topic Chapter	hapter	
Detail E			
Cockpit Roof Panel			
1 Miscellancous switch panel		8, 10	
2 Engine anti-icing and lighting switches	18C	4	
3 Engine control panel	1SC	4	
4 Fuel control panel	15C	3	
5 AC control panel	15C	1	
6 DC control panel	15C	_	
Detail F			
Engine Control Quadrant	18C	4	
7 Starting isolation microswitch	18C	4	
8 Starting isolation microswitch	150	4	
9 Cabin air control lever	15C	01	
10 PCU microswitch	15C	4	
11 No 2 engine condition lever	15C	4	
12 Starter button and fire warring light	15C	4	
13 No 2 engine speed trimming knob	15C	Ť	
14 Speed select lever	15C	4	
15 PCU microswitch	15C	4	
16 No 1 engine condition lever	15C	4	
17 Starter button and fire warning light	15C	4	
18 Rotor brake lever	180	5	
Detail G			
Cold Start Switch Panel	15C	<b>J</b>	
Detail H			
Ground Servicing Switch Panel	15C	6.8	





5 - 1 Fig 5 Cockpit Controls and Indicators (5)

